MASTER MINIMUM EQUIPMENT LIST BOMBARDIER BUSINESS JET

CANADAIR CHALLENGER

CL 600/601/601-3A/601-3R/604/605/650

TRANSMITTAL LETTER

This letter presents Revision 13 of the Challenger 600 Master Minimum Equipment List for all variants, dated August 24, 2023.

The associated Challenger 600 Dispatch Deviation Guide (DDG) will be issued to support this MMEL. The DDG is to be issued within 30 days after approval.

TRANSPORT CANADA

MASTER MINIMUM EQUIPMENT LIST

Bombardier Business Jet

CL 600/601/601-3A/601-3R/604/605/650

First Issued: 19 April 1989 Revision 13: 24 August 2023

MINISTÈRE DES TRANSPORTS LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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TRANSPORT CANADA

MASTER MINIMUM EQUIPMENT LIST

CANADAIR CHALLENGER

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CL-605	(CL600-2B16)
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Manual Approval

Approved:

Jason Clinistopher Kandall

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Chief, Flight Test, National Aircraft Certification for Minister of Transport

Date: August 24, 2023

Revision 13

MINISTÈRE DES TRANSPORTS LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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RECORD OF REVISIONS

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Rev. 3	All	
Rev. 4	Title Page, TOC-1, TOC-2, Manual Approval Page, Rev-1, Rev-2, TpRv-1, TpRv-2, LEP-1 to LEP-3, HOC-1, HOC-2, 21-5 to 21-8, 21-11,21-12 / 22-5, 22-6 / 24-1, 24-2, 24-5 to 24-10 / 26-3, 26-4 / 27-1, 27-2 / 28-3, 28-4,28-9, 28-10 / 30-3 to 30-8 / 32-1 to 32-4 / 34-1 to 34-4, 34-7 to 34-10 / 36-1, 36-2 / 45-1, 45-2 / 52-1, 52-2 / 74-1, 74-2 / 76-1, 76-2 / 77-1, 77-2 / 78-1, 78-2 / 79-1, 79-2	
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Rev 11	Transmittal Letter Page, Title Page, Manual Approval Page, REV-2, TpRv-3, MMEL USER COMMENTS, LEP-1 to LEP-4/ HOC-1 to HOC-2 / ABB-7 / 22-2, 22-3, 22-4, 22-5, 22-6, 22-7, 22-8, / 24-8, 24-9 / 25-3, 25-7 / 27-3, 27-4, 27-5 /28-4 / 29-4 / 30-3, 30-4, 30-5, 30-6, 30-7, 30-8, 30-9, 30-10 / 31-2 / 32-2 / 33-11 / 34-9, 34-11, 34-14, 34-16 / 78-1	
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RECORD OF TEMPORARY REVISIONS

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Temp Rev#	Pages Affected	TC approval	Description of Revision	Date Incorporated Signature
1	30-6	K.J. Mansfield chief, Flight test, 20 December, 1995	Revises MMEL relief for Cowl Anti-Ice Pressure Relief Blow- off valves.	Superseded by Rev. 3 dated 23 December 1995
2	28-2	K.J. Mansfield chief, Flight test, 19 May, 1995	Removes limitation to carry additional 150 pounds of fuel when one electric Fuel Pump is inoperative.	Superseded by Rev. 3 dated 23 December 1995
3	24-1	K.J. Mansfield chief, Flight test, 20 December, 1995	Deletes MMEL ADG Auto- Deply System relief	Superseded by Rev. 3 dated 23 December 1995
4	28-2	K.J. Mansfield chief, Flight test, 20 December, 1995	Adds left boost pump proviso requiring APU to be operating	Superseded by Rev. 3 dated 23 December 1995
5	26-3 36-1 78-1	L. Farrell for K.J. Mansfield chief, Flight test, 17 January, 1996	Deletes proviso from 26-14-1, adds NOTE to 36-10-1, and adds proviso to 78-30-1.	Superseded by Rev.4 dated 23 February 1996
6	24-4	R. Walker for K.J. Mansfield chief, Flight test, 27 March, 1996	Reinstates MMEL relief for ADG Auto-Deployment System	Superseded by rev. 5 dated January 06/97
7	52-1	L. Galvin K.J. Mansfield chief, Flight test, 05 March, 1996	Recinds MMEL relief for Emergency Exit	Superseded by Rev. 5 dated January 06/97
8	22-1, -2 22-3, -6 30-6, 34-4, -9 34-17	L. Farrell for K.J. Mansfield chief, Flight test, 14 August, 1996	Introduces RVSM exceptions to existing MMEL relief	Superseded by Rev. 5 dated January 06/97
9	28-2	R. Walker A/Chief flight Test for Minister of Transport 22 May 1997	Adds relief for damaged/missing Turolator Vane	Superseded by TR#11, dated 19 June 1997
10	32-3	L. Farrell A/Chief Flight Test for Minister of Transport 22 May 1997	Add provisos for Nose wheel Steering System	Superseded by Rev. 6 dated 22 November 2001
11	28-2	R. Walker A/Chief flight Test for Minister of Transport, 19 June 1997	Corrects item callout for TR#9 to 28-13-3 (was identified as 28-13-2 in TR #9	Superseded by Rev. 6 dated 22 November 2001
12	32-4	William Jupp for Chief Flight Test for Minister of Transport, 25 November 1997	Revises MMEL relief for Nose Ldg Gear Door Release Mechanism	Superseded by Rev. 6 dated 22 November 2001

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13	56-1	William Jupp for Chief Flight Test for Minister of Transport, 11 October 2002	Re-introduces MMEL relief for Windshield Face Ply	Superseded by Rev. 7 dated 12 November 2004
14	30-1 30-2	William Jupp for Chief Flight Test for Minister of Transport, 29 March 2005	Removes the relief to dispatch in wing anti-ice standby mode	Superseded by TR 14-1 dated 29 March 2005
14-1	30-1 30-2	William Jupp for Chief Flight Test for Minister of Transport, 20 April 2005	Revises TR 14 to replace Modsum 600-10426 by SB 601-0575 and 604-30-002	Superseded by TR 14-2 dated 12 January 2006
14-2	30-1 30-2	William Jupp for Chief Flight Test for Minister of Transport, 12 January 2006	Revises TR 14-1 to remove CL-601 (CL-600-2A12) from the restrictions imposed to A/C without SB 601-0575	Superseded by revision 8 dated 15-October 2007
15	Cancelled			
16	27-3	R. Walker, A/Chief, Flight Test for Minister of Transport, March 17, 2009	Revises MMEL item 27-40-1	Superseded by TR 16-1 dated 29 March 2005
16-1	27-3	J. Martin, A/Chief, Flight Test for Minister of Transport, June 24, 2009	Revises TR 16 to correct the number installed and number required for dispatch and add model CL605 to the applicable aircraft types	Superseded by Rev. 9 dated October 24/14
17	24-1 24-2	J. Martin, A/Chief, Flight Test for Minister of Transport, November 20, 2009	 Allows dispatch with one engine IDG inoperative for CL604 aircraft with APU 36-150 and SB 604-24-020 incorporated. SB 604-24-020 changes the APU generator exciter cut-off logic and makes it possible to keep the APU generator to ON during flight. Allows dispatch for one flight with one engine IDG inoperative and cross-side ACMP selected ON for 601-3A, 601-3R aircraft with SB 601-0568 incorporated. 	Superseded by Rev. 9 dated October 24/14

DEPARTMENT OF TRANSPORT

MINISTÈRE DES TRANSPORTS MASTER MINIMUM EQUIPMENT LIST LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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18	30-5	Walter Istchenko, Chief, Flight Test for Minister of Transport, December 16, 2009	Allows dispatch with the Ice Protection sub-system of the Enhanced Vision System (EVS) inoperative. The EVS is an optional system.	Superseded by Rev. 9 dated October 24/14
19	56-1	Walter Istchenko, Chief, Flight Test for Minister of Transport, May 12, 2011	 Includes the glass face ply in the MMEL relief. Acrylic face ply from other manufacturers than Sierracin are also included and the statement "Sierracin Only" is removed. Repair procedures are added accordingly. 	Superseded by Rev. 9 dated October 24/14
20	30-2	Walter Istchenko Chief, Flight Test for Minister of Transport, June 15, 2015.	New Item 30-12-5 introduced as part of Bombardier Take- Off Safety Enhancement (TOSE) long term action plan.	Superseded by Rev. 10 dated Nov 25/15
21	30-2	TRS No 24 Walter Istchenko Chief, Flight Test for Minister of Transport, 17 December 2015.	Item 30-12-5 was inadvertently omitted from the latest Revision 10, thus, this item is reintroduced as part of Bombardier Take-Off Safety Enhancement (TOSE) long-term action plan.	Not incorporated. Superseded by TR 22.
22	30-2	TRS No 25 Walter Istchenko Chief, Flight Test for Minister of Transport, 24 November 2016.	This TR provides additional information that was omitted from TR#21 (see above). TR#21 did not clearly state effectivity for model CL-650 along with additional Service Bulletin details associated with models 604 and 605.	Superseded by Rev. 11 dated Nov 23/17
23	30-5	TRS No 26 Walter Istchenko Chief, Flight Test for Minister of Transport, 03 November 2017.	Item 30-41-2 is updated to remove the restrictions on operation in known or forecast icing conditions.	Superseded by Rev. 11 dated Nov 23/17

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MMEL USER COMMENTS

ON ERRORS, OMISSIONS, PROCEDURES (IF APPLICABLE), ETC.

FAX to:	(514) 855-7401 Bombardier Aerospace RM&S - Reliability, Maintainability, and Safety Department 15T Systems Engineering			
OR				
Email to:	martin.zeppetelli@aero.bombardier.com			
OR				
Mail to:	address on back of this page			
From:	Telephone No.:			
Company:	Email address:			
Date:				
The MMEL re	quires the following correction or clarification:			
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	OUR RESPONSE			
Thank you for	your comment;			
 We will issue Temporary Revision No We will include in Revision scheduled for: 				
Comment:				
Date:				

Bombardier Aerospace 500 Cote-Vertu Road West Dorval, Quebec H4S 1Y9

Canada

Attention: Martin Zeppetelli

RM&S - Reliability, Maintainability and Safety

Dept. 15T

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33-4	Nov 25/15		34-27	Nov 25/15	
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33-6	Aug 24/23		34-29	Aug 24/23	
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<u>HIGHLIGHTS OF CHANGE</u> <u>SUMMARY OF TECHNICAL AND EDITORIAL CHANGES:</u>

Item No	Type of change	Summary of changes
N/A	Technical	Category A revised per TCCA requirement.
N/A	Technical	ADDITIONAL DEFINITIONS item 3 revised per TCCA MMEL Guidance Book Rev 8.
N/A	Technical	ADDITIONAL DEFINITIONS item 5 added per TCCA MMEL Guidance Book Rev 8.
21-24-1	Technical	(M) revised to (O).
21-30-2 1)	Editorial	The second proviso is revised per TCCA MMEL Guidance Book Rev 8.
21-30-2 2)	Editorial	The third proviso is deleted.
21-30-3	Technical	Repair interval is revised from "D" to "C" per TCCA MMEL Guidance Book Rev 8 for the second proviso.
21-30-4	Technical	The second proviso repair interval is revised from "D" to "C" per TCCA MMEL Guidance Book Rev 8.
21-31-1 1)	Technical	Proviso c) added per TCCA MMEL Guidance Book Rev 8.
21-51-1	Technical	The third proviso revised to add proviso c) per TCCA MMEL Guidance Book Rev 8.
21-61-5	Technical	(M) revised to (O).
22-11-1	Technical	The second proviso repair interval revised from C to B per TCCA MMEL Guidance Book Rev 8.
22-30-1	Technical	(M) revised to (O); "Routine" deleted for the second proviso per TCCA requirement.

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Item No	Type of change	Summary of changes
23-11-1 2)	Technical	The first proviso revised per TCCA MMEL Guidance Book Rev 8.
23-11-4	Technical	Proviso revised per TCCA MMEL Guidance Book Rev 8.
23-11-4 1)	Technical	Subtitle revised (Applicability revised); Proviso revised per TCCA MMEL Guidance Book Rev 8.
23-11-4 1) a)	Technical	Proviso revised per TCCA MMEL Guidance Book Rev 8.
23-11-4 2)	Technical	Proviso revised per TCCA MMEL Guidance Book Rev 8.
23-21-2	Editorial	The second proviso revised: "routine" deleted per TCCA requirement.
23-31-1 1)	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
23-41-1	Technical	The second proviso added per TCCA MMEL Guidance Book Rev 8.
23-41-1 1)	Technical	Repair interval revised from "B" to "C" per TCCA MMEL Guidance Book Rev 8.
23-41-1 2)	Technical	Repair interval revised from "B" to "C" per TCCA MMEL Guidance Book Rev 8.
23-41-1 3)	Technical	The first proviso (M) removed per TCCA MMEL Guidance Book Rev 8.
23-50-2 2)	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
23-50-2 3)	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
23-51-1	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
23-51-6	Technical	New item introduced.

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Item No	Type of change	Summary of changes
23-71-1 1)	Editorial	Subtitle revised per TCCA MMEL Guidance Book Rev 8.
23-71-1 1) a)	Technical	Applicability revised.
23-71-1 2)	Editorial	Subtitle revised per TCCA MMEL Guidance Book Rev 8.
23-71-1 2) a)	Technical	Applicability revised.
23-71-1 3)	Editorial	Subtitle revised per TCCA MMEL Guidance Book Rev 8.
23-71-1 3) a)	Technical	Applicability revised.
24-50-1	Technical	(M) revised to (O).
25-12-1	Technical	The second proviso a) revised per TCCA MMEL Guidance Book Rev 8.
25-20-1	Technical	(M) revised to (O).
25-20-1 1)	Editorial	Revised per TCCA MMEL Guidance Book Rev 8.
25-20-1 2)	Editorial	Revised per TCCA MMEL Guidance Book Rev 8.
25-20-1 3)	Editorial	Revised per TCCA MMEL Guidance Book Rev 8.
25-20-1 4)	Editorial	Revised per TCCA MMEL Guidance Book Rev 8.
25-29-1	Technical	(M) revised to (O).

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Item No	Type of change	Summary of changes
25-29-2	Technical	New item introduced.
25-30-1	Editorial	The first proviso a) revised per TCCA MMEL Guidance Book Rev 8.
25-60-2 1)	Technical	The second proviso added per TCCA MMEL Guidance Book Rev 8.
25-60-3 1)	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
25-60-3 2)	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
25-60-9	Technical	Proviso b) revised per TCCA MMEL Guidance Book Rev 8.
25-60-9 1)	Technical	Proviso d) revised per TCCA MMEL Guidance Book Rev 8.
25-70-1	Technical	(M) revised to (O).
25-70-2	Technical	Revised per TCCA MMEL Guidance Book Rev 8 ("days" revised to "calendar days").
25-70-3	Technical	New item introduced.
26-10-2	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
26-14-1	Editorial	NOTE revised per TCCA MMEL Guidance Book Rev 8.
26-20-1	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
26-20-3	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
26-25-1	Technical	(M) revised to (O); Proviso c) revised per TCCA MMEL Guidance Book Rev 8.

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Item No	Type of change	Summary of changes
27-50-1	Technical	(M) revised to (O) and NOTE revised.
27-60-2	Editorial	Edit change, "ft" revised to "feet" per TCCA MMEL Guidance Book Rev 8.
28-11-1	Editorial	Edit change of proviso a) and d).
28-11-3	Technical	New item introduced.
28-27-1	Technical	NOTE added.
30-12-1 1)	Technical	(M) revised to (O).
30-12-1 3)	Technical	(M) revised to (O).
30-41-2 2)	Editorial	The second proviso a) revised: "Routine" deleted per TCCA requirement.
30-80-1	Technical	(M) revised to (O).
31-50-2	Editorial	The second proviso b) revised: "Routine" deleted per TCCA requirement.
31-50-4	Editorial	The second proviso b) revised: "Routine" deleted per TCCA requirement.
33-13-1	Technical	The second proviso revised per TCCA MMEL Guidance Book Rev 8.
33-23-1	Technical	The first proviso (M) revised to (O); The second proviso revised per TCCA MMEL Guidance Book Rev 8.
33-40-2	Technical	The repair interval of the second proviso revised from "D" to "C" per TCCA MMEL Guidance Book Rev 8.

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Item No	Type of change	Summary of changes
33-40-3	Technical	The repair interval of the second proviso revised from "D" to "C" per TCCA MMEL Guidance Book Rev 8.
33-40-4 3)	Technical	(M) revised to (O).
34-12-7 2)	Technical	(M) revised to (O) and proviso revised per TCCA requirement.
34-13-2	Technical	Proviso revised per TCCA requirement.
34-12-7 3)	Technical	New item introduced.
34-22-2	Technical	The first and second proviso added per TCCA MMEL Guidance Book Rev 8.
34-28-1	Technical	Revised per TCCA requirement.
34-32-1	Technical	(M) revised to (O).
34-40-1	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
34-42-1	Technical	Repair interval revised per TCCA MMEL Guidance Book Rev 8.
34-42-2 1) d)	Technical	Repair interval revised per TCCA MMEL Guidance Book Rev 8.
34-42-2 1) e)	Technical	The first proviso for Class A TAWS Equipment Required revised per TCCA MMEL Guidance Book Rev 8.
34-42-2 2)	Technical	Proviso for Class A TAWS Equipment Required revised per TCCA MMEL Guidance Book Rev 8.
34-42-2 4)	Editorial	Edit change of subtitle.

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Item No	Type of change	Summary of changes
34-43-1	Technical	Revised per TCCA MMEL Guidance Book Rev 8. (M) revised to (O).
34-43-1 1)	Technical	For TCAS II: Revised per TCCA MMEL Guidance Book Rev 8.
34-43-1 2)	Technical	For TCAS II: Revised per TCCA MMEL Guidance Book Rev 8.
34-43-1 3)	Technical	For TCAS II: Revised per TCCA MMEL Guidance Book Rev 8.
34-51-1	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
34-52-1	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
34-54-1	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
34-54-1 1)	Technical	Subtitle revised per TCCA MMEL Guidance Book Rev 8.
34-54-1 2)	Editorial	Subtitle number revised.
34-61-1 1)	Technical	The second proviso added per TCCA MMEL Guidance Book Rev 8.
34-61-1 2)	Technical	The second proviso added per TCCA MMEL Guidance Book Rev 8.
34-61-1 3)	Technical	The second proviso added per TCCA MMEL Guidance Book Rev 8.
34-61-1 4)	Technical	The second proviso added per TCCA MMEL Guidance Book Rev 8.
34-61-1 6)	Technical	Revised per TCCA MMEL Guidance Book Rev 8.
35-20-1	Editorial	Edit change of proviso a), "ft" revised to "feet" per TCCA MMEL Guidance Book Rev 8.

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Item No	Type of change	Summary of changes
35-20-2	Technical	(M) removed.
35-20-3	Technical	Original item 1) removed and original item 2) renumbered as 1).
35-25-1	Technical	(M) removed.
35-30-1	Technical	(M) removed.
38-10-1	Technical	The second proviso added per TCCA MMEL Guidance Book Rev 8.
38-10-2	Technical	The first proviso revised per TCCA MMEL Guidance Book Rev 8. The second proviso repair interval revised from "D" to "C" per TCCA MMEL Guidance Book Rev 8.
46-10-1	Technical	Title revised and Subtitle "1) Class 3 EFB" deleted per TCCA MMEL Guidance Book Rev 8.
46-10-1 1)	Technical	Subtitle revised per TCCA MMEL Guidance Book Rev 8.
46-10-1 2)	Technical	Subtitle revised per TCCA MMEL Guidance Book Rev 8.
46-10-1 3)	Technical	Subtitle revised per TCCA MMEL Guidance Book Rev 8.
46-10-2 1)	Editorial	The second proviso revised: "Routine" deleted per TCCA requirement.
46-10-2 2)	Editorial	The second proviso revised: "Routine" deleted per TCCA requirement.

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SUMMARY OF TECHNICAL AND EDITORIAL CHANGES:

(<u>C</u>	0	n	ıť	d)

Item No	Type of change	Summary of changes
46-10-2 3)	Editorial	The second proviso revised: "Routine" deleted per TCCA requirement.
46-10-2 4)	Editorial	The second proviso revised: "Routine" deleted per TCCA requirement.
46-20-1	Editorial	Proviso revised: "Routine" deleted per TCCA requirement.
49-00-4	Technical	The first proviso number installed revised from "1" to "-"; The subtitle deleted.
52-30-1	Technical	(M) revised to (O).
56-10-1	Editorial	Edit change, "ft" revised to "feet" per TCCA MMEL Guidance Book Rev 8.

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DEFINITIONS

COLUMN 1

- "System & Sequence Numbers" are based on Air Transport Association (ATA) Specification No. 100 and items are numbered sequentially.
- 2. "Item" means the equipment, system, component, or function listed in the "Item" column.
- 3. **Repair Intervals:** All users of an MEL must effect repairs of inoperative systems or components, deferred in accordance with the MEL at or prior to the repair times established by the following letter designators:

Category "A":

Items in this category shall be repaired within the time interval specified in the operator's approved MEL Whenever the proviso in the "Remarks or Exceptions" column of the MMEL states cycles or flight time, the time interval begins with the next flight. Whenever the time interval is listed as flight days, the time interval begins on the flight day following the day of discovery.

Category "B": Items in this category shall be repaired within three (3) consecutive calendar days (72 hours), excluding the day the malfunction was recorded in the Aircraft Maintenance Record/Logbook. For example, if it were recorded at 10 a.m. on January 26th, the three-day interval would begin at midnight the 26th and end at midnight the 29th.

Category "C": Items in this category shall be repaired within ten (10) consecutive calendar days (240 hours), excluding the day the malfunction was recorded in the Aircraft Maintenance Record/Logbook. For example, if it were recorded at 10 a.m. on January 26th, the 10-day interval would end at midnight February 5th.

Category "D": Items in this category shall be repaired within one hundred and twenty (120) consecutive calendar days (2880 hours), excluding the day the malfunction was recorded in the Aircraft Maintenance Record/Logbook.

The letter designators are inserted adjacent to Column 2.

4. "***" symbol indicates an item which is not required by Regulations but which may have been installed on some models of aircraft covered by this MMEL. This item may be included on the operator's MEL after the approving office has determined that the item has been installed on one or more of the operator's aircraft. The symbol, however, shall not be carried forward into the operator's MEL. It should be noted that neither this policy nor the use of this symbol provide authority to install or remove an item from an aircraft. The "***" symbol may be considered equivalent to the term "if installed".

COLUMN 2

- 1. **"Number Installed"** is the number (quantity) of items installed in the aircraft. This number represents the aircraft configuration considered in developing this MMEL. Should the number be a variable (e.g., passenger cabin items) a number is not required.
- 2. "-" symbol indicates a variable number (quantity) of the item installed.

MOTE: Where the MMEL shows a variable number installed, the MEL must reflect the actual number installed or an alternate means of configuration control approved by Transport Canada.

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COLUMN 3

- 1. **"Number Required For Dispatch"** is the minimum number (quantity) of items required for operation provided the conditions specified in Column 4 are met.
- 2. "-" symbol indicates a variable number (quantity) of the item required for dispatch.

NOTE:

Where the MMEL shows a variable number required for dispatch, the MEL must reflect the actual number required for dispatch or an alternate means of configuration control approved by Transport Canada.

COLUMN 4

- 1. **"Remarks or Exceptions"** in this column includes a statement either prohibiting or permitting operation with a specific number of items inoperative, provisos (conditions and limitations) for such operation, and appropriate notes.
- 2. "(M)" symbol indicates a requirement for a specific maintenance procedure which must be accomplished prior to operation with the listed item inoperative. If there is a requirement to perform the procedures more frequently, for example, before each flight, then this is clearly stated in the provisos. Normally these procedures are accomplished by maintenance personnel; however, other personnel may be qualified and authorized to perform certain functions. Procedures requiring specialized knowledge or skill, or requiring the use of tools or test equipment should be accomplished by maintenance personnel. The satisfactory accomplishment of all maintenance procedures, regardless of who performs them, is the responsibility of the operator.

Appropriate procedures, including the identification of qualified personnel authorized to accomplish them are required to be published as part of the operator's manual or MEL.

3. "(O)" symbol indicates a requirement for a specific operations procedure which must be accomplished in planning for and/or operating with the listed item inoperative. Normally these procedures are accomplished by the flight crew; however, other personnel may be qualified and authorized to perform certain functions. The satisfactory accomplishment of all procedures, regardless of who performs them, is the responsibility of the operator.

Appropriate procedures are required to be published as part of the operator's manual or MEL.

These procedures are outlined in the Dispatch Deviation Guide (PSP-621). If a procedure is not covered in the Dispatch Deviation Guide, the procedural requirements can be found in the provisos of the MMEL or in the relevant Airplane Flight Manual (AFM).

NOTE: The (M) and (O) symbols are required in the operator's MEL unless otherwise authorized by Transport Canada.

4. **Alphabetical symbol** indicates a proviso (condition or limitation) that must be complied with for operation with the listed item inoperative.

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- 5. **NOTES** provide additional information for crewmember or maintenance consideration. Notes are used to identify applicable material which is intended to assist with compliance, but do not relieve the operator of the responsibility for compliance with all applicable requirements. Notes are not a part of the provisos.
- 6. **Deleted** in the remarks column after a sequence item indicates that the item was previously listed but is now required to be operative if installed in the aircraft.

ADDITIONAL DEFINITIONS

- 1. "Administrative Control Item" means an item listed by the operator in the MEL for tracking and information purposes. It may be added to an operator's MEL provided no relief is granted, or provided conditions and limitations are contained in an approved document (i.e., Structural Repair Manual, Airworthiness Directive, etc.). If relief other than that granted by an approved document is sought for an administrative control item, a request must be submitted to Transport Canada. If the request results in review and approval, the item becomes an MMEL item rather than an administrative control item.
- 2. "Airplane Flight Manual" (AFM) is the document required for type approval and approved by Transport Canada. The approved AFM for the specific aircraft is listed on the applicable Type Approval Data Sheet.
- 3. "As required by Regulations" may include such things as Canadian Aviation Regulations (CARs), both operational and design related; Aviation Occupational Health and Safety (AOH&S) regulations, etc. It is noted that detailed relief provided in the CARs is only applicable where a MEL is not required.
- 4. **"Change bar":** A vertical bar in the margin indicates a change, addition or deletion in the adjacent text for the current revision of that page only. The change bar is dropped at the next revision of that page.
- 5. **"Combustible materials**", for the purpose of this document, refers to materials which are capable of catching fire and burning (baggage, livestock, etc.).
 - Where loading of flammable or combustible material is prohibited, no material may be loaded except the following:
 - a) Cargo handling equipment (ballast or empty ULD). ULD must not contain any energy supplied device of any kind (e.g. ULD integrated trackers/monitoring devices, cooled containers), must be without oxygen supply, or any other flammable or combustible material,
 - b) Fly away kits (excluding e.g. cans of hydraulic fluid, cleaning solvents, batteries, capacitors, chemical generators, etc.),
 - c) In-flight service material (Return catering only closed catering trolleys/boxes, no newspapers, no alcohol or duty free goods).
- 6. **Day of Discovery** is the calendar day an equipment/instrument malfunction was recorded in the Aircraft Maintenance Logbook and/or Record. This day is excluded from the calendar days or flight days specified in the MMEL for the repair of an inoperative item of equipment, and is applicable to all MMEL items in categories A.B,C, and D.
- 7. **"Deactivated" and "Secured":** means that the specified component must be put into an acceptable condition for safe flight. An acceptable method of securing or deactivating will be established by the operator.

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ADDITIONAL DEFINITIONS (Cont'd)

- Engine Indicating Crew Alerting System (EICAS) that provides electronic messages refers to a system capable of providing different priority levels of system information messages (e.g., WARNING, CAUTION, ADVISORY and STATUS). Any airplane discrepancy message that affects dispatchability will be at the WARNING, CAUTION or STATUS level.
- 9. **"ER"** refers to extended range operations of a two-engine airplane which has a type design approval for ER operations and complies with the provisions of TP 6327 (ETOPS).
- 10. "Excess Items" means those items that have been installed but are redundant to the requirements.
- 11. **"Extended Overwater Operations":** means operations over water at a horizontal distance of more than 50 nautical miles from the nearest shoreline.
- 12. **"Federal Aviation Regulations" (FARs):** means the applicable portions of the Federal Aviation Act and Federal Aviation Regulations.
- 13. **"Flight Day"** means a 24-hour period (from midnight to midnight) either Universal Coordinated Time (UCT) or local time, as established by the operator, during which at least one flight is initiated for the affected aircraft.
- 14. "Heavy Maintenance Visit": means a scheduled C-check/D-check or airworthiness maintenance program inspection where the aircraft is scheduled to be out of service for 4 or more days.
- 15. "Icing Conditions": means an atmospheric environment that may cause ice to form on the aircraft or in the engine(s).
- 16. **"Inoperative"** means a system and/or a component malfunction to the extent that it does not accomplish its intended purpose and/or is not consistently functioning normally within its approved operating limit(s) or tolerance(s).
- 17. **Inoperative components of an inoperative system:** Inoperative items which are components of a system which is inoperative are usually considered components directly associated with and having no other function than to support that system. (Warning/caution systems associated with the inoperative system must be operative unless relief is specifically authorized per the MMEL).
- 18. "Passenger Convenience Items": means those items related to passenger convenience, comfort or entertainment such as, but not limited to, galley equipment, movie equipment, ash trays, stereo equipment, overhead reading lamps, etc.
- 19. **Placarding:** Each inoperative item must be placarded to inform and remind the crewmembers and maintenance personnel of the equipment condition.
 - **NOTE:** To the extent practical, placards should be located adjacent to the control or indicator for the item affected; however, unless otherwise specified, placard wording and location will be determined by the operator.

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ADDITIONAL DEFINITIONS (Cont'd)

- 20. "Reduced Vertical Separation Minimum" (RVSM) airspace: means any airspace or route where aircraft are separated by 1000 feet vertically between FL 290 and FL 410. RVSM Operations means operations conducted in RVSM airspace.
- 21. "Visual Flight Rules" (VFR): is as defined in the CARs. This precludes a pilot from filing an Instrument Flight Rules (IFR) flight plan.
- 22. "Visual Meteorological Conditions" (VMC): means the atmospheric environment is such that would allow a flight to proceed under the visual flight rules applicable to the flight. This does not preclude operating under Instrument Flight Rules.
- 23. **"Visible Moisture":** means an atmospheric environment containing water in any form that can be seen in natural or artificial light; for example, clouds, fog, rain, sleet, hail, or snow.
- 24. "Considered Inoperative" means that item must be treated for dispatch, taxi and flight purposes as though it were inoperative. The item shall not be used or operated until the deferred item is repaired. Additional actions include: documenting the item on the dispatch release (if applicable), placarding, and complying with all remarks, exceptions, and related MMEL provisions, including any (M) and (O) procedures and observing the repair category.
- 25. "Is not used" in the provisos, remarks or exceptions for an MMEL item may specify that another item relieved in the MMEL "is not used." In such cases, crewmembers should not activate, actuate, or otherwise utilize that component or system under normal operations. It is not necessary for the operators to accomplish the (M) procedures associated with the item. However, operational requirements must be complied with, and an additional placard must be affixed, to the extent practical, adjacent to the control or indicator for the item that is not used to inform crewmembers that a component or system is not to be used under normal operations.

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ABBREVIATIONS and ACRONYMS

The following abbreviations and acronyms may be used on flight compartment displays, radio tuning units and the flight management system or may be found in this manual. Some abbreviations may appear in upper or lower case letters. Abbreviations which have limited usage are explained in the chapters where they are used.

Α

A/C	Air Conditioning	AFT	Afterward
A/G	Air/Ground	AGI	Above Ground Level
A/ICE	Anti-ice	AHC	Attitude Heading
A/P	Autopilot	71110	Computer
A/S	Airspeed	AHRS	Attitude Heading
A/SKID	Anti-skid	-	Reference System
ABS	Absolute	AIL	Aileron
AC	Alternating Current	ALIGN	Aligning, alignment
ACARS	ARINC Communications	ALPHA	alpha
11011110	Addressing and	ALT	Altitude, Altimeter
	reporting System	ALT	Altitude Hold
ACCEL	Acceleration,		(PFD/FD)
	accelerate(d),	ALT CAP	Altitude Capture
	accelerometers		(PFD/FD)
ACM	Air Cycle Machine	ALT HOLD	Altitude Hold
ACMP	Alternating Current	ALTN	Alternate
	Motor Pump/ Electric	ALTS	Selected Altitude
	Hydraulic Pump		Arm/Abort (PFD/FD)
ACT	Active	AM	Amplitude Modulation
ACU	Air Conditioning Unit	AMB	Ambient
ADC	Air Data Computer	AMP	Amperes
ADDR	Address	ANNUN	Annunciator
ADF	Automatic Direction	ANO	Air Navigation Orders
	Finder	ANT	Antenna
ADG	Air Driven Generator	AOA	Angle of Attack
ADI	Attitude Director	AP	Autopilot
	Indicator	APC	Auxiliary Power
ADS	Air Data System		Control
AFCS	Automatic Flight	APP	Approach
7.77.0	Control System	APPROX	Approximately
AFIS	Airborne Flight	APR	Automatic Performance
7) 1711/1	Information System		Reserve
AFM	Airplane Flight Manual	APU	Auxiliary Power Unit
	ranual		

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Authority (UK) CAL Calibrate CDU Control Display Unit CAP Capture CAPT Captain CAS Calibrated Air Speed CAT Category CAT II Category CAT II Category II CB, C/B Circuit Breaker CHAN CHGR Control Display Unit CAPT Control Display Unit CAPT Cubic Feet Per Minut CAPT Canadian Federal Aviation Regulation CCG Center of Gravity CHAN Chapter, Channel CHAN Charger				
ARP Air Data Reference Panel AUTO BAL Automatic Balance Panel AUTO XFER Automatic Transfer AUTO XFER Automatic Transfer AUTO Auxiliary AUX Auxiliary AUX Auxiliary AUX AVX Auxiliary AUX ASSOCIATION ASSOCIATION AVAIL AVAILABLE AUXILIARY A	ARINC	Aeronautical Radio	ATTND	Attendant
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CAT Category CG Center of Gravity CAT II Category II CHAN Chapter, Channel CB, C/B Circuit Breaker CHAN Channel CBP Circuit Breaker Panel CHGR Charger		-		Aviation Regulation
CAT II Category II CH Chapter, Channel CB, C/B Circuit Breaker CHAN Channel CBP Circuit Breaker Panel CHGR Charger			CG	Center of Gravity
CB, C/B Circuit Breaker CHAN Channel CBP Circuit Breaker Panel CHGR Charger		J 1	СН	-
CBP Circuit Breaker Panel CHGR Charger				
•	•			
	CCW	Counter Clockwise		-

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CHR	Chronograph	CONN	Connection
CHRT	Chart	CONT	Control, Continuous,
CK	Check		Contactor, Controlle
CKPT	Cockpit	COOL	Cooling
CKT	Circuit	CORR	Correction
CLB	Climb	CPAM	Cabin Pressure
CLK	Clock		Acquisition Module
cm	Centimeters	CPDLC	Controlled-Pilot Dat
CMD	Command		Link Communication
CMPS	Compass	CPLT	Copilot
CMPTR	Computer	CRS	Course
CO ₂	Carbon Dioxide	CRT	Cathode Ray Tube
COM	Communication	CRZ	Cruise
COMM	Communication	CSD	Constant Speed Drive
COMP	Compressor,	CTR	Center
	Comparator	CVR	Cockpit Voice
COMPT	Compartment		Recorder
COND	Condition, Continued	CW	Clockwise
CONFIG	Configuration	CYL	Cylinder
	D		
DA	Drift Angle	DG	Directional Gyro
DBU	Data Base Unit	DH	Decision Height
DC	Direct Current	DIFF	Differential
DCP	Display Control Panel	DIM	Dimming
DCU	Data Concentrator	DIR	Direct
	Unit	DIS	Distance (to way
DECEL	Decelerate(d)		point), Disconnect
DECR	Decrease	DISC	Disconnect
DEFL	Defuel	DISCH	Discharge
DEG	Degree	DISP	Dispatch, Display
DEPR	Depressurize	DIST	Distance
DEPT	Departure	DME	Distance Measuring
DEST	Destination		Equipment
DET	Detector	DN	Down
DEV	Deviation	DOT	Department of
DFDAU	Digital Flight Data		Transport (Canada)
	Acquisition Unit	DR	Door
DFDR	Digital Flight Data Recorder	DUCT MON	Duct Monitor

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E						

	:	E	
EADI	Electronic Attitude	ELEC	Electrical
	Direction Indicator	ELEV	Elevator, Elevation
EAS	Equivalent Airspeed	ELT	Emergency Locator
ECAM	Electronic		Transmitter
	Centralized Aircraft	EMER(G)	Emergency
	Monitoring System	EMI	Electro-Magnetic
ECP	EICAS Control Panel		Interference
ECS	Environmental Control	ENG	Engine
	System	EPC	External Power
ECU	Electronic Control Unit		Contactor
ED	****	EQUIP	Equipment
ED	EICAS Display	ERP	Eye Reference
EDP	Engine Driven Pump/ Engine Primary		Position Datum
	Hydraulic Pump	ESS	Essential
EFB	Electronic Flight Bag	ET ——-	Elapsed Time
EFIS	Electronic Flight	ETA	Estimated Time of
што	Instrument System	200	Arrival
EGT	Exhaust Gas	ETE	Estimated Time Enroute
	Temperature	EVAC	Evacuation
EICAS	Engine Indication and	EVAC	Enhanced Vision
	Crew Alerting System	System	Emianced vision
EHSI	Electronic Horizontal	EXH	Exhaust
	Situation Indicator	EXTIN	Extinguish (ed)
EL	Elevation	2111111	incinguion (ea)
		F	
F/CTL	Flight Controls	FIRE BTL	Fire Bottle
FAA	Federal Aviation	FIREX	Fire Extinguisher
	Administration (USA)	FL CH	Flight Level Change
FAIL	Failure	FLD	Field
FCC	Flight Control	FLT	Flight
	Computer	FLT DIR	Flight Director
FCU	Fuel Control Unit	FLUOR	Fluorescent
FD, F/D	Flight Director	FM	Fan Marker
FDAU	Flight Data	FMS	Flight Management
	Acquisition Unit		System
FDR	Flight Data Recorder	FPM	Feet Per Minute
	(Digital)	FREQ	Frequency
FECU	Flaps Electronic	ft	Feet, Foot
	Control Unit	FW	Fire Wall
FEED	Feeder	FWD	Forward
FF, F/F	Fuel Flow		

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	G		
G (+/-)	Receiver Gain	GLD	Ground Lift Dumping
G/S	Glide slope	GMT	Greenwich Mean Time
GA	Go-around	GND	Ground
GAL	Gallon	GPM	Gallons Per Minute
GALY	Galley	GPWS	Ground Proximity
GCS	Ground Clutter		Warning System
	Suppression	GR	Gear
GCU	Generator Control	GRAV	Gravity
	Unit	GS	Ground Speed
GE	General Electric	GUIDE	Guidance
GEN	Generator	GW	Gross Weight
	н		
HDG	Heading	HOT	High Oil Temperature
HDG HOLD	Heading Hold	HP	High Pressure
HDG SEL	Heading Select	${ t HP}_{ t A}$	Hecto Pascals
HEAT	Heater	HSI	Horizontal Situation
HF	High Frequency		Indicator
	(3 - 30 mHz)	HSTA	Horizontal Stabilizer
Нд	Mercury		Trim
HGS	Head-up Guidance System	HSTCU	Horizontal Stabilizer Trim Control Unit
HI	- High	HTR	Heater
HIRF	High Intensity	HUD	Heads-up Display
	Radiated Fields	HYD	Hydraulic
HLDR	Holder	Hz	Hertz
HOR, HOP	RIZ Horizontal		
	I		
IB,I/B,I	INBD	ID	Identification
	Inboard	IDENT	Identification
I/C	Intercom, Inspection	IDG	Integrated Drive
	Check		Generator
IAPS	Integrated Avionics	IFIS	Integrated Flight
	Processor System		Information System
IAS	Indicated Air Speed	IFR	Instrument Flight
ICAO	International Civil		Rules
TCC	Aviation Organization		
ICS IGN	Idle Corrected Speed Ignition	ILS	Instrument Landing
TGIN	19111011	тпо	System

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L					
IM	ILS Inner Marker		INT	Internal,	_
IMC	Instrument		T11777 C	Intersection	on
	Meteorological		INTEG	Integral	
TMD	Conditions		IRS	Inertial Re	eference
IMP.	Imperial		TDII	System	
in.	Inch, Inches		IRU	Inertial Re	erence
in. Hg	Inches of Mercury		ISA	Internation	2.2.1
INCR	Increase		ISA	Standard A	
IND	Indication, Indica	ator	ISDU	Inertial S	_
INFLT	In Flight		1500	Display Uni	
INHIB	Inhibit		ISO	Internation	
INOP	Inoperative		100	Standard O	
INPH	Interphone		ISOL	Isolation,	
INSP	Inspection		ITT	Inter Turb	
INST(S)	<pre>Instrument(s)</pre>		111	Temperature	_
INST, INST				romporacur	
	Instrument				
		J			
JAA	Joint Aviation				
	Authority				
		ĸ			
K, KT, KTS	Knots		KIAS	Knots India	cated
kg	Kilogram(s)		11110	Airspeed	34334
kHz	KiloHertz		kW(s)	KiloWatt(s))
11112	KIIOHCI CZ		(- /		
		L			
L	Left, Landing		LCV	Load Contro	ol Valve
L/R	Left/Right		LDG	Landing	
L/T	Landing/Taxi		LDG GR	Landing Gea	ar
LAV	Lavatory		LDU	Lamp Drive	. Unit
lb	Pound(s)		LE	Leading Edg	ge
LCN	Load Classification	on	LED	Light Emit	_
LDS	Lightning Detection	on	LK	Leak	,
	System		LN	Left Nose	
LG	Landing Gear		LNAV	Lateral Nav	rigation
LGC	Landing Gear		LOC	ILS Localiz	-
	Controller		LOGO	Logo Graphi	
LGW	Landing Gross Weig	ght	LOM, MM	Compass Loc	
LH	Left Hand		TO11, 1.11.1	Outer Marke	
LIM	Limit		LOP	Low Oil Pre	

MSL Mean Sea Level

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LP	Low Pressure		LT(s)	Light(s)	
LPM	Liter Per Minute		LTGWAIS	_	ature Grou
			T 1-7	_	ice System
LR	Left Rear		TM	Left Wing	Dorra
LRC	Long Range Cruise		LWD	Left Wing	DOMU
LRU	Line Replaceable	Unit	LWR	Lower	
LSB	Lower Side Band				
		M			
М	Mach Number		MEL	Minimum Ec	quipment
m	Meter			List	
MAA	Maximum Authorize	ed	MFD	Multifunct	tion Displa
	IFR Altitude		MGP	MLS Glides	slope
MAC	Mean Aerodynamic		MHz	MegaHertz	
	Chord		MI	Miles	
MAG	Magnetic		MIC	Microphone	2
MAINT	Maintenance		MID AFT	Middle Afterward	
MALF	Malfunction		MID FWD	Middle For	rward
MAN	Manual		MILS	.001 of ar	inch
MAP	Ground Map (WXR)		MIN	Minimum	
MASI	Mach Air Speed		MISC	Miscellane	eous
	Indicator		MKR	Marker	
MAX	Maximum		MLG	Main Landi	ng Gear
MAZ	MLS Azimuth		MLI	Magnetic I	Level
MB	Millibars			Indicator	
MCA	Minimum Crossing Altitude		MLU	Midlife Up	odate
MCT	Maximum Continuou Thrust	ıs	MLW	Maximum La Weight	anding
MDA	Minimum Descent		MM	ILS Middle	e Marker
	Altitude		MMEL	Master Mir	nimum
MEA	Minimum Enroute I	FR		Equipment	List
	Altitude		M_{MO}	Maximum Op	
MECH	Mechanic				Mach Number
MED	Medium				
MOCA	Minimum Obstructi Clearance Altitud		MTBF	Mean Time Failure	Before
MOD	Module		MTG	Miles to 0	Go
MON	Monitor		MTOW	Maximum Ta	akeoff
MPH	Miles Per Hour			Weight	
MRA	Minimum Reception	n	MTW	Maximum Ta	axi Weight
	Altitude		MZFW	Maximum Ze	ero Fuel
MSG	Message			Weight	
MOT	- M O T 1				

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	N	Misc		
90	Percent		°C	Degrees Centigrade
&	and		°F	Degrees Fahrenheit
		N		
N/A	Not applicable		NEUT	Neutral
N1	Low Pressure Rotor		NL	Nose Left
N2	High Pressure Rotor		NLG	Nose Landing Gear
NAV	Navigation		NM	Nautical Mile(s)
ND	Nose Down, Navigation		No.	Number
	Display		NOPT	No Procedure Turn
NDB (ADF)	Nondirectional Beacon			Required
	(Automatic Direction		NORM	Normal
	Finder)		NOSE	Nosewheel
NEG	Negative		NR	Nose Right
NEF	Non-Essential		NU	Nose Up
	Equipment and Furnishings			
		0		
OAT	Outside Air		OK	Okay
	Temperature		OM	ILS Outer Marker
OB/OUTBD	Outboard		OVBD	Overboard
OBS	Observer		OVHT, OH	Overheat
OEI	One Engine		OVLD	Overload
OPH	Inoperative		OVSP	Overspeed
OEW	Operating Empty Weight		OVSPD	Overspeed
OH, OVHD	Overhead		OVTEMP	Over Temperature
OII, OVIID	Overnead		OXY, 02	Oxygen
		P		
P#6	Panel 6		PDU	Power Drive Unit
P/S	Pitot/Static		PF	Pilot Flying
PA	Passenger Address		PFD	Primary Flight
PASS	Passenger			Display
PBE	Portable Breathing		PLA	Power Lever Angle
	Equipment (Smoke		PLT(s)	Pilot(s)
	Hood)		PNF	Pilot Not Flying
PCU	Power Control Unit		PNLS(s)	Panel(s)

MINISTÈRE DES TRANSPORTS LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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D.O.			DOT	D 1 D	0
PO POS	Outside Air Pressu	ıre	PSI	Pounds Per Inch	Square
PPH	Pounds Per Hour		PSIG	Pounds Per	Square
PRESS	Pressure,			Inch Gauge	_
	Pressurization		PSS	Proximity	Sensor
PRI	Primary			System	
PRIM	Primary		PSU	Passenger	Service
PROC	Procedure			Unit	
PROT	Protection		PT2	Engine Inl	et Pressui
PROX	Proximity		PTCT	Protect	7 1
PSEU	Proximity Sensor		PTT	Push To Ta	l K
	Electronics Unit		PWR	Power	
		Q			
QAR	Quick Access Reco	rder	QNH	Altimeter	Setting
QEC	Quick Engine Chang	ge	QTY	Quantity	
QFE	Local Station	-		_	
	Pressure				
		R			
R	Right		RMI	Radio Magn	etic
RA	Radio Altitude			Indicator	
	Resolution Advisor	-Y	RMU	Radio Mana	gement Un:
RAI	Registro Aeronaut	ico			
	Italiano (Italy)		ROT	Rotation	
RAT	Ram Air Turine		RPM	Revolution	s Per
RCCB	Remote Controlled		,	Minute	
	Circuit Breaker		RT, R/T	Receiver-T	ransmitte
RCDR	Recorder		RTE	Route	
RCVR	Receiver		RTE DATA	Route Data	
RDR	Radar		RTO	Rejected T	
REC	Receiver, Recorder	<u>-</u>	RTU	Radio Tuni	ng Unit
RECOG	Recognition		RUD	Rudder	
REF(s)	Reference(s)		RVR	Runway Vis	_
REFL	Refuel		RVSM	Reduced Ve Separation	
REV	Reverse		RVSR	Reverser	PITITIUUII
RH	Right Hand		RV5R RW	Reverser Right Wing	
RIPS	Recorder Independe	ent		Right Wing Right Wing	
	Power Supply		RWD	DIGHT WITE	1)01.10

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s

S	Status	_	SPKR	Speaker
SAT	Static Air		SPLR(s)	Speaker Spoiler(s)
DAI	Temperature		SQL	Squelch
SCAT	Speed Command		SR/SL	Smart Runway /
2 0111	Attitude Thrust		SI(/ SII	Smart Landing
SCAV	Scavenge		SSB	Single Side Band
SDS	Status Display System		STA	Station
SEC	Second, Secondary		STAB	Stabilizer
SECS	Spoiler Electronic		STAT	Status
	Control System		STBY	Standby
SECU	Spoiler Electronic		STEER	Steering
	Control Unit		SUPPL	Supply
SEL	Select, Selector		SW(s)	Switches
SEL CAL	Selective Call		SYN	Synchronize
SENS	Sensitivity, Sensor		SYNC	Synchronous
SERV, SVCE			SYS, SYST	System
SMKG	Smoking		SVS	Synthetic Vision
SOV	Shutoff Valve			System
SP, SPD	Speed			-
		T		
T/C	Top of Climb		TE	Trailing Edge
T/D	Top of Descent		TEMP	Temperature
T/R	Thrust Reverser		TGT	Target
TA	Traffic Alert		TO, T/O	Takeoff
TACAN	UHF Tactical Air		TOC	Top of Climb
	Navigation Aid		TOD	Top of Descent
TAS	True Airspeed		TOL	Tolerance
TAT	Total Air		TRB, TURB	Turbulence
	Temperature		TRK	Track
TAWS	Terrain Awareness		TRM	Trim
	and Warning System		TRU	Transformer
TCAG	Transport Canada			Rectifier
ELC A. C.	Airworthiness Group			Unit
TCAS	Traffic Alert and Collision Avoidance		TT2	Engine Inlet
	System			Temperature

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		υ		
UHF UNSCHD USB	Ultra High Frequency Unscheduled Upper Side Band		USG UTIL	United States Gallons Utility
		v		
V	Volt		V_S	Stalling Speed or the
V_A	Design Maneuvering Speed			Minimum Steady Flight Speed at which the
V_B	Design Speed for Maximum Gust			Airplane is Controllable
V_C	Intensity Design Cruising Speed		V_{S0}	Stalling Speed or the Minimum Steady Flight
${ m V}_D$ ${ m V}_{DF}/{ m M}_{DF}$	Design Diving Speed Demonstrated flight			Speed in the Landing Configuration
	diving speed.		$V_{\mathcal{S}1}$	Stalling Speed or the Minimum Steady Flight
$egin{array}{l} egin{array}{l} Design Flap Speed Maximum Speed for			Speed Obtained in a	
V DF/ FIFC	Stability Characteristics		V_X	SpecificConfiguration Speed for Best Angle
$V_{\it FE}$	Maximum Flap Extended Speed		V_Y	of Climb Speed for Best Rate of Climb
V_H	Maximum Speed in Level Flight with Maximum Continuous Power		V ₁	Takeoff Decision Speed (formerly Denoted as Critical
V_{LE}	Maximum Landing Gear Extended Speed		V_2	Engine Failure Speed) Takeoff Safety Speed
V_{LO}	Maximum Landing Gear Operating Speed		V _{2 MIN}	Minimum Takeoff Safety Speed
V_{LOF}	Lift-off Speed			
V_{MC}	Minimum Control Speed with the Critical Engine Inoperative		V/S VERT	Vertical Speed Vertical
V_{MO}/M_{MO}	Maximum Operating Limit Speed		VFR VG	Visual Flight Rules Vertical Gyro
V_{MU}	Minimum Unstick Speed		VHF	Very High Frequency (30 - 300 mHz)
V_{NE}	Never-exceed Speed		VIB	Vibration
V_{NO}	Maximum Structural Cruising Speed		VMC	Visual Meteorological
V_R	Rotation Speed			Conditions

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VNAV VOL	Vertical Navigation	on	VORTAC	VOR and TAG	CAN	
VOLT VOR	Voltage VHF Omnidirections Range Station	al	VSI	Vertical S _l Indicator	peed	
		W				
W	Warning		WIND	Window		
W/C	Wind Component		WOW	Weight-On-N		
W/S	Wind Shear		WPT(s)	Waypoint(s)	
W/W	Wheel Well		WRN	Warning		
WARN	Warning	_	WS	Second Segr		
WF	Runway Length Lim:	ited	MOHED	Limited We:	ignt	
WGT	Weight Weight		WSHLD WX	Windshield Weather		
WHLS	Wheels		WXR	Weather Rad	dar	
		x				
X	Cross Transfer		XPNDR	Transponde:	r	
XFER, XFR	Transfer		XTK	Cross Trac	k	
XFLOW	Cross Flow		XWC	Cross Wind	Component	
XMIT	Transmit					
		Y				
YD, Y/D	Yaw Damper					
		Z				
ZFW	Zero Fuel Weight					

MINISTÈRE DES TRANSPORTS LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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PREAMBLE

All equipment installed on an aircraft in compliance with the Airworthiness Standards and Operating Rules must be operative. However, Canadian Aviation Regulations (605.07, 704.07 and 705.07) permit the publication of a Minimum Equipment List (MEL) where compliance with certain equipment requirements is not necessary in the interests of safety under all operating conditions. Experience has shown that with the various levels of redundancy designed into aircraft, operation of every system or installed component may not be necessary when the remaining operative component can provide the required level of safety.

A Master Minimum Equipment List (MMEL) is developed by Transport Canada, with participation by the aviation industry, to improve aircraft utilization and thereby provide more convenient and economic air transportation for the public. The approved MMEL includes those items of equipment related to airworthiness and operating regulations and other items of equipment Transport Canada finds may be inoperative and yet maintain the required level of safety by appropriate conditions and limitations; it does not contain obviously required items such as wings, flaps, and rudders.

The MMEL is the basis for development of individual operator MELs which take into consideration the operator's particular aircraft equipment configuration and operational conditions. Operator MELs, for administrative control, may include items not contained in the MMEL; however, relief for administrative control items must be approved. An operator's MEL may differ in format from the MMEL, but cannot be less restrictive than the MMEL. The individual operator's MEL, when approved and authorized, permits operation of the aircraft with inoperative equipment.

Equipment not required by the operation being conducted and equipment in excess of the requirements are included in the MEL with appropriate conditions and limitations. The MEL must not deviate from the Aircraft Flight Manual Limitations, Emergency Procedures or with Airworthiness Directives. It is important to remember that all equipment related to the airworthiness and the operating regulations of the aircraft not listed on the MMEL must be operative.

Suitable conditions and limitations in the form of placards, maintenance procedures, crew operating procedures and other restrictions as necessary are specified in the MEL to ensure that the required level of safety is maintained.

The MEL is intended to permit operation with inoperative items of equipment for a period of time until repairs can be accomplished. It is important that repairs be accomplished at the earliest opportunity. In order to maintain the required level of safety and reliability, the MMEL establishes limitations on the duration of and conditions for operation with inoperative equipment. The MEL provides for release of the aircraft for flight with inoperative equipment. When an item of equipment is discovered to be inoperative, it is reported by making an entry in the Aircraft Maintenance Record/Logbook. The item is then either repaired or may be deferred per the MEL or other approved means acceptable to Transport Canada prior to further operation. MEL conditions and limitations do not relieve the operator from determining that the aircraft is in condition for safe operation with items of equipment inoperative.

Operators are responsible for exercising the necessary operational control to ensure that the required level of safety is maintained. When operating with multiple inoperative items, the interrelationships between those items and the effect on aircraft Operation and crew workload must be considered.

MINISTÈRE DES TRANSPORTS LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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PREAMBLE

(Cont'd)

Operators are to establish a controlled and sound repair program including the parts, personnel, facilities, procedures, and schedules to ensure timely repair.

WHEN USING THE MEL, COMPLIANCE WITH THE STATED INTENT OF THE PREAMBLE, DEFINITIONS, AND THE CONDITIONS AND LIMITATIONS SPECIFIED IN THE MEL IS REQUIRED.

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Svstem	& Sequence No Item	1.	2.		er Installed ere d'articles installés	
-	système/série article			3.	Number Required For Dispatch	
	•				Nombre d'articles à expédier	
21 – <u>AIF</u>	R CONDITIONING				4. Remarks or Exceptions Remarques ou exceptions	
20-1	Heat Exchanger Fan (Footwarmer- Demist) (600, 601, 601-3A)	С	1	0		
24-1	Cockpit Displays Cooling Exhaust Fan	С	1	0	(O) May be inoperative provided both ACUs are operative.	1
	(604, 605, 650)				NOTE: During ground operations at ambient temperatures above 40°C (104°F), operation of electrical/avionics equipment must be limited to 30 minutes unless at least one ACU is operating and cabin doors are closed.	
30-1	Emergency Pressurization System (600, 601, 601-3A)	С	1	0	(O) May be inoperative provided aircraft is operated unpressurized.	
30-2	Cabin Altitude Warning System (600, 601, 601-3A, 601-3R)					
	1) Visual (CABIN PRESS LOW light)	С	1	0	(O) May be inoperative provided Cabin Altitude Aural Warning System is operative.	
		С	1	0	(O) May be inoperative provided aircraft is operated at or below 10,000 feet MSL.	
	2) Aural (Horn) ***	D	1	0	(O) May be inoperative provided Cabin Altitude Visual Warning System is operative.	
		С	1	0	May be inoperative provide aircraft is operated at or below 10,000 feet MSL.	
					Deleted, revision 13.	١

DEPARTMENT OF TRANSPORT

MINISTÈRE DES TRANSPORTS MASTER MINIMUM EQUIPMENT LIST LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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1.		2.		er Installed	
System & Sequence No Item				ore d'articles installés	
Nº de système/série article			3.	Number Required For Dispatch	
21 _ ΔΙ	R CONDITIONING				Nombre d'articles à expédier 4. Remarks or Exceptions
21 <u>Al</u>	IN CONDITIONING				Remarques ou exceptions
30-3	Cabin Altitude	С	1	0	(O) May be inoperative provided:
	Indicator (600, 601, 601-3A, 601-3R)				 a) Cabin Differential Pressure Indicator is operative, and
					 b) A chart is provided to convert cabin differential pressure to cabin altitude.
		С	1	0	(O) May be inoperative provided aircraft is operated unpressurized.
30-4	Cabin Differential	С	1	0	(O) May be inoperative provided:
	Pressure Indicator				a) Cabin Altitude Indicator is operative, and
	(600, 601,601-3A, 601-3R)				 b) A chart is provided to convert cabin altitude to cabin differential pressure.
		С	1	0	(O) May be inoperative provided aircraft is operated unpressurized.
30-5	Cabin Rate of Climb	С	1	0	May be inoperative provided:
	Indicator				a) Cabin Altitude Indicator is operative,
	(600, 601, 601-3A, 601-3R)				b) Cabin Differential Pressure Indicator is
	,				operative, and
					c) Cabin Pressurization Control System Automatic Mode is operative.
		С	1	0	(O) May be inoperative provided aircraft is operated unpressurized.

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System	& Sequence Nº Item	1.	2.		per Installed pre d'articles installés
-	ystème/série article			3.	Number Required For Dispatch
	<u>, , , , , , , , , , , , , , , , , , , </u>				Nombre d'articles à expédier
21 – <u>AIR</u>	R CONDITIONING				4. Remarks or Exceptions Remarques ou exceptions
30-6	Cabin Pressurization Panel EMER DEPRESS "ON" Switch Light (light function only) (604, 605, 650)	С	1	0	(O) May be inoperative provided the EMER DEPRESS caution message is verified operative.
31-1	Cabin Pressurization Control System				
	1) Automatic and Manual Modes	С	2	0	 (O) Both may be inoperative provided: a) Aircraft is operated unpressurized, b) Aircraft is not operated in extended overwater operations, and c) Cargo compartments are empty or do not contain combustible materials. NOTE 1: (604, 605, 650) During ground operations at ambient temperatures above 40°C (104°F), operation of electrical/avionics equipment must be limited to thirty minutes unless at least one ACU/Pack is operating and cabin doors are closed as per AFM. NOTE 2: (600, 601, 601-3A, 601-3R) During ground operations at ambient temperatures above 45°C (113°F), operation of electrical/avionics equipment must be limited to thirty minutes unless at least one ACU/Pack is operating and cabin doors are closed as per AFM.
	(Cont'd)				

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System & Sequence No Item		1.	2.	Number Installed Nombre d'articles installés			
No de système/série article				3. Number Required For Dispatch			
in- ue systeme/sene diticle				•	Nombre d'articles à expédier		
21 – <u>Alf</u>	R CONDITIONING				4. Remarks or Exceptions Remarques ou exceptions		
31-1	Cabin Pressurization Control System (Cont'd)						
	2) Automatic Mode	С	1	0	(O) May be inoperative provided:		
	(600, 601, 601- 3A, 601-3R)				 a) Cabin Pressurization Control System Manual Mode is operative, 		
					b) Autopilot is operative, and		
					c) Aircraft is not operated more than 60 minutes from a suitable airport.		
		С	1	0	(O) May be inoperative provided:		
					a) Total fuel carried includes at least 50% more than the fuel load required for the planned flight, and		
					b) Aircraft speed is limited to 300 KIAS when at or below 12,500 feet.		
	3) Automatic Mode	С	1	0	May be inoperative provided:		
	(604, 605, 650)				a) Cabin Pressurization Control System Manual Mode is operative,		
					b) Autopilot is operative, and		
					c) Aircraft is not operated more than 60 minutes from a suitable airport.		
	4) Manual Mode	С	1	0	May be inoperative provided:		
					a) Cabin Pressurization Control System Automatic Mode is operative,		
					b) Aircraft is not operated more than 60 minutes from a suitable airport, and		
					 c) Aircraft is not operated in extended overwater operations. 		

DEPARTMENT OF TRANSPORT

MINISTÈRE DES TRANSPORTS MASTER MINIMUM EQUIPMENT LIST LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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		1.	2.		per Installed
	n & Sequence No Item				ore d'articles installés
Nº de s	système/série article			3.	Number Required For Dispatch
	D COMPLETION IN C				Nombre d'articles à expédier
21 – <u>Al</u>	R CONDITIONING				4. Remarks or Exceptions Remarques ou exceptions
32-1	Outflow	С	2	0	(M)(O) One or both may be inoperative provided:
	Valve/Safety Valves				 a) Affected Valve(s) are secured OPEN,
					 b) Airplane is operated unpressurized at or below 10,000 feet MSL,
					 c) Extended overwater operations are prohibited, and,
					d) Takeoffs and landings must not be conducted on runways that may lead to imminent ditching.
33-1	Cabin Pressure Acquisition Module (CPAM)				
	1) (604)	С	1	0	(O) May be inoperative provided:
	1) (004)	C	'	0	
					 a) Aircraft is operated unpressurized, and b) Aircraft is not operated in extended overwater operations.
	2) (605, 650)	С	2	1	May be inoperative provided:
					a) Automatic Mode of the Pressurization Control System is operative, and
					b) Aircraft is not operated more than 60 minutes from a suitable airport.
		С	2	0	(O) May be inoperative provided: a) Aircraft is operated unpressurized, and
					 a) Aircraft is operated unpressurized, and b) Aircraft is not operated in extended overwater operations.

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		1.	2.		er Installed
Systen	n & Sequence Nº Item				re d'articles installés
Nº de	système/série article			3.	Number Required For Dispatch
04 41	D COMPLETIONING				Nombre d'articles à expédier
21 – <u>Al</u>	R CONDITIONING				4. Remarks or Exceptions Remarques ou exceptions
33-2	Standby Cabin Altimeter (604)	С	1	0	(O) May be inoperative provided flight planning allows for descent to unpressurized flight altitudes in the event of a CPAM failure.
51-1	Air Conditioning	С	2	1	One may be inoperative provided:
	Units (ACUs)				a) Ram Air Vent Valve is operative,
	(600, 601, 601-3A, 601-3R)				b) Associated cowl anti-ice is selected ON when aircraft is operated above FL 400, and
					c) Aircraft is not operated more than 60 minutes from a suitable airport.
		С	2	1	One may be inoperative provided:
					a) Total fuel carried includes at least 50% more than the fuel load required for the planned flight; and
					b) Aircraft speed is limited to 300 KIAS when at or below 12,500 feet.
		С	2	0	(O) Both may be inoperative provided:
					a) Ram Air Vent Valve is operative,
					b) Aircraft is operated unpressurized, and
					c) Cargo compartments are empty or do not contain combustible materials.
					NOTE: During ground operations at ambient temperatures above 45°C (113°F) operation of electrical / avionics equipment must be limited to 30 minutes unless at least one ACU is operating and cabin doors are closed as per AFM.

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System	2 Coguenes NO Item	1.	2.		per Installed pre d'articles installés
_	& Sequence Nº Item ystème/série article			3.	Number Required For Dispatch
111 40 0	yotomoroono untiolo				Nombre d'articles à expédier
21 – <u>AIF</u>	R CONDITIONING				4. Remarks or Exceptions Remarques ou exceptions
51-2	Air Conditioning Packs (604, 605, 650)	С	2	1	One may be inoperative provided: a) Ram Air Vent Valve is operative, b) Associated cowl anti-ice is selected ON when aircraft is operated above FL 400, and c) Aircraft is not operated more than 60 minutes from a suitable airport.
		С	2	0	 (O) Both may be inoperative provided: a) Ram Air Vent Valve is operative, and b) Aircraft is operated unpressurized. NOTE: During ground operations at ambient temperatures above 40°C, (104°F) operation of electrical / avionics equipment must be limited to 30 minutes unless at least one pack is operating and cabin doors are closed as per AFM.
51-3	Air Conditioning Pack "L/R FAIL" Switchlight (light function only) (604, 605, 650)	С	2	0	
51-4	Cabin Pressurization Panel "FAIL/MANUAL" Switchlight (light function only) (604, 605, 650)	В	1	0	(O) May be inoperative provided the manual and automatic mode are verified operative on EICAS.

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_	a & Sequence Nº Item système/série article			3.	Number Required For Dispatch
					Nombre d'articles à expédier
21 – <u>All</u>	R CONDITIONING				4. Remarks or Exceptions Remarques ou exceptions
52-1	Ram Air Vent Valve	С	1	0	(M) May be inoperative closed provided:a) Valve is deactivated closed, andb) Both ACUs / Packs are operative.
		С	1	0	 (M) May be inoperative open provided: a) Valve is deactivated open, b) RH ACU/Pack is operative, c) LH ACU/Pack is selected OFF, d) Associated cowl anti-ice is selected ON when aircraft is operated above FL 400, and e) Aircraft is not operated more than 60 minutes from a suitable airport.
53-1	Right Footwarmer De-Mist Pressure Regulating and Shut-Off Valve (600, 601, 601-3A)	С	1	0	
61-1	Temperature Control System				
	1) Cockpit Modes (NORMAL, STANDBY, MANUAL)	С	3	1	
		С	3	0	Any or all may be inoperative provided all Cabin modes are operative.

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_	ystème/série article			3.	Number Required For Dispatch
11 0.00	,				Nombre d'articles à expédier
21 – <u>AIF</u>	R CONDITIONING				4. Remarks or Exceptions Remarques ou exceptions
61-1	Temperature Control System (Cont'd)				
	2) Cabin Modes (NORMAL, STANDBY, MANUAL)	С	3	1	
		С	3	0	Any or all may be inoperative provided all Cockpit Modes are operative.
61-2	Cabin Temperature Indicator (600, 601, 601-3A, 601-3R)	С	1	0	
61-3	EICAS "CABIN TEMP" Indication (604, 605, 650)	С	1	0	
61-4	Cockpit Heat System (601-3R, 604)	С	1	0	(M) May be inoperative provided the system is deactivated.
61-5	Pilots Heated Mats (605, 650)	С	4	0	(O) Any or all may be inoperative provided heated mats are deactivated.

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21 – <u>AIR CONDITIONING</u>				4. Remarks or Exceptions Remarques ou exceptions
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Nº de système/série article			3.	Number Required For Dispatch
				Nombre d'articles à expédier
22 – <u>AUTO FLIGHT</u>				4. Remarks or Exceptions Remarques ou exceptions
10-1 Autopilot System				
1) 600, 601	В	1	0	(M)(O) Except when enroute operations or approach procedures require its use, may be inoperative provided autopilot is deactivated after Mach Trim test is verified operative each flight day.
a) Roll axis channel	С	2	1	(O) May be inoperative provided the opposite roll channel and roll monitor is verified operative prior to each departure.
b) Pitch axis channel	С	2	1	(O) May be inoperative provided the opposite pitch channel is verified operative prior to each departure.
2) 601-3A, 601-3R (AFCS)	С	2	1	(M) Except where enroute operations or approach procedures require dual autopilot systems, may be inoperative provided system is deactivated.
				NOTE: Autopilot and Transponder must use same ADC data for RVSM Operations.
	В	2	0	(M) Except when enroute operations or approach procedures require its use both may be inoperative.
3) 604, 605, 650 (AFCS)	С	2	1	Except where enroute operations or approach procedures require dual autopilot systems, may be inoperative provided alternate AFCS is selected.
				NOTE: Autopilot and Transponder must use same ADC data for RVSM Operations.
	В	2	0	(O) Except when enroute operations or approach procedures require its use both may be inoperative.
a) VNAV Mode Switch	С	1	0	May be inoperative provided enroute operations or approach procedures are not dependent on its use.

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_	ystème/série article			3.	Number Required For Dispatch
					Nombre d'articles à expédier
22 – <u>AU</u>	TO FLIGHT				4. Remarks or Exceptions Remarques ou exceptions
10-2	Flight Directors				
	1) 600, 601, 601-3A, 601-3R	С	2	1	(O) May be inoperative provided approach minimums or operating procedures are not dependent on its use.
					NOTE 1 : Flight Director and Transponder must use same ADC data for RVSM Operations.
					NOTE 2 : Associated TOGA switch will be inoperative.
		В	2	0	(O) Except where enroute operations or approach procedures require its use, both may be inoperative.
					NOTE 1: Associated TOGA switches will be inoperative.
					NOTE 2: Autopilot will be inoperative.
	2) 604 (FCC channels)	С	4	3	
					NOTE: Associated autopilot (AFCS 1 or 2) will be inoperative.
		С	4	2	(O) Two FCC channels may be inoperative provided AFCS 1 or 2 is operative.
		В	4	1	(O) Except for RVSM Operations, may be inoperative provided:
					a) AFCS 1 and 2 are considered inoperative,
					b) One Yaw Damper is verified operative, andc) Approach minimums or operating procedures
					are not dependent on their use.

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22 – <u>AL</u>	<u>JTO FLIGHT</u>				4. Remarks or Exceptions Remarques ou exceptions
10-2	Flight Directors (Cont'd)				
	3) 605, 650 (FCC channels)	С	4	3	(O) Except where enroute operations or approach procedures require its use, one may be inoperative provided associated autopilot (AFCS 1 or 2) is considered inoperative.
					NOTE: For RNP AR approaches, refer to AFM chapter 07, supplement 18, RNP Authorization Required Approach Operations for required equipment.
		С	4	2	(O) Except where enroute operations or approach procedures require its use, two may be inoperative provided AFCS 1 or 2 is operative.
					NOTE: For RNP AR approaches, refer to AFM chapter 07, supplement 18, RNP Authorization Required Approach Operations for required equipment.
		В	4	1	(O) Except where enroute operations or approach procedures require its use, three may be inoperative provided:
					a) AFCS 1 and 2 are considered inoperative,
					b) One Yaw Damper is verified operative, and
					 c) Approach minimums or operating procedures are not dependent on their use.
					NOTE: For RNP AR approaches, refer to AFM chapter 07, supplement 18, RNP Authorization Required Approach Operations for required equipment.

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				3.	Number Required For Dispatch
22 – <u>AL</u>	JTO FLIGHT				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
11-1	Autopilot Disconnect Switches (Control Wheel Buttons)	С	2	1	 One may be inoperative provided: a) Pusher of the stall protection system can be disengaged using the control wheel disconnect button, b) With autopilot engaged the pilot controlling the aircraft has an operative disconnect switch, and c) Autopilot is not engaged below 1500 feet AGL.
		В	2	0	Both may be inoperative provided: a) Pusher of the stall protection system can be disengaged using both control wheel disconnect buttons, and b) Autopilot system is not used.
11-2	Touch Control Steering (TCS) Switches (600, 601, 601-3A, 601-3R)	С	2	0	
11-3	AP/FD Sync Switches (604, 605, 650)	С	2	0	
11-4	Take-Off/Go-Around (TOGA) Switches	С	2	0	(O) May be inoperative provided procedures are established and used.
11-5	Flight Guidance Computers (601-3A, 601-3R)	С	2	1	One may be inoperative provided aircraft is not operated more than 60 minutes from a suitable airport. NOTE: Each FGC provides Mach Trim, Yaw Damper, Autopilot and Flight Director functions independent of the other FGC. Failure of the second FGC after dispatch will result in the complete loss of these functions and AFM restrictions will apply.

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No de s	système/série article			3.	Number Required For Dispatch
22 – <u>AL</u>	JTO FLIGHT				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
12-1	V-Speed Auto- Synchronization System (604, 605, 650)	С	1	0	May be inoperative provided V Speed settings are made manually by each pilot. NOTE: "SPEED REFS INDEP" status message will be displayed on EICAS if selected speeds are different.
13-1	Input/Output Concentrator (IOC) (Integrated Avionics Processing System) (604, 605, 650)				
	1) (604 (without Autothrottle System)	С	4	3	 (M)(O) One IAPS IOC may be inoperative provided: a) Remaining IOCs are verified operative before the first flight of the day, b) RTU 2 is operative, and c) Lightning Detection System (if installed) is considered inoperative if the failed IOC is the 2A quadrant.
					NOTE: "IAPS DEGRADED" status message will be displayed on EICAS.
	2) (604 with Autothrottle System)	С	4	3	 (M)(O) One IAPS IOC may be inoperative provided: a) Remaining IOCs are verified operative before the first flight of the day, b) RTU 2 is operative, c) Lightning Detection System (if installed) is considered inoperative if the failed IOC is the 2A quadrant, and d) The Autothrottle system is considered inoperative if the failed IOC is the 1A quadrant. NOTE: "IAPS DEGRADED" status message will be displayed on EICAS.
	(Cont'd)				

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22 41	ITO ELICUT				Nombre d'articles à expédier 4. Remarks or Exceptions
22 – <u>AC</u>	JTO FLIGHT				4. Remarks or Exceptions Remarques ou exceptions
13-1	Input/Output Concentrator (IOC) (Integrated Avionics Processing System) (604, 605, 650) (Cont'd)				
	3) (605, 650)	С	4	3	(M)(O) One IAPS IOC may be inoperative provided:
					Remaining IOCs are verified operative before the first flight of the day,
					b) CDU 2 radio tuning function is operative,
					c) The Autothrottle system is considered inoperative if the failed IOC is the 1A quadrant, and
					d) Lightning Detection System (if installed) is considered inoperative if the failed IOC is the 2A quadrant.
					e) The Integrated Flight Information System (IFIS) File Server Unit (FSU) is considered inoperative if the failed IOC is the 1A quadrant.
					NOTE: "IAPS DEGRADED" status message will be displayed on EICAS.
20-1	Mach Trim System	С	1	0	May be inoperative provided autopilot is engaged in flight.
		С	1	0	May be inoperative provided airspeed is limited to 250 KIAS/0.7 Mach when the autopilot is not engaged.
					NOTE: Autopilot must be engaged during RVSM Operations.

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22 – <u>AU I</u>	<u>O FLIGHT</u>				Remarques ou exceptions	
20-2	Yaw Damper Channels 1) 600, 601, 601-	С	2	1	(O) One channel may be inoperative provided the	
	3A, 601-3R, 604,605 <u>without</u> SB 605-34-021, and 650 <u>without</u> SB 650-34-009				rudder is verified in the central position before flight.	
	2) 605 <u>with</u> SB 605-34-021, and 650 <u>with</u> SB 650-34-009	С	2	1	(O) Except where enroute operations or approach procedures require its use, one channel may be inoperative provided the rudder is verified in the central position before flight.	
					NOTE: For RNP AR approaches, refer to AFM chapter 07, supplement 18, RNP Authorization Required Approach Operations for required equipment.	
20-3	Yaw Rate Gyros (600, 601, 601-3A, 601-3R)	С	2	1	(M) One may be inoperative provided Standby Altitude Indicator is operative.	
30-1	Autothrottle System (604***, 605, 650)	С	1	0	(O) May be inoperative provided the autothrottle system is deactivated.	I
		D	1	0	(O) May be inoperative provided:	1
					a) The autothrottle system is deactivated, andb) Procedures do not require its use.	I
	1) Autothrottle Disconnect Pushbutton Switch (On Thrust Lever) (605, 650)	С	2	1	(O) May be inoperative provided the Engage/Disengage Pushbutton Switch is verified operative.	
		С	2	0	May be inoperative provided the Autothrottle System is considered inoperative.	
	(Cont'd)					

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22 – <u>Al</u>	UTO FLIGHT				4. Remarks or Exceptions Remarques ou exceptions
30-1	Autothrottle System (604***, 605, 650) (Cont'd)				
	2) Autothrottle Mode Status Display (MSD) (605, 650)	С	2	1	May be inoperative on the non-flying pilot side and Autothrottle System is not used on that side.
		С	2	0	May be inoperative provided the Autothrottle System is considered inoperative.
	3) Autothrottle N2 Sync PBA (605, 650)	С	1	0	
	4) Engage LED Annunciator (605, 650)	С	2	1	
		С	2	0	May be inoperative provided the Mode Status Display (MSD) on the flying pilot side is operative.

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11 45	oyotomoroono unioio				Nombre d'articles à expédier
23 – <u>C</u>	<u>OMMUNICATIONS</u>				4. Remarks or Exceptions Remarques ou exceptions
11-1	Communication Systems (VHF, UHF, HF, etc)				
	1) VHF and UHF	D	-	2	Any in excess of those required by regulations may be inoperative provided: a) It is not powered by a standby or emergency bus, and b) It is not required for emergency purposes.
	2) HF	С	-	1	(O) May be inoperative while conducting operations that require two Long Range Communication Systems (LRCS) provided: a) SATVOICE or Data Link operates normally, b) SATVOICE or Data Link services are available over the intended route of flight, and, c) Alternate procedures are established and used.
		D	-	-	Any in excess of those required by regulations may be inoperative.
	(Cont'd)				

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23 – <u>C</u>	<u>OMMUNICATIONS</u>				4. Remarks or Exceptions Remarques ou exceptions		
11-2	COMM Control Head # 2 (COMM Digital Frequency Selector/LED Display Units) (601-3A, 601-3R)	С	1	0	May be inoperative provided the FMS CDU tuning capability is operative.		
11-3	Inflight Telephone System ***	D	-	0			
11-4	Datalink System (605, 650)*** (Includes CPDLC and ADS-C functions)***	С	1	0	(O) May be inoperative provided alternate procedures are established and used.		
	idiolons)	D	1	0	May be inoperative provided procedures do not require its use.	I	
	1) Controller Pilot Data Link Communications (CPDLC) ***	С	2	0	(O) May be inoperative provided alternate procedures are established and used.	1	
		D	2	0	May be inoperative provided procedures do not require its use.	I	
	(Cont'd)						

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23 – <u>CO</u>	MMUNICATIONS				4. Remarks or Exceptions Remarques ou exceptions	
					itemarques ou exceptions	
11-4	Datalink System					
	(605, 650)***					
	(Includes					
	CPDLC					
	and ADS-C					
	functions)***					
	(Cont'd)					
	(====)					
	1) Controller Pilot					
	Data Link					
	Communication					
	s (CPDLC)					

	(Cont'd)					
	a) CPDLC In-	С	2	0	May be inoperative provided In-Coming Message Aural	
	Coming				Alerting (chime) is operative.	
	Message					
	Visual Alerting					
	(CDU and					
	EICAS)					
		D	2	0	May be inoperative provided procedures do not require	
					the use of CPDLC.	
	0) Andrewell's	0			(O) March 1: 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	
	Automatic Dependent	С	2	0	(O) May be inoperative provided alternate procedures are established and used.	
	Surveillance				are established and used.	
	Contract					
	(ADS-C)					

		D	2	0	May be inoperative provided procedures do not require	
		ט		"	its use.	
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23 – <u>CO</u>	<u>MMUNICATIONS</u>				4. Remarks or Exceptions Remarques ou exceptions
21-1	Selective Call (SELCAL) System ***	С	-	0	(O) May be inoperative provided alternate procedures are established and used.
		D	-	0	May be inoperative provided procedures do not require its use.
	1) Channels (604, 605, 650)	D	2	1	
		С	-	0	(O) May be inoperative provided alternate procedures are established and used
		D	-	0	May be inoperative provided procedures do not require its use.
21-2	Satellite Communications (SATCOM) Systems ***	С	-	0	(O) May be inoperative provided alternate procedures are established and used.
		D	-	0	May be inoperative provided procedures do not require its use.
	1) File Server Unit (FSU)	С	1	0	Deleted, moved to 46-10-2.
		D	1	0	Deleted, moved to 46-10-2.
21-4	Integrated Flight Information System (IFIS) (605)				Deleted, moved to 46-10-2
	(Cont'd)				

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					Nombre d'articles à expédier			
23 – <u>C(</u>	OMMUNICATIONS				4. Remarks or Exceptions Remarques ou exceptions			
31-1	Passenger Address System							
	Passenger Configuration	С	1	0	 (O) May be inoperative provided: a) It is not required by regulations and b) Alternate, normal and emergency procedures, and/or operating restrictions are established and used. 			
					NOTE: Any station function(s) that operates normally may be used.			
		Α	1	0	(O) May be inoperative provided: a) Crewmembers are the only occupants of the aircraft,			
					b) Alternate procedures are established and used, and			
					c) Repairs are made within one flight day.			
	Cargo Configuration	D	1	0	May be inoperative provided all crew members are on the flight deck.			
41-1	Crewmember Interphone System	Α	1	0	(O) May be inoperative for non-passenger carrying operations provided:			
					 a) Crewmembers are the only occupants of the aircraft, 			
					b) Alternate procedures are established and used, and			
					c) Repairs are made within one flight day.			
		С	1	0	(O) May be inoperative provided:	1		
					a) It is not required by regulations, and			
					 b) Alternate, normal and emergency procedures, and/or operating restrictions are established and used. 	 		
					NOTE: Any station function(s) that operates normally may be used.			
	1) Flight	С	1	0	(O) May be inoperative provided:	I		
	Compartment to				a) PA system is verified operative, and			
	Cabin				b) Alternate procedures are established and used.			
	(Cont'd)							

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23 – <u>COMMUNICATIONS</u>					Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions		
41-1	Crewmember Interphone System (Cont'd)						
	Cabin to Flight Compartment	С	1	0	(O) May be inoperative provided: a) PA system is verified operative, and b) Alternate procedures are established and used.		
	Flight Compartment to Ground	С	1	0	(O) May be inoperative provided alternate procedures are established and used.		
		D	1	0	May be inoperative provided procedures are not dependent on its use.		
50-1	Cockpit Speakers	С	2	0	One or both may be inoperative provided: a) Aural alerts, messages and other communication which are normally routed through the speakers are audible through the headsets, b) Headsets are installed and used by each person on flight compartment duty, and c) Spare headset is available in flight compartment.		
50-2	Cabin Alerting System (Chime/Light)						
	1) Flight Deck Call Light	В	1	0	May be inoperative provided the flight deck chime is operative. NOTE: Flight deck chime must always be operative.		
	(Cont'd)						

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				4. Remarks or Exceptions Remarques ou exceptions	_		
50-2	Cabin Alerting System (chime/light) (Cont'd)						
	2) Visual Alert	В	-	0	(O) May be inoperative provided:	I	
	(Flight				a) PA system is operative,		
Cabin	Attendant / Cabin Call Lights) ***				b) If affected visual alerting system is used for lavatory smoke detector alerting, an alternate lavatory smoke detector alert (audio or visual) is installed and operates normally, and		
					 c) Alternate procedures for contacting flight attendants are established and used. 	1	
		В	_	0	(O) May be inoperative provided:	ı	
					a) Audio alerting system operates normally,	1	
					 b) Audio alerting system differentiates between normal and emergency calls, 	1	
					c) If affected visual alerting system is used for lavatory smoke detector alerting, an alternate lavatory smoke detector alert (audio or visual) is installed and operates normally, and	 	
					 d) Alternate procedures for contacting flight attendants are established and used. 		
	3) Aural Alert	В	_	0	(O) May be inoperative provided:	ļ	
	(Flight Attendant				a) PA system is operative,		
Cabin Chimes *** (Cont'd)					b) If affected audio alerting system is used for lavatory smoke detector alerting, an alternate lavatory smoke detector alert (audio or visual) is installed and operates normally, and		
					c) Alternate procedures for contacting flight attendants are established and used.		
	(Cont'd)						

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		1.	2.		er Installed	
•	n & Sequence No Item				re d'articles installés	
No de s	système/série article			3.	Number Required For Dispatch	
23 – C0	OMMUNICATIONS				Nombre d'articles à expédier 4. Remarks or Exceptions	
					Remarques ou exceptions	
50-2	Cabin Alerting System (chime/light) (Cont'd)					
	3) Aural Alert	В	1	0	(O) May be inoperative provided:	١
	(Flight				a) Visual alerting system operates normally,	
	Attendant Cabin Chimes)				b) Visual alerting system differentiates between normal and emergency calls,	
	***				c) If affected audio alerting system is used for lavatory smoke detector alerting, an alternate lavatory smoke detector alert (visual or audio) is installed and operates normally, and	
					d) Alternate procedures for contacting flight attendants are established and used.	
50-3	Pre-recorded Announcement (Passenger Briefing) System ***	С	1	0	(O) May be inoperative provided alternate procedures are established and used.	
50-4	Lavatory Speakers	С	-	0	(O) May be inoperative provided alternate procedures are established and used.	
51-1	Boom Microphones					
	CVR required to be capable of recording boom Microphone	Α	-	0	May be inoperative for three flight days provided associated hand microphone is installed and operates normally.	
	(Cont'd)					

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Nº de système/série article				3.	Number Required For Dispatch
					Nombre d'articles à expédier
23 – <u>CC</u>	OMMUNICATIONS .				4. Remarks or Exceptions Remarques ou exceptions
51-1	Boom Microphones (Cont'd)				
	2) CVR Not required to be capable of recording boom microphone	D	-	0	May be inoperative.
51-2	Flight Deck Hand Microphones	С	-	0	May be inoperative provided: a) May be inoperative provided associated boom microphone operates normally, and b) Each (Control Wheel) Press-To-Talk Switch is
		D	-	-	operative. Any in excess of those required by regulation may be inoperative.
51-3	Press-To-Talk (PTT) Switches	С	2	0	One or both may be inoperative provided: a) Switch (es) is not stuck in the transmit position,
	(Control Wheel) (600, 601, 601-3A, 601-3R)				b) Flight Deck Hand Microphone on affected side is operative.
51-4	RT/IC Switches (Pilot / Co-Pilot) (604, 605, 650)	С	4	2	One or two may be inoperative provided: a) Switch is not failed in the transmit mode; b) One RT/IC switch operates normally for each crew-member, and c) Flight Deck Hand Microphone on affected side is operative.

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Nº de système/série article				3.	Number Required For Dispatch	
		JNICATIONS				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
51-5	Hea	adsets	С	-	1	May be inoperative provided associated flight deck speaker operates normally.
			D	-	-	Any in excess of those required by regulation may be inoperative.
51-6		ckpit Speaker te System				
	1)	Cockpit Speaker Mute Pushbutton	С	1	0	May be inoperative in the mute state provided Cockpit Speakers are considered inoperative.
			D	1	0	(O) May be inoperative provided the system is not failed in the mute state.
	2)	Cockpit Speaker Mute Annunciators	D	2	0	
71-1	Red	ckpit Voice corder (CVR) stem				
	,	CVR and FDR required by regulations	Α	1	0	May be inoperative provided: a) Flight Data Recorder (FDR) is operative, and b) Repairs are made within three flight days.
	a)	Recorder Independent Power Supply (RIPS) ***	С	1	0	(M) May be inoperative provided it is removed from the CVR installation.
	(Co	ont'd)				

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Nº de s	systeme/serie article	stème/série article 3. Number Required For Dispatch Nombre d'articles à expédier				
23 – <u>CC</u>	<u>OMMUNICATIONS</u>				4. Remarks or Exceptions Remarques ou exceptions	
71-1	Cockpit Voice Recorder (CVR) System *** (Cont'd)					
	CVR only required by regulations	Α	1	0	May be inoperative provided repairs are made within three flight days.	
	a) Recorder Independent Power Supply (RIPS)	С	1	0	(M) May be inoperative provided it is removed from the CVR installation.	I
	CVR not required by regulations	D	-	0		
	a) Recorder Independent Power Supply (RIPS)	D	1	0	(M) May be inoperative provided it is removed from the CVR installation.	I
71-2	Electronic Checklist (604, 605, 650) ***	D	1	0	(O) May be inoperative provided alternate procedures are established and used.	

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	système/série article			3.	Number Required For Dispatch Nombre d'articles à expédier	
23 – <u>CC</u>	<u>OMMUNICATIONS</u>				4. Remarks or Exceptions Remarques ou exceptions	
81-1	Radio Tuning Units (RTUs) (604)	C	2	1	(O) RTU 2 may be inoperative provided: a) Radio Tuning from at least one FMS unit is verified operative, b) RTU 1 cross-side tuning is verified operative, and c) RTU #2 is de-selected using its RTU INHIBIT switch to ensure cross-side tuning by RTU #1, and d) Operations do not require the use of HF. NOTE 1: If RTU 1 fails, to restore ATC, the ATC selector switch, on the Reversionary/Inhibit panel, must be selected from STBY to 1. NOTE 2: Also, if RTU 1 fails, all RTU mode selections (COM 1 Squelch ON/OFF, NAV Marker sens. LO/HI, NAV DME-HOLD, ATC Altitude ON-OFF, ATC ID, ADF ANT, ADF BFO (ON/OFF), TCAS and HF) will be lost, however, COM/NAV frequency tuning will be available through FMS.	

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23 – <u>COMMUNICATIONS</u>			4. Remarks or Exceptions Remarques ou exceptions
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•	n & Sequence No Item				re d'articles installés
Nº de	système/série article			3.	Number Required For Dispatch
04 51	ECTRICAL DOWER				Nombre d'articles à expédier
24 – <u>EL</u>	<u>LECTRICAL POWER</u>				4. Remarks or Exceptions Remarques ou exceptions
21-1	Engine Integrated Drive Generators (IDGs)				
	1) (601-3A, 601-3R	Α	2	1	(M)(O) One may be inoperative provided:
	without SB 601-0568 and				 a) Inoperative generator GEN 1/2 switch is selected to OFF,
	604 without SB 604-49-006)				 APU is operated continuously throughout flight and the APU generator is verified operative prior to flight,
					c) APU generator is switched off throughout flight,
					d) Cross-side Hydraulic AC Motor Pump ACMP (1B or 2B) is selected ON,
					e) APU AC Loadmeter is operative,
					f) Aircraft is operated in day VMC,
					g) Aircraft is operated at or below FL 100 MSL,
					h) Aircraft is not operated more than 60 minutes from a suitable airport,
					i) If the operative IDG or APU generators have P/Ns 720845, 720845A or 720845B, they must have achieved an operational life in excess of 150 operating hours, and
					j) Repairs are made after one flight.
	(Cont'd)				

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•	système/série article			3.	Number Required For Dispatch
	- ,				Nombre d'articles à expédier
24 – <u>El</u>	<u>LECTRICAL POWER</u>				4. Remarks or Exceptions Remarques ou exceptions
21-1	Engine Integrated Drive Generators (IDGs) Cont'd				
	2) (604 without	В	2	1	(M) One may be inoperative provided:
	SB 604-49-006)				 a) Inoperative generator GEN 1/2 switch is selected to OFF,
					 APU is operated continuously throughout flight and the APU generator is verified operative prior to flight,
					c) APU generator is switched off throughout flight,
					d) Cross-side Hydraulic AC Motor Pump ACMP (1B or 2B) is considered inoperative,
					e) APU AC Loadmeter is operative,
					f) Aircraft is operated in day VMC,
					g) Aircraft is operated at or below FL 100 MSL,
					h) Aircraft is not operated more than 60 minutes from a suitable airport, and
					i) If the operative IDG or APU generators have P/Ns 720845, 720845A or 720845B, they must have achieved an operational life in excess of 150 operating hours.
	3) (601, 601-3A,	Α	2	1	(M)(O) One may be inoperative provided:
	601-3R with				a) Inoperative generator GEN 1/2 switch is selected to OFF,
	SB 601-0568)				 APU and its generator is operated continuously throughout flight,
					c) Cross-side Hydraulic AC Motor Pump ACMP (1B or 2B) is selected to ON,
					d) Aircraft is operated at or below FL 200,
					e) If the operative IDG or APU generators have P/Ns 720845, 720845A or 720845B, they must have achieved an operational life in excess of 150 operating hours, and
					f) Repairs are made after one flight.
	(0 4 5				
	(Cont'd)				

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No de s	système/série article			3.	Number Required For Dispatch
24 – <u>EL</u>	ECTRICAL POWER				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
21-1	Engine Integrated Drive Generators (IDGs) Cont'd				
	4) (604 with	Α	2	1	(M)(O) One may be inoperative provided:
	SB 604-49-006 and				 a) Inoperative generator GEN 1/2 switch is selected to OFF,
	SB 604-24-020)				 b) APU and its generator is operated continuously throughout flight,
					c) Cross-side Hydraulic AC Motor Pump ACMP (1B or 2B) is selected to ON,
					d) Aircraft is operated at or below FL 200,
					e) If the operative IDG or APU generators have P/Ns 720845, 720845A or 720845B, they must have achieved an operational life in excess of 150 operating hours, and
					f) Repairs are made after one flight.
		В	2	1	(M) One may be inoperative provided:
					 a) Inoperative generator GEN 1/2 switch is selected to OFF,
					 b) APU and its generator is operated continuously throughout flight,
					c) Cross-side Hydraulic AC Motor Pump (ACMP) (1B or 2B) is considered inoperative,
					d) Aircraft is operated at or below FL 200, and
					e) If the operative IDG or APU generators have P/Ns 720845, 720845A or 720845B, they must have achieved an operational life in excess of 150 operating hours.
	(Cont'd)				

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•	n & Sequence No Item				re d'articles installés
Nº de	système/série article			3.	Number Required For Dispatch
24 EI	ECTRICAL DOWER				Nombre d'articles à expédier 4. Remarks or Exceptions
24 – <u>CL</u>	<u>LECTRICAL POWER</u>				Remarques ou exceptions
21-1	Engine Integrated Drive Generators (IDGs) Cont'd				
	5) (605, 650)	A	2	1	 (O) One may be inoperative provided: a) Inoperative generator GEN 1/2 switch is selected to OFF, b) APU and its generator is operated continuously throughout flight, c) Cross-side Hydraulic AC Motor Pump ACMP (1B or 2B) is selected to ON, d) Aircraft is operated at or below FL 200, and e) Repairs are made after one flight.
		В	2	1	One may be inoperative provided: a) Inoperative generator GEN 1/2 switch is selected to OFF, b) APU and its generator is operated continuously throughout flight, c) Cross-side Hydraulic AC Motor Pump (ACMP) (1B or 2B) is considered inoperative, and d) Aircraft is operated at or below FL 200.

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_		1.	2.		er Installed
System & Sequence No Item					re d'articles installés
Nº de	système/série article			3.	Number Required For Dispatch
04 5	LECTRICAL DOMER				Nombre d'articles à expédier
24 – <u>El</u>	LECTRICAL POWER				4. Remarks or Exceptions Remarques ou exceptions
22-1	Auxiliary Power Unit (APU) AC Loadmeter				
	1) Aircraft with IDG	С	1	0	(M) May be inoperative provided:
	P/Ns 720845, 720845A,				 a) Any main generator with these P/Ns has more than 150 operating hours,
	720845B				b) APU generator is not used, and
					c) APU generator switch is selected to OFF/RESET.
	2) Aircraft with	С	1	0	May be inoperative provided:
	other IDG P/Ns	O	'		a) APU generator is not used, and
					b) APU generator switch is selected to OFF/RESET.
22-2	Auxiliary Power Unit (APU) Generator System	0			
	1) Aircraft with IDG P/Ns 720845, 720845A, 720845B	С	1	0	(M) May be inoperative provided any main generator with these P/Ns has more than 150 operating hours.
	Aircraft with other IDG P/Ns	С	1	0	
23-1	Air Driven Generator (ADG) Auto- Deployment System	С	1	0	(M) May be inoperative provided the Auto-Deployment system is deactivated.

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	ystème/série article			3.	Number Required For Dispatch
					Nombre d'articles à expédier
24 – <u>ELI</u>	ECTRICAL POWER				4. Remarks or Exceptions Remarques ou exceptions
30-1	Main Airplane Battery (600, 601, 601-3A, 601-3R)	D	-	1	 (M) Main Airplane Battery may be inoperative provided: a) Additional Main Airplane Battery and its associated Charger are operative, and b) Affected Main Airplane Battery and its associated Charger are isolated by disconnecting and securing them.
30-2	Main Airplane Battery Charger (600, 601, 601-3A, 601-3R)	D	-	1	 (M) Main Airplane Battery Charger may be inoperative provided: a) Additional Main Airplane Battery and its associated Charger are operative, and b) Affected Main Airplane Battery and its associated Charger are isolated by disconnecting and securing them.
30-3	APU Auxiliary Battery (601-3A) (600,601)	D	-	0	
30-4	Refueling Panel Battery ***	D	1	0	(M) May be inoperative provided Airplane Battery can be used to fuel airplane.
30-5	Battery Ammeter ***	D	1	0	
31-1	TRU Cooling Fans				
	1) 600, 601	С	3	1	 (M) One or two may be inoperative provided: a) ESS TRU Cooling Fan is operative, and b) Forward equipment doors are opened for stationary ground operations at temps >30 °C.
	2) 601-3A, 601-3R, 604, 605, 650	С	4	2	 (M) One or two may be inoperative provided: a) ESS TRU 1 & 2 Cooling Fans are operative, and b) Forward equipment doors are opened for stationary ground operations at temps >30 °C.

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11 40 3	yotomoroone article				Nombre d'articles à expédier
24 – <u>ELE</u>	ECTRICAL POWER				4. Remarks or Exceptions Remarques ou exceptions
32-1	Main Battery				
	1) (604)	A	1	0	 (M) May be inoperative provided: a) APU Battery is operative, b) APU Battery Charger is operative, c) Boarding lights are considered inoperative, d) Service lights are considered inoperative, e) APR is selected off, f) Operations are conducted in accordance with AFM APR inoperative performance data, and g) Repairs are made within one flight day. (M)(O) May be inoperative provided: a) APU Battery is operative, b) APU Battery Charger is operative, c) Boarding lights are considered inoperative, d) Service lights are considered inoperative, and e) Repairs are made within one flight day.
32-2	Main Battery Charger 1) (604)	Α	1	0	(M) May be inoperative provided:
	1) (004)		1	U	 a) Main Battery is considered inoperative, b) APU Battery is operative, c) APU Battery Charger is operative, d) Boarding lights are considered inoperative, e) Service lights are considered inoperative, f) APR is selected off, g) Operations are conducted in accordance with AFM APR inoperative performance data, and h) Repairs are made within one flight day.
	(Cont'd)				

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				3.	Number Required For Dispatch		
					Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions		
32-2	Main Battery Charger (Cont'd)						
	2) (605, 650)	A	1	0	 (M)(O) May be inoperative provided: a) Main Battery is considered inoperative, b) APU Battery is operative, c) APU Battery Charger is operative, d) Boarding lights are considered inoperative, e) Service lights are considered inoperative, and f) Repairs are made within one flight day. 		
32-3	APU Battery (604, 605, 650)	Α	1	0	 (M) May be inoperative provided: a) Main Battery is operative, b) Main Battery Charger is operative, c) Both IDGs are operative, and d) Repairs are made within one flight day. 		
					NOTE 1: External DC power can be used to start APU if required.		
					NOTE 2 : External DC power may be required for refueling.		
32-4	APU Battery Charger (604, 605, 650)	A	1	0	 (M)(O) May be inoperative provided: a) APU Battery is considered inoperative after APU is started, b) Main Battery is operative, c) Main Battery Charger is operative, d) Both IDGs are operative, and e) Repairs are made within one flight day. NOTE 1: External DC power can be used to start		
					APU if required. NOTE 2: External DC power may be required for refueling.		

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24 – <u>EL</u>	LECTRICAL POWER				4. Remarks or Exceptions Remarques ou exceptions
32-5	Battery Temperature Monitors ***	D	-	0	
41-1	AC External Power System	С	1	0	
41-2	AC External Power "AVAIL / IN USE" Switch Lights (light function only) (604, 605, 650)	С	2	0	
42-1	DC External Power System	С	1	0	
42-2	DC Power Main Bus Switch Light (light function only) (604, 605, 650)	С	1	0	
42-3	DC External Power "IN USE" Light (604, 605, 650)	С	1	0	
42-4	AC/DC UTILITY Switch light (light function only) (605, 650)	С	1	0	
50-1	USB Charger System (650 A/C 6071 and Subs)	D	1	0	(O) May be inoperative provided it is deactivated.

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24 – <u>ELECTRICAL POWER</u>			4. Remarks or Exceptions Remarques ou exceptions
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25 _ F(QUIPMENT/				Nombre d'articles à expédier 4. Remarks or Exceptions		
	JRNISHINGS				Remarques ou exceptions		
					·		
10-2	Chart Holders (CL600, 601, 601-3A, 601-3R, 604, 605, 650 aircraft 6050 to 6070) (650 aircraft 6071 and Subs***)	С	-	0			
	1) Integrated Lights	С	2	0	May be inoperative provided cockpit lighting configuration is acceptable to the crewmembers.		
11-1	Pilot Seats						
	1) Lumbar	С	2	0	May be inoperative provided:		
	supports				a) The lumbar support position is at the lowest position, and		
					b) Seat is acceptable to affected crewmember.		
	2) Arm rests	С	4	0	(M) May be inoperative or missing provided:		
					Affected arm rest is secured in the upright position,		
					b) Egress is not impaired, and		
					c) Seat is acceptable to affected crewmember.		
	3) Height	В	2	0	(M) May be inoperative provided:		
	Adjustments				 a) Seat is secured in vertical position acceptable to affected crewmember, 		
					b) Fore/Aft Adjustment is verified operative,		
					c) Egress is not impaired, and		
					d) If Head-up display is installed and required for flight, the vertical position of the seat must be acceptable to affected crewmember.		
	4) Fore/Aft	В	2	0	(M) May be inoperative provided:		
	Adjustments				Seat is secured in fore/aft position acceptable to affected crewmember,		
					b) Height Adjustment is verified operative, and		
					c) Egress is not impaired.		
	(Cantid)						
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	•				Nombre d'articles à expédier		
	QUIPMENT/ IRNISHINGS				4. Remarks or Exceptions Remarques ou exceptions		
11-1	Pilot Seats (Cont'd)						
	5) Recline Adjustments	В	2	0	(M) May be inoperative provided backrest is secured in a position acceptable to affected crewmember.		
	6) Thigh Supports	С	2	0	May be inoperative provided seat is acceptable to affected crewmember.		
	7) Seat Belt Inertia Reel	С	2	0	May be inoperative in the rearward position (locked position)		
11-2	Cockpit Sunvisors	С	2	0	May be inoperative or missing provided there are no visual restrictions to crewmembers.		
11-3	Glareshield Retractable Visors (604, 605, 650)	С	2	0	(M) May be inoperative or missing provided visor is secured in the retracted position or removed.		
12-1	Observer Seat (Including associated equipment)	В	1	0	(M) May be inoperative except when required by a person in an official capacity provided the seat is removed, stowed or secured in the retracted position.		
	ечирненку	D	1	0	(M) May be inoperative provided:a) Procedures do not require its use, andb) The seat is removed, stowed, or secured in the retracted position.		
					NOTE: Associated equipment includes safety belt, shoulder harness, audio control panel, oxygen system, microphone, headset, lights, etc.		

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i -	système/série artic	i		3.	Number Required For Dispatch	
	-,				Nombre d'articles à expédier	
	QUIPMENT/ JRNISHINGS				4. Remarks or Exceptions Remarques ou exceptions	
20-1	Passenger Seats	D	-	0	(O) May be inoperative provided:	I
					a) Seat does not block an emergency exit,	
					 b) Seat does not restrict any passenger from access to the main aircraft aisle, and 	
					c) Affected seat(s) is blocked and placarded "DO NOT OCCUPY".	
					NOTE 1: A seat with an inoperative seat belt and/or shoulder harness is considered inoperative.	
					NOTE 2 : Affected seat(s) may include the seat(s) behind and/or adjacent outboard seats.	
	Recline Mechanism	D	-	-	(M) May be inoperative and seat occupied provided seat is secured in the taxi, takeoff, and landing position.	1
		С	-	-	May be inoperative and seat occupied provided seat is immovable in the taxi, takeoff, and landing position.	I
	2) Moveable Armrest	D	-	-	(M) May be inoperative or missing and seat occupied provided:	1
	***				a) Armrest does not block an emergency exit,	-
					b) Armrest does not restrict any passenger from access to the main aircraft aisle, and	
					c) If armrest is missing, seat is secured in the taxi, takeoff, and landing position.	
	3) Track/Swivel Mechanism	D	-	-	(M) May be inoperative and seat occupied provided the seat is secured in the taxi, takeoff, and landing position.	I
		С	-	-	May be inoperative and seat occupied provided seat is immovable in the taxi, takeoff, and landing position.	1
	4) Legrest	D	-	-	(M) May be inoperative and seat(s) occupied provided legrest is secured in the taxi, takeoff, and landing position.	I
		С	-	-	May be inoperative and seat occupied provided legrest is immovable in the taxi, takeoff, and landing position.	

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_	système/série article			3.	Number Required For Dispatch
	•				Nombre d'articles à expédier
	<u>RUIPMENT</u> / RNISHINGS				4. Remarks or Exceptions Remarques ou exceptions
20-2	Underseat Baggage Restraining Bars	С	-	-	 (M)(O) May be inoperative or missing provided: a) Baggage is not stowed under seat or bank of seats, b) Seat back is placarded "DO NOT STOW BAGGAGE UNDER THIS SEAT", c) Restraining bar does not restrict any passenger from access to the main aircraft aisle, and d) Procedures are established to alert crewmembers of an inoperative or missing restraining bar.
20-3	Cabin Seat Storage Door (Under Seat) ***	D	-	0	 (M)(O) May be inoperative or missing provided: a) If missing, contents must be relocated, b) If inoperative, door must be closed and secured and contents relocated, and c) If contents contain emergency equipment, the equipment must be relocated and accessible for the passenger or the seat must be considered inoperative. If the relocated emergency equipment is an approved means of flotation, it must be accessible to the passenger while seated.

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Nº de système/série article			3.	Number Required For Dispatch		
11 40	oyotomoroono uniolo				Nombre d'articles à expédier	
	<u>QUIPMENT</u> / <u>JRNISHINGS</u>				4. Remarks or Exceptions Remarques ou exceptions	
20-4	Divan Seat Berthing Belt ***	D	-	0	May be inoperative or missing provided divan seat is not used for berthing.	
20-5	Cabin Gasper Vents	D	-	0		
29-1	Cabin Power Outlet	D	-	0	(O) May be inoperative provided affected outlet is deactivated.	I
29-2	Flight Deck Power Outlets	С	-	0	(O) May be inoperative provided affected outlet is deactivated.	
		D	-	0	(O) May be inoperative provided: a) Affected outlet is deactivated, and	
					b) Procedures do not require its use.	
30-1	Storage	С	-	-	(M) May be inoperative provided:	
	Bin(s)/Cabin, Galley, and Lavatory Storage				 a) Procedures are established and used to secure the affected bin, compartment, or closet closed, 	
	Compartment/ Closets				 b) Affected bin, compartment or closet is prominently placarded DO NOT USE, 	
					 c) Any emergency equipment located in the affected compartment is considered inoperative and, 	
					 d) Affected bin, compartment, or closet is not used for storage of any items except for those permanently affixed. 	
					NOTE: An inoperative lid/door latch renders the door inoperative.	
	(Cont'd)					

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	système/série article			3.	Number Required For Dispatch
111 40 0	yotomoroono uniolo				Nombre d'articles à expédier
	<u>QUIPMENT/</u> RNISHINGS				4. Remarks or Exceptions Remarques ou exceptions
30-1	Storage Bin(s)/Cabin, Galley, and Lavatory Storage Compartment/ Closets (Cont'd)				
	(30.11.2)	С	-	_	(M)(O) May be inoperative provided:
					a) For non-retractable doors, affected door is removed,
					 b) For retractable doors, affected door is removed or secured in the retracted (fully open) position,
					 c) Affected bin, compartment or closet is not used for storage of any items, except those permanently affixed,
					 d) Affected bin, compartment or closet is prominently placarded DO NOT USE,
					Procedures are established and used to alert crew members and passengers of inoperative bins, compartments or closets, and
					f) Passengers are briefed that affected bin, compartment or closet is not used.
					NOTE 1 : Any emergency equipment located in the associated compartment (permanently affixed) is available for use.
					NOTE 2 : An inoperative lid/door latch renders the lid/door inoperative.
50-1	Cargo Restraint Systems	С	-	-	May be inoperative or missing provided baggage compartment remains empty.
		С	-	-	All cargo is secured by an alternate approved method.
60-1	Flotation Equipment (Crew and Passenger)	D	-	-	As required by Regulations.

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-	système/série article			3.	Number Required For Dispatch
	•				Nombre d'articles à expédier
	<u>QUIPMENT</u> / JRNISHINGS				4. Remarks or Exceptions Remarques ou exceptions
60-2	Emergency Locator Transmitter (ELT) ***				
	1) Fixed ELTs	Α	1	0	(M) May be inoperative provided:
					a) System is deactivated if not removed,
					 b) Placard is displayed in the flight deck indicating the removal date if the ELT has been removed, and
					 c) Repair or replacement is made within the time interval prescribed by regulations.
		D	-	-	Any in excess of those required by regulations may be inoperative or missing.
	Survival Type ELTs	D	-	-	Any in excess of those required by regulations may be inoperative or missing.

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-	ystème/série article			3.	Number Re	quired For Dispatch	
25 – <u>EQ</u>	<u>UIPMENT</u> / RNISHINGS				4. Re	nticles à expédier marks or Exceptions marques ou exceptions	
60-3	Passenger Convenience Items/NEF Items 1) Passenger Convenience Items (Expires on 24 February 2025)		-	-	MMEL, are comfort or galley equi equipment elsewhere and (O) pro	convenience items, as expressed in this a those related to passenger convenience, entertainment such as, but not limited to, pment, movie equipment, ashtrays, stereo, overhead reading lamps. Items addressed in this document shall not be included. (M) ocedures may be required and included in ier's appropriate document. Exterior lavatory door ashtrays are not considered convenience items. (Only applicable to transport category airplanes affected by FAA AD 74-08-09) Galley Equipment restraining devices such as latches, etc. must be serviceable or compartment must not be used for storage and placarded "INOPEATIVE DO NOT USE". Movie equipment individual screens, if applicable, must be capable of being stowed. Audio or audio-visual entertainment equipment which is used as the sole means of providing safety briefings and demonstrations is not considered as passenger convenience item.	

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01	0 O NO 16	1.	2.		er Installed re d'articles installés
1 -	& Sequence Nº Item ystème/série article			3.	Number Required For Dispatch
Nº de S	ysterne/serie article			0.	Nombre d'articles à expédier
	<u>UIPMENT</u> / RNISHINGS				4. Remarks or Exceptions Remarques ou exceptions
60-3	Passenger Convenience Items/NEF Items (Cont'd)				
	Non-Essential Equipment and Furnishings (NEF)		-	0	May be inoperative, damaged or missing provided that the item(s) is deferred in accordance with the NEF program outlined in the operator's Maintenance Control Manual (MCM) or Maintenance Control System, as applicable. (M) and (O) procedures, if required, must be available to the flight crew and included in the operator's appropriate document.
					NOTE: Exterior lavatory door ashtrays are not considered NEF items. (Only applicable to transport category airplanes affected by FAA AD 74-08-09)
60-4	Airshow Cockpit Monitor System	D	1	0	(M) May be inoperative provided system is deactivated.
60-5	Security System	D	1	0	
60-6	Flashlights / Flashlight Holders				
	1) Flashlights	С	-	0	(O) May be inoperative or missing provided affected crewmember has a flashlight of equivalent characteristics readily available.
	2) Holders	С	-	0	(O) May be inoperative or missing provided alternate stowage provisions are provided.
60-7	Emergency Vision Assurance System (EVAS) ***	D	2	0	

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_		1.	2.		er Installed	
-	n & Sequence No Item				re d'articles installés	
No de	système/série article		-	3.	Number Required For Dispatch	
25 _ E(QUIPMENT/				Nombre d'articles à expédier 4. Remarks or Exceptions	
	JRNISHINGS				Remarques ou exceptions	
60-8	"FASTEN SEAT BELT WHILE SEATED" sign or placard	С	-	-	One or more signs or placards may be illegible or missing provided a legible sign or placard is readable from each occupied passenger seat.	
60-9	First Aid Kit	D	-	-	(O) Any kit or items contained in the kit in excess of those required by Regulations may be incomplete or missing provided:	
					a) Required distribution is maintained, and	
					b) Procedures are established and used to alert crewmembers of missing or incomplete kits.	I
	1) First Aid Kit Seal	Α	-	0	(O) The seal affixed on the exterior of any required first aid kit may be missing broken for three flight days provided:	
					 a) The first aid kit is fully equipped or the kit has a maximum of one missing item, 	
					b) The kit includes a list of its contents,	
					 c) An inventory is taken on the content of the kit prior to departure, and 	
					 d) Procedures are established and used to alert crewmembers of: 	I
					 The missing or broken seal, and 	
					The need to perform an inventory under proviso c).	
70-1	Galley Waste Receptacles Access Doors/Covers	С	-	-	(O) May be inoperative provided: a) The container is empty and the access is secured to prevent waste introduction into the compartment, and	I
					 b) Procedures are established to ensure that sufficient galley/cabin waste receptacles are available to accommodate all waste that may be generated on a flight. 	
			1			

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_	n & Sequence Nº Item système/série article			3.	Number Required For Dispatch
Nº GE S	systemerserie article				Nombre d'articles à expédier
	QUIPMENT/ JRNISHINGS				4. Remarks or Exceptions Remarques ou exceptions
70-2	Exterior Lavatory Door Ashtrays				
	Airplanes with more than one external lavatory door ashtray installed	Α	-	-	Up to and including 50 percent may be missing or inoperative for 10 calendar days.
		Α	-	-	More than 50 percent may be missing or inoperative for 3 calendar days.
	2) Airplanes with only one external lavatory door ashtray installed	Α	1	0	May be missing or inoperative for 10 calendar days.
70-3	Lavatory Waste Receptacle	C	-	0	(M) May be inoperative provided: a) Associated Lavatory Waste Bin is empty, and b) Associated Lavatory Waste Bin is secured closed, and placarded DO NOT USE.

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Nº de système/série article		3.	Number Required For Dispatch
	1		Nombre d'articles à expédier
25 – <u>EQUIPMENT</u> / <u>FURNISHINGS</u>			4. Remarks or Exceptions Remarques ou exceptions
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	système/série article			3.	Number Required For Dispatch	
	,				Nombre d'articles à expédier	
26 – <u>FI</u>	RE PROTECTION				4. Remarks or Exceptions Remarques ou exceptions	
10-1	APU Fire Detection System					
	1) (600, 601,	С	1	0	(M) May be inoperative provided:	
	601-3A, 601-3R)				 a) If both main generators have P/Ns 720845, 720845A and 720845B they must have more than 150 operating hours, and 	
					b) APU is not used.	
		С	1	0	(M) May be inoperative provided:	
					a) If both main generators have P/Ns 720845, 720845A and 720845B they must have more than 150 operating hours,	
					b) APU is used on ground for engine start only,	
					c) APU is not used after engine start,	
					d) APU external control system is operative, and	
					e) Both IDGs are operative.	
	2) (604, 605, 650)	С	1	0	(M) May be inoperative provided:	
					a) APU is used on ground for engine start only,	
					b) APU is not used after engine start, and	
					c) APU external control system is operative.	
		С	1	0	May be inoperative provided APU is not used.	
10-2	Lavatory Smoke Detection Systems	D	-	0	Any in excess of that required by regulations may be inoperative.	

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1	système/série article			3.	Number Required For Dispatch	_
00 511	DE DDOTECTION				Nombre d'articles à expédier	_
26 – <u>FII</u>	RE PROTECTION				4. Remarks or Exceptions Remarques ou exceptions	
10-3	Baggage Compartment Fire Detection / Suppression System ***	С	-	0	May be inoperative provided the baggage compartment is empty.	
14-1	Main Landing Gear	С	1	0	(M)(O) May be inoperative provided:	
	Bay Overheat Detection System				 a) Brakes are inspected prior to each flight and are cool to the touch, 	
					 b) After take-off the landing gear is left extended for ten minutes, 	
					c) Takeoff performance is in accordance with the AFM (Flight with Landing Gear Down), and	
					d) Take-off is not conducted in icing conditions.	
					NOTE: In case of engine failure after V1, performance is the prime consideration and the landing gear should be retracted normally until performance penalty with gear down is not a problem.	1
20-1	Lavatory Fire Extinguisher System ***	С	-	0	(O) May be inoperative in each lavatory provided the Lavatory Smoke Detection System is operative,	1
		С	-	-	(M)(O) May be inoperative in each lavatory provided: a) Lavatory waste receptacle is empty,	I
					 b) Associated lavatory door is locked and placarded "INOPERATIVE – DO NOT ENTER", and 	I
					c) Lavatory is not used for any purpose.	
					NOTE: These provisos are not intended to prohibit lavatory use or inspections by crew members.	
		D	-	0	Any in excess of that required by regulations may be inoperative.	
20-2	Fuselage Fire Extinguishing System (Total Flood) ***	С	-	0	(M) May be inoperative provided the system is secured.	

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System & Sequence No Item No de système/série article				ore d'articles installés	
No de	système/sèrie article			3.	Number Required For Dispatch Nombre d'articles à expédier
26 – FI	RE PROTECTION				4. Remarks or Exceptions
<u></u>					Remarques ou exceptions
20-3	Baggage Compartment Smoke Detection System	С	1	0	May be inoperative provided the associated compartment is empty or does not contain combustible materials.
					NOTE: Unit Load Devices (ULD's) may be carried in the associated compartment provided no cargo is carried on or in these devices.
20-4	Galley Fire	С	1	0	May be inoperative provided:
	Extinguishing				a) The galley is not used for any purpose, and
	System ***				b) Any waste receptacles are empty.
20-5	APU Fire Extinguishing System				
	1) (600, 601, 601-3A, 601-3R)	С	1	0	 (M) May be inoperative provided: a) If both main generators have P/Ns 720845, 720845A and 720845B they must have more than 150 operating hours, and b) APU is not used.
		С	1	0	(M) May be inoperative provided:
					a) If both main generators have P/Ns 720845, 720845A and 720845B they must have more than 150 operating hours,
					b) APU is used on ground for engine start only, and
					c) APU is not used after engine start.
					d) APU external control system is operative, and
					e) Both IDGs are operative.
		С	1	0	May be inoperative provided APU is not used.
	(Cont'd)				

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				3.	Number Required For Dispatch
-					Nombre d'articles à expédier
26 – <u>F</u>	IRE PROTECTION				4. Remarks or Exceptions Remarques ou exceptions
20-5	APU Fire Extinguishing System (Cont'd)				
	2) (604, 605, 650)	С	1	0	(M) May be inoperative provided:
					a) APU is used on ground for engine start only,
					b) APU is not used after engine start.
					c) APU external control system is operative, and
					d) Both IDGs are operative.
		С	1	0	May be inoperative provided APU is not used.
20-6	APU FIREX Bottle Low Pressure Switch	0	4		(M)(O) Mov ha in an aretive may ide du
	1) (600, 601, 601-3A, 601-3R)	С	1	0	(M)(O) May be inoperative provided: a) If both main generators have P/Ns 720845, 720845A and 720845B they must have more than 150 operating hours,
					b) The APU is used for ground operations only,
					c) Firex bottle charge pressure is visually verified at bottle pressure gauge prior to each APU start,
					d) APU fire detection system is operative,
					e) APU external control system is operative,
					f) APU is not used after engine start, and
					g) Both IDGs are operative.
	2) (604, 605, 650)	С	1	0	(M) (O) May be inoperative provided: a) The APU is used for ground operations only,
					b) Firex bottle charge pressure is visually verified at bottle pressure gauge prior to each APU start,
					c) APU fire detection system is operative,
					d) APU external control system is operative, and
	(Cont'd)				e) APU is not used after engine start.

DEPARTMENT OF TRANSPORT

MINISTÈRE DES TRANSPORTS MASTER MINIMUM EQUIPMENT LIST LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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26 – FIRE PROTECTION					Nombre d'articles à expédier
26 – <u>FI</u>	RE PROTECTION				4. Remarks or Exceptions Remarques ou exceptions
25-1	Portable Fire Extinguishers	D	-		 (O) Any of excess of those required by regulations may be inoperative or missing provided: a) The fire extinguisher is removed from the aircraft, or removed from the installed location, placarded INOPERATIVE and secured out of sight, b) Required distribution is maintained, and c) Procedures are established and used to alert crewmembers of missing portable fire extinguisher.

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Nº de s	système/série article			3.	Number Required For Dispatch
27 – <u>FL</u>	IGHT CONTROLS				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
05-1	Control Surface Position Indicating Systems				
	1) FLT SPLR L/R AILERON L/R ELEVATOR L/R RUDDER (600, 601, 601- 3A, 601-3R)	С	7	0	 (M)(O) Any or all indicator(s) may be inoperative provided: a) Affected indicator is clearly identified as inoperative or covered, and b) Affected control surface is visually checked for full, free and correct movement prior to each flight.
	2) EICAS Indications (FLT SPLR, AIL, RUDDER) (604, 605, 650)	С	5	0	(O) Any or all may be inoperative provided affected Control Surface is visually checked for full, free and correct movement prior to each flight.
	3) EICAS Indications (ELEV) (604, 605, 650)				
	a) L/H	С	1	0	 (O) May be inoperative provided: a) Affected Control Surface is visually checked for full, free and correct movement prior to each flight, and b) Autopilot is not engaged below 1500 feet AGL.
	b) R/H	D	1	0	(O) May be inoperative provided affected control surface is visually checked for full, free and correct movement prior to each flight.

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1 -	ystème/série article			3.	Number Required For Dispatch
27 – FLIGHT CONTROLS					Nombre d'articles à expédier
21 – <u>FLI</u>	GHT CONTROLS				4. Remarks or Exceptions Remarques ou exceptions
05-2	Ground Spoiler Control Surface Indications (LH & RH) (604, 605, 650)	С	2	0	 (M) One or both may be inoperative provided: a) "ARM" mode is operative, b) Ground Spoilers switch is selected to AUTO, and c) Ground spoilers are verified stowed before each take-off. NOTE: "PROX SYS FAULT" status message appears
05-3	Servo Monitor System (600, 601, 601-3A, 601-3R)	С	1	0	with "GND SPLRS" caution message. (M)(O) May be inoperative provided: a) All Powered Control Units (PCUs) are checked prior to take-off to ensure correct operation using each hydraulic system, and b) Parking Brake SOV is open.
05-4	LH/RH Flight Spoiler Indication (Glareshield) (600, 601, 601-3A, 601-3R)	С	2	0	 (O) Both may be inoperative provided: a) Flight Spoiler Control Surface Position Indication is operative, b) Flight Spoiler Indication on center pedestal is verified operative prior to each flight, c) Flight Spoilers are verified stowed before takeoff, and d) Flight Spoilers are not used during approach.
12-1	Aileron Trim System	В	1	0	May be inoperative provided: a) Autopilot is operative, b) Aileron Trim System is centered, and c) Main fuel tanks left and right are balanced prior to take-off.
35-1	Stall Warning Switch Lights (light function only) (604)	С	2	1	One may be inoperative provided the shaker and the pusher are checked operative prior to each flight.

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27 EI	LIGHT CONTROLS				Nombre d'articles à expédier 4. Remarks or Exceptions		
21 – <u>Fl</u>	LIGITI CONTROLS				Remarques ou exceptions		
40-1	Trim Indicators (Ailerons, Rudder)	С	2	1	(O) One may be inoperative provided the affected control surface is checked visually as being in the appropriate take-off position prior to each take-off.		
50-1	Flap Power Drive	С	2	1	(O) May be inoperative provided:	I	
	Unit (PDU) Motors				a) Affected motor is deactivated, and		
					 b) Operations are conducted in accordance with AFM Supplement: 		
					AFM Supplement 14, Operations with Airplane Systems Inoperative (600, 601, 601-3A, 601-3R)		
					AFM Supplement 8, Operations with Airplane Systems Inoperative (604, 605, 650).		
					NOTE: With only one motor operational, the flaps will operate at half speed.		
52-1	Rudder Pedal Adjustment Mechanism	С	2	0	 (M)(O) May be inoperative provided: a) Rudder pedals can be secured in a position acceptable to affected crewmember, and b) Rudder and brake pedals are checked for full and unrestricted movement at both pilot stations. 		
					NOTE: Full range of operation must be available to the crewmember in seated position.		

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-	système/série article			3.	Number Required For Dispatch		
-					Nombre d'articles à expédier		
27 – <u>FL</u>	IGHT CONTROLS				4. Remarks or Exceptions Remarques ou exceptions		
60-1	Ground Spoiler Manual Arming Switch A/C with SB 600-0452 SB 601-0113 A/C 3060 & subs (604, 605, 650)	С	1	0	(O) May be inoperative provided the ground spoiler automatic deployment is verified operative before each departure.		
60-2	Flight Spoiler Lights Flashing Feature A/C with SB 600-0385 SB 601-0040 A/C 3013 & 3018 to 3990	В	2	0	 May be inoperative provided: a) Flight Spoilers are not used below 700 feet AGL, b) Flight spoilers are not used when flaps are extended, and c) All flight spoilers indications are operative. 		
67-1	Ground Spoiler System						
	1) 600, 601	С	1	0	 (O) May be inoperative provided: a) System is selected to OFF, b) Ground spoilers are verified in the retracted position prior to each flight, and c) AFM Performance Chart entitled "Landing Distance (Without Ground Spoilers)" is used. 		
					NOTE: Service Bulletin 600-0368 must be installed on the 600, or Service Bulletin 601-0020 must be installed on the 601.		
	2) 601-3A, 601-3R	С	1	0	 (O) May be inoperative provided: a) System is selected to OFF, b) Ground spoilers are verified in the retracted position prior to each flight, and c) AFM Performance Chart entitled "Landing Distance (Without Ground Spoilers)" is used. 		
	(Cont'd)						

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14- 46	systeme/serie article				Nombre d'articles à expédier
27 – <u>Fl</u>	LIGHT CONTROLS				4. Remarks or Exceptions Remarques ou exceptions
67-1	Ground Spoiler System (Cont'd)				
	3) 604, 605, 650	C	1	0	 (O) May be inoperative provided: a) System is selected to DISARM, b) Ground spoilers are verified in the retracted position prior to each flight, and c) Operations are conducted in accordance with AFM Supplement 8 (Operations with Airplane Systems inoperative).

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27 - FLIGHT CONTROLS			4. Remarks or Exceptions
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28 – <u>FU</u>	<u>IEL</u>				4. Remarks or Exceptions Remarques ou exceptions	
11-1	Aux Tank Fuel Panel with Drain Valve	Α	1	0	 (M) May be inoperative (leaking) provided: a) Aux and Tail fuel tanks are verified empty prior to the first flight of each day and remain empty, b) Powered Crossfeed is considered inoperative and not used, 	I
					c) Aux and Tail fuel tank quantity indication or EICAS readout is operative, and	
					d) Repairs are made within 350 flight hours.	
					NOTE: The Wing Tank Fuel Panel with Drain Valve is interchangeable with the Aux Tank Fuel Panel with Drain Valve. Refer to AMM.	
		D	1	0	(M) May be inoperative (leaking) provided panel with drain valve is replaced with a blank panel.	
11-2	Water Drain Valves	С	-	5	 May be inoperative provided: a) One drain valve at each wing tank, lowest level point is operative, b) One drain valve in the center tank is operative, c) Main Boost Pump drain valves are operative, and d) There is no evidence of leakage. 	
11-3	Gravity Filler Cap Locks	D	3	0	(M) May be inoperative UNLOCKED provided there is no fuel leakage from affected Gravity Filler Cap.	
		С	3	0	 (M)(O) May be inoperative LOCKED provided: a) There is no fuel leakage from affected Gravity Filler Cap, and b) Pressure Refueling System is operative. 	
13-1	Tail Cone Fuel Tank System (601, 601-3A, ***) (601-3R)	С	1	0	 (M) May be inoperative provided: a) Tail Cone fuel tank refuel / defuel SOVs are verified in the closed position, and b) Tail cone fuel tank is empty. 	

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28 – <u>FU</u>	-				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
13-2	Tail Tank Fuel System (including Saddle Tanks) (604, 605, 650)	С	1	0	 (M) May be inoperative provided: a) Tail tank refuel / defuel SOVs are verified in the closed position, b) Tail tank is empty, c) Aux tank is fueled as per AMM Chapter 12, and d) CG load for the fuel distribution of all fuel added is calculated to be aft of the Take-Off Limit Line (ref. AFM).
					NOTE: It it is necessary to add fuel to the Aux tanks, precautions should be taken to keep the aircraft CG in the safe limits at all times.
13-3	Turbulator Vanes (Fuel Jettison Mast, Tail Tank Fuel Systems)	С	2	0	One or both turbulator vanes may be damaged or missing provided the Tail Cone Fuel Tank System (601, 601-3A, 601-3R) or Tail Tank Fuel System (604, 605, 650) as applicable is declared inoperative.
15-1	Transfer Ejectors (Aux Tank)	В	2	1	One may be inoperative provided Aux Tank (and Tail Tank if installed) remains empty.
15-2	Crossfeed Shutoff Valves (600, 601, ***) (601-3A, 601-3R)				
	1) Left to Right Right to Left	С	2	0	One or both may be inoperative closed provided: a) All Fuel Quantity Indications or EICAS Readouts are operative, and b) Aircraft is operated in accordance with AFM procedures if fuel crossfeed is required.

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28 – <u>FL</u>	<u>JEL</u>				4. Remarks or Exceptions Remarques ou exceptions
15-3	Tail Tank Transfer Boost Pumps (604, 605, 650)				
	1) Primary	Α	1	0	(M) May be inoperative provided:
	,				a) Pump is deactivated,
					b) Secondary pump is operative,
					c) Fuel jettison system is verified operative before each flight, and
					d) Repairs are made within three flight days.
	2) Secondary	A	1	0	 (M) May be inoperative provided: a) Pump is deactivated, b) Primary pump is operative, c) Fuel jettison system is verified operative before each flight, and d) Repairs are made within three flight days.
15-4	Crossflow Shutoff Valves				
	1) Left to Aux Right to Aux (604, 605, 650)	С	2	0	One or both may be inoperative closed provided: a) All Fuel Quantity Indications or EICAS Readouts are operative, and b) Aircraft is operated in accordance with AFM procedures if fuel crossflow is required.

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28 – <u>FU</u>	<u>JEL</u>				4. Remarks or Exceptions Remarques ou exceptions
23-1	Fuel (Boost) Pumps				
	1) 600, 601-3R	Α	2	1	(M)(O) One may be inoperative provided:
					 a) Affected fuel (boost) pump is selected off and deactivated, b) Crossfeed (Left to Right, Right to Left) SOVs are operative, c) Both fuel feed line check valves are verified operative prior to the first flight, d) If left boost pump is inoperative, APU is operated continuously throughout flight and APU generator is verified operative prior to flight, and e) Repairs are made within one flight day.
	2) 604, 605, 650	A	2	1	 (M)(O) One may be inoperative provided: a) Affected fuel (boost) pump is selected off and deactivated, b) Crossflow (Left to Aux, Right to Aux) SOVs are operative, c) Both fuel feed line check valves are verified operative prior to the first flight, d) If left boost pump is inoperative, APU is operated continuously throughout flight and APU generator is verified operative prior to flight, and e) Repairs are made within one flight day.
23-2	Fuel Boost Pump "INOP/ON" Switch Lights (light function only) (604, 605, 650)	С	2	0	

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28 – <u>FU</u>	<u>EL</u>				4. Remarks or Exceptions Remarques ou exceptions
25-1	APU Fuel Pump	С	1	0	 (M) May be inoperative provided: a) APU is operated for a maximum of 10 minutes from start-up to shutdown, b) APU is only used to start one engine, c) APU generator adapter temperature is below 40°C, or warm to the touch, prior to engine start,
					d) APU is started with the left fuel boost pump, ande) APU is shutdown and considered inoperative in flight.
		С	1	0	May be inoperative provided APU is considered inoperative and is not used.
25-2	APU Fuel Shutoff Valves	С	2	0	(M) One or both may be inoperative closed provided APU is not used.
25-3	APU Negative G Feed Check Valve.	С	1	0	(O) May be inoperative open provided flight operations are not dependent on use of the APU.
27-1	Single Point Pressure Refueling	С	1	0	(O) May be inoperative provided alternate procedures are established and used.
	System				NOTE: Any function of Pressure Refueling System that is operative may be used.
	External Refuel/Defuel Control Panel	С	1	0	May be inoperative provided the internal Refuel/Defuel control panel is operative.
	2) Internal Refuel/Defuel Control Panel	D	1	0	
	a) Fuel Quantity Display	D	1	0	(O) May be inoperative provided alternate means is used to determine the amount of fuel.

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28 – <u>FUEL</u> 4. Remarks or E	
Remarques o	
27-2 Refuel / Defuel Shut-Off Valves	
1) Left, Right, Aux C 3 0 (O) May be inoperativ	ve provided:
a) Alternate refu and used,	eling procedures are established
with full mission	ne calculated static aircraft CG on fuel load is aft of the fwd off Limit Line (ref. AFM), and
c) Fuel Quantity operative.	Indication Systems (cockpit) are
2) Tail C 2 0 (O) May be inoperativ Fuel Tanks remain en	re closed provided Aux and Tail npty.
C 2 0 (O) May be inoperativ	ve open provided:
a) Alternate refu and used,	reling procedures are established
with full mission	ne calculated static aircraft CG on fuel load is aft of the fwd off Limit Line (ref. AFM), and
c) Left, Right, Au closed.	ux Refuel/Defuel SOVs are
27-3 Refuel / Defuel C 1 0 (M) May be inoperativ	ve (missing) provided:
	el adapter is visually checked for n prior to each refueling,
b) Adapter popp each refueling	et is visually checked closed after g,
c) No leakage ca complete, and	an be detected after refueling is
d) All Refuel/Def	fuel Shut-off valves are verified each refueling.
40-1 EICAS Bulk Fuel C 1 0 (M)(O) May be inoperated	-
Indication dispatch, and	erified to be > -20°C before
(604, 605, 650) b) Total air Tem	perature is not < -50°C.

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it de systeme/serie article			"	Nombre d'articles à expédier		
28 – <u>FUEL</u>				4. Remarks or Exceptions Remarques ou exceptions		
Fuel Quantity Indicating Systems (Cockpit) (600, 601, 601-3A, 601-3R)						
1) Main Fuel Tanks Left and	С	2	1	(M) (O) One may be inoperative provided:		
Right				a) Alternate gravity refueling procedures are established and used,		
				b) Both main tanks are visually confirmed full,		
				c) Total fuel carried includes 5% more than the required fuel load for the planned flight,		
				d) Aux fuel Tank Quantity Indicator is operative or the Aux Fuel Tank is empty,		
				e) Fuel Flow Indicating Systems and associated engine instruments are operative,		
				f) Gravity Crossflow SOV is verified operative,		
				g) Crossfeed SOVs (if installed) remain closed,		
				h) Lateral fuel balance is maintained during flight,		
				i) Fuel Totalizer is considered inoperative,		
				j) After takeoff, power is set by matching fuel flow indications on both engines, and		
				k) An in-flight log of fuel burned is maintained.		
2) Aux Fuel Tank (Without Tail Tank installed)	С	1	0	(M) May be inoperative provided the Aux Fuel Tank remains empty.		
,	С	1	0	(O) May be inoperative provided:		
	-			a) Aux tank is completely filled,		
				b) Left and Right Main Fuel Tank Quantity Indicators are operative,		
				c) Fuel Flow Indicating Systems and associated engine instruments are operative,		
				d) After takeoff, power is set by matching fuel flow indications on both engines, and		
				e) An in-flight log of fuel burned is maintained.		

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28 – <u>F</u> L					Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
41-1	Fuel Quantity Indicating Systems (Cockpit) (600, 601, 601-3A, 601-3R) (Cont'd)				
	2) Aux Fuel Tank(Without Tail Tank installed) Cont'd	С	1	0	 (M)(O) May be inoperative provided: a) Aux Fuel Tank is defueled each refueling, b) Aux Fuel Tank is refueled with a known quantity of fuel, c) Required Aux Fuel Tank fuel is increased by 5%, d) Left and Right Main Fuel Tank Quantity Indicators are operative, e) Fuel Flow Indicating Systems and associated engine instruments are operative, f) Aux Fuel Tank is verified to empty at the appropriate time, g) After takeoff, power is set by matching fuel flow indications on both engines, and h) An in-flight log of fuel burned is maintained.
	3) Aux Fuel Tank (With Tail Tank installed)	С	1	0	(M) May be inoperative provided the Aux and Tail Fuel Tanks remain empty.
	4) Fuel Totalizer	С	1	0	
41-2	EICAS Aux and Total Fuel Tank Quantity Readouts (604, 605, 650)	С	2	0	(M) Both may be inoperative provided the Aux and Tail tanks remain empty.
41-3	EICAS Tail Tank and Total Fuel Quantity Readouts (604, 605, 650)	С	2	0	 (M) Both may be inoperative provided: a) Tail tank remains empty, b) Aux tank is fueled as per AMM Chapter 12, and c) Ensure that the calculated static aircraft CG with full mission fuel load is aft of the fwd aircraft Take-off Limit Line (ref. AFM).

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28 – <u>FUEL</u>					4. Remarks or Exceptions Remarques ou exceptions
41-4	Fuel Computer Channels (604, 605, 650)	В	2	1	One may be inoperative provided: a) Both Transfer Ejectors are operative, b) Both Crossflow SOVs are operative, c) Both Fuel Flow Readouts are operative, d) Fuel Used on FMS is operative, and e) Opposite IRS is operative.

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	y como como unicio				Nombre d'articles à expédier
29 – <u>HY</u>	DRAULIC POWER				4. Remarks or Exceptions Remarques ou exceptions
11-1	Hydraulic AC Motor Pump (ACMP) (Systems 1 and 2)				
	1) ACMP 1B	С	1	0	(M) May be inoperative provided:
	(604, 605, 650)				a) All other hydraulic pumps are operative,
					b) Affected pump is deactivated and selected OFF,
					c) Nosewheel Steering is operative,
					d) Both Thrust Reverser Systems are operative, and
					e) Operations are conducted in accordance with AFM Supplement (Operations with Airplane Systems Inoperative).
	2) ACMP 2B	С	1	0	(M) May be inoperative provided:
	(604, 605, 650)				a) All other hydraulic pumps are operative,
					b) Affected pump is deactivated and selected OFF,
					c) Ground Spoilers are operative,
					d) Nosewheel Steering is operative,
					e) Both Thrust Reverser Systems are operative,
					f) Both Anti-Skid System Channels are operative,
					g) Take-off or landing is not conducted from a contaminated runway, and
					h) Operations are conducted in accordance with AFM Supplement (Operations with Airplane Systems Inoperative).
11-2	Hydraulic Accumulator Pressure Gauges (Systems 1, 2 and 3)	С	3	0	(M) All may be inoperative provided accumulator precharge is checked using a suitable gauge before the first flight of each day.
11-3	Hydraulic Accumulators (Systems 1, 2 and 3)	В	3	1	Systems 1 and/or System 2 accumulator(s) may be inoperative.

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29 – <u>H</u>	YDRAULIC POWER				4. Remarks or Exceptions Remarques ou exceptions
11-4	Hydraulic Heat Exchanger Cooling Fan (600, 601, 601-3A, 601-3R)	С	1	0	May be inoperative provided ground operation of hydraulic systems 1 and 2 is limited to 30 minutes when OAT is above 45 deg C.
11-5	Engine Driven Hydraulic Pumps (604, 605, 650)	В	2	1	 (M) One may be inoperative provided: a) All other hydraulic pumps are operative, b) Hydraulic AC Motor Pump (ACMP) 1B and 2B are selected ON before each flight, c) Both Integrated Drive Generators (IDG) are operative, d) Affected pump is mechanically removed and a blanking plate is installed, and e) Operations are conducted in accordance with AFM Supplement (Operations with Airplane Systems Inoperative).
11-6	Hydraulic "AUTO" Switches (Electric Pumps) (604, 605, 650)	С	3	0	Any or all may be inoperative provided the affected pumps are manually selected on before take-off and landing.
11-7	Return Manifold Filter – Differential Pressure Indications	Α	3	0	 (M) Any or all may be inoperative provided: a) Filter element is verified free of any contaminant, and b) Repairs are made within three flight days.
12-1	Hydraulic AC Motor Pump (ACMP) 3A	С	1	0	 May be inoperative provided: a) 3B pump is operated continuously throughout the flight, and b) All other hydraulic pumps (engine driven and electrical) are operative.

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29 – <u>HY</u>	'DRAULIC POWER				4. Remarks or Exceptions Remarques ou exceptions
31-1	Hydraulic Pressure Indicators (Cockpit) (600, 601, 601-3A, 601-3R)	С	3	2	One may be inoperative provided: a) All four Electric Motor Driven Hydraulic Pumps are operative, b) Associated Hydraulic Pump Low Pressure (ELECT PUMP or L/R ENG PUMP) Caution Lights are operative, c) Associated Hydraulic Pump HI TEMP Caution Light is operative, and d) Brake Pressure Indicator is operative.
31-2	Hydraulic Reservoir Quantity Indicators (Systems 1, 2 and 3) (Cockpit) (600, 601, 601-3A, 601-3R)	С	3	0	One or more may be inoperative provided: a) Affected reservoir quantity is confirmed within limits prior to each flight, b) Associated Hydraulic Pressure Indicator is operative, c) Associated Hydraulic Pump Low Pressure (ELEC PUMP or L/H ENG PUMP) Caution Lights are operative, and d) Associated Hydraulic HI TEMP Caution Light is operative.
31-3	EICAS Hydraulic Pressure Readout Indications (604, 605, 650)	С	3	0	(O) Any or all may be inoperative provided the associated pressure switches are operative.
32-1	EICAS Hydraulic Reservoir Quantity Readouts (Systems 1, 2 and 3) (604, 605, 650)	С	3	0	(M) Any or all may be inoperative provided the quantity in the associated reservoir(s) is checked on the reservoir sight glass prior to each flight.

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34-1	Hydraulic Pump Low Pressure Switches						
	1) 1A,1B, 2A, 2B and 3A Low Pressure Switch	С	5	3	One in each system may be inoperative provided associated Hydraulic Pressure and Quantity Indicating Systems are operative.		
	2) 3B Low	С	1	0	(O) May be inoperative provided:		
	Pressure Switch				 a) Pump 3B functionality is verified prior to each flight. 		
					b) Associated hydraulic Pressure and Quantity Indicating Systems are operative.		
35-1	Hydraulic HI TEMP	С	3	2	(M) One may be inoperative provided:		
	Caution Lights (Systems 1, 2 and 3)				 a) Associated Hydraulic Pressure Indicator is operative, 		
	(600, 601, 601-3A, 601-3R)				b) Associated Hydraulic Reservoir Quantity Indicator is operative, and		
					c) Hydraulic Heat Exchanger Cooling Fan is operative.		
37-1	Hydraulic Pump Low	С	6	3	One per system may be inoperative provided:		
	Pressure Caution Lights (light function				 a) Associated Hydraulic Pressure Indicator is operative, 		
	only) (ELEC PUMP, L ENG PUMP, R				b) Associated Hydraulic Reservoir Quantity Indicator is operative, and		
	ENG PUMP) (600, 601, 601-3A, 601-3R)				c) Associated Hydraulic HI TEMP Caution Light is operative.		

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30 – ICE AND RAIN PROTECTION 12-1 Wing Anti-ice System 1) Auto Control (600)		1		4. Remarks or Exceptions
PROTECTION 12-1 Wing Anti-ice System 1) Auto Control (600)		1		
12-1 Wing Anti-ice System 1) Auto Control (600)		1		Remarques ou exceptions
1) Auto Control (600)		1		rtomarquoo ou oxooptiono
(600)		1		
	- 1		0	(O) May be inoperative provided Wing Anti-ice System Manual Control is operative.
	C	1	0	May be inoperative provided:
				a) Both Ice Detection Systems are operative, and
				b) Flight is not conducted in known or forecast
				icing conditions.
2) Manual Control	С	1	0	May be inoperative provided Wing Anti-ice System
(600)		•		Auto Control is operative.
	С	1	0	May be inoperative provided:
				a) Both Ice Detection Systems are operative, and
				b) Flight is not conducted in known or forecast
				icing conditions.
3) Normal Control (601) (601-3A, 601-3R with SB 601- 0575) (604 with SB 604-30-002 and aircraft S/N 5583 and subsequent) (605, 650)	С	1	0	(O) May be inoperative provided Wing Anti-ice System Standby Control is operative.
	c	1	0	May be inoperative provided:
				a) Both Ice Detection Systems are operative, and
				b) Flight is not conducted in known or forecast
				icing conditions.
4) Normal Control	С	1	0	May be inoperative provided:
(601-3A,				a) Both Ice Detection Systems are operative, and
601-3R <u>without</u>				b) Flight is not conducted in known or forecast
SB 601-0575)				icing conditions.
001-03/3)				
(604 <u>without</u> SB 604-30-002)				
(Cont'd)				

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14° de s	14- de systeme/sene anticle				Nombre d'articles à expédier			
	E AND RAIN ROTECTION				4. Remarks or Exceptions Remarques ou exceptions			
12-1	Wing Anti-ice System (Cont'd)							
	5) Standby Control (601, 601-3A, 601-3R, 604, 605, 650)	С	1	0	May be inoperative provided Wing Anti-ice System Normal Control is operative.			
		С	1	0	May be inoperative provided:			
					a) Both Ice Detection systems are operative, and b) Flight is not conducted in known or forecast icing conditions.			
12-2	Wing Anti-ice System Modulating and Shut-off Valves	С	2	0	One or both may be inoperative CLOSED provided: a) Both Ice Detection Systems are operative, and b) Flight is not conducted in known or forecast icing conditions.			
12-3	Lower Bleed Air Isolation Valve (600)	С	1	0	May be inoperative OPEN.			
		С	1	0	May be inoperative CLOSED provided: a) Both Ice Detection Systems are operative, and b) Flight is not conducted in known or forecast icing conditions.			
12-4	14 th Stage Engine Bleed Air Isolation Valve (601, 601-3A, 601-3R, 604, 605, 650)	С	1	0	May be inoperative OPEN.			
		С	1	0	May be inoperative CLOSED provided: a) Both Ice Detection Systems are operative, and b) Flight is not conducted in known or forecast icing conditions.			

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			4. Remarks or Exceptions			
			Remarques ou exceptions			
C	1	0	(M)(O) May be inoperative provided:			
O	'		a) The SUPP GND WING ANTI-ICE SYS OFF			
			switch is selected to OFF, and			
			b) The wing anti-ice system must be selected and confirmed ON for take-off when, the OAT is 5°C (41°F) or below.			
			NOTE 1: This applies regardless of environmental conditions.			
			NOTE 2: If the wing anti-ice system is selected ON			
			for take-off, the cowl anti-ice system must also be selected ON.			
			NOTE 3: When Type II, Type III or Type IV anti-icing fluids have been applied, the wing anti-ice system must only be selected and confirmed ON just prior to thrust increase for takeoff.			
С	1	0	 (M)(O) May be inoperative as indicated by continuous illumination of TEMP 'FAIL LO' amber light provided: a) The SUPP GND WING ANTI-ICE SYS OFF switch is selected to OFF, b) Prior to dispatch, ensure wing leading edges are clean, anti-icing fluid is re-applied if required, and c) The wing anti-ice system must be selected and confirmed ON for take-off when, the OAT is 5°C (41°F) or below. NOTE 1: This applies regardless of environmental 			
			conditions. NOTE 2: If the wing anti-ice system is selected ON for take-off, the cowl anti-ice system must			
			also be selected ON.			
			NOTE 3: When Type II, Type III or Type IV anti-icing fluids have been applied, the wing anti-ice system must only be selected and confirmed ON just prior to thrust increase for takeoff.			
	C	C 1	C 1 0			

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Nº de système/série article				3.	Number Required For Dispatch
	E AND RAIN ROTECTION				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
12-5	Low Temperature Ground Wing Anti- ice System (LTGWAIS) (601, 601-3A, 601-3R aircraft with SB 601-0641; 604 aircraft with SB 604-30-004 or SB 604-30-006; 605 aircraft with SB 605-30-001 or SB 605-30-003; 650 aircraft with or without SB 650-30-001)	С	1	0	 (M)(O) May be inoperative as indicated by continuous illumination of TEMP 'FAIL HI' amber light provided: a) The SUPP GND WING ANTI-ICE SYS OFF switch is selected to OFF, b) Prior to dispatch, ensure wing leading edges are clean, anti-icing fluid is re-applied if required, and c) The wing anti-ice system must be selected and confirmed ON for take-off when, the OAT is 5°C (41°F) or below. NOTE 1: This applies regardless of environmental conditions. NOTE 2: If the wing anti-ice system is selected ON for take-off, the cowl anti-ice system must also be selected ON. NOTE 3: When Type II, Type III or Type IV anti-icing fluids have been applied, the wing anti-ice system must only be selected and confirmed ON just prior to thrust increase for takeoff.
20-1	Engine Cowl Anti- Ice Pressure Regulating and Shutoff Valves	С	2	1	 (M) One may be inoperative provided: a) Affected valve is secured closed, b) Both Ice Detection Systems are operative, c) Associated ACU is selected ON above FL 400, and d) Flight is not conducted in known or forecast icing conditions.

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	E AND RAIN OTECTION				4. Remarks or Exceptions Remarques ou exceptions
20-2	Engine Cowl Anti- Ice Pressure Relief Blow-off Valves				
	1) (601, 601-3A,	С	2	1	One may be inoperative OPEN provided:
	601-3R)				a) Associated Thrust Reverser is operative,
					 b) Associated Bleed Air 14th Stage SOV is operative, and
					 c) Operations are conducted in accordance with AFM Supplement 14, Operations with Airplane Systems Inoperative.
	2) (604, 605, 650)	С	2	1	One may be inoperative OPEN provided:
					a) Associated Thrust Reverser is operative,
					 b) Associated Bleed Air 14th Stage SOV is operative, and
					 c) Operations are conducted in accordance with AFM Supplement 8, Operations with Airplane Systems Inoperative.
31-1	Probe Heaters				
	1) Pitot Probes	В	2	1	(M) Except for RVSM Operations, one of each may be inoperative provided:
					a) Both Ice Detection Systems are operative,
					 Flight is not conducted in known or forecast icing conditions or visible moisture,
					c) Aircraft is operated in day VMC only, and
					d) Standby Pitot Probe Heater is operative.
	2) Static Probes	В	2	1	(M) Except for RVSM Operations, one of each may be inoperative provided:
					a) Both Ice Detection Systems are operative,
					 Flight is not conducted in known or forecast icing conditions or visible moisture, and
					c) Aircraft is operated in day VMC only.
	Cont'd				

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_	System & Sequence No Item				Nome	ore d'articles installés
Nº de système/série article				J.	Number Required For Dispatch Nombre d'articles à expédier	
30 – <u>IC</u>	E AND	RAIN				4. Remarks or Exceptions
		CTION				Remarques ou exceptions
31-1		bbe Heaters ont'd)				
	3)	Standby Pitot Probe	С	1	0	(M) Except for RVSM Operations, one of each may be inoperative provided:
		(604, 605, 650)				a) Both Ice Detection Systems are operative,
						b) Flight is not conducted in known or forecast icing conditions or visible moisture,
						c) Aircraft is operated in day VMC only, and
						d) Both Pitot Probes Heaters are operative.
	4)	TAT Probe	В	1	0	(M) May be inoperative provided:
						a) Both Ice Detection Systems are operative,
						b) Flight is not conducted in known or forecast icing conditions or visible moisture, and
						c) Aircraft is operated in day VMC only.
	5)	Angle of Attack	В	2	1	(M) May be inoperative provided:
		Vanes				a) Both Ice Detection Systems are operative,
						b) Flight is not conducted in known or forecast icing conditions or visible moisture, and
						c) Aircraft is operated in day VMC only.
	6)	AUX Angle of	В	1	0	(M) May be inoperative provided:
		Attack Case /				a) Both Ice Detection Systems are operative,
		Vane Heater (604, 605, 650)				b) Flight is not conducted in known or forecast icing conditions or visible moisture, and
						c) Aircraft is operated in day VMC only.

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Nº de s	système/série article			J .	Nombre d'articles à expédier
30 – <u>ICE AND RAIN</u> <u>PROTECTION</u>					4. Remarks or Exceptions Remarques ou exceptions
31-2	Speed Command Attitude Thrust (SCAT) Vane Heater STC SA4770SW	С	1	0	(M) May be inoperative provided:a) SCAT Vane Heater is deactivated, andb) Airplane is not operated in known or forecast icing conditions.
41-1	Windshield and Side Window Anti-Ice Controllers	С	4	3	One may be inoperative provided: a) Operations are not conducted in known or forecast icing conditions, and b) Pilot's (Left) side window heating is operative.
		С	4	2	Two may be inoperative provided: a) Operations are not conducted in known or forecast icing conditions, b) Pilot's (Left) side window heating is operative, and c) Both Ice Detection Systems are operative.
41-2	Enhanced Vision System - Ice Protection (605 aircraft with SB 605-31-002 and SB 605-34-010) *** (650 aircraft with SB 650-31-002 and SB 650-34-006) ***				
	1) EVS Fairing & IR Window heat (EVS Fairing DE- ICE function/ IR Window ANTI- ICE function)	С	2	0	May be inoperative provided the EVS is not used if EVS image is affected. NOTE: With IR Window heat inoperative, the image may degrade in high humidity or icing conditions, to the point of disappearance, as IR Window misting <i>I</i> icing increases. The EVS must not be used if the image is affected.

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30 – <u>ICE AND RAIN</u> <u>PROTECTION</u>				4. Remarks or Exceptions Remarques ou exceptions
41-2 Enhanced Vision System - Ice Protection (605 aircraft with SB 605-31-002 and SB 605-34-010) *** (650 aircraft with SB 650-31-002 and SB 650-34-006) *** (Cont'd)				
1) EVS Fairing & IR Window heat (EVS Fairing DE- ICE function/ IR Window ANTI- ICE function) (Cont'd)				
	D	2	0	(M) One or both may be inoperative provided IR Window and EVS fairing are removed as an assembly and replaced with an approved blanking plate.
2) IR Window - DE-MIST function	С	1	0	May be inoperative provided EVS is not used if EVS image is affected.
				NOTE: The image may degrade in high humidity or icing conditions, to the point of disappearance, as IR Window misting / icing increases. The EVS must not be used if the image is affected.
	С	1	0	May be inoperative provided: a) Procedures do not require the use of EVS system, b) EVS IR Window heat is considered inoperative, and c) The EVS system (imaging) is considered inoperative. (ref: MMEL item 34-32-2).
(Cont'd)				

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				Nombre d'articles à expédier
30 – <u>ICE AND RAIN</u> <u>PROTECTION</u>				4. Remarks or Exceptions Remarques ou exceptions
41-2 Enhanced Vision System - Ice Protection (605 aircraft with SB 605-31-002 and SB 605-34-010) *** (650 aircraft with SB 650-31-002 and SB 650-34-006) *** (Cont'd)				
3) EVS Fairing Temperature Sensors	D	2	1	One may be inoperative.
	С	2	0	Both may be inoperative provided EVS fairing heat is considered inoperative.
4) IR Window Temperature Sensors	D	2	1	One may be inoperative.
Consolic	С	2	0	Both may be inoperative provided EVS IR Window heat is considered inoperative.

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No de s	système/série article			3.	Number Required For Dispatch
30 10	E AND RAIN				Nombre d'articles à expédier 4. Remarks or Exceptions
	ROTECTION				Remarques ou exceptions
80-1	Ice Detection	С	2	1	(O) One may be inoperative provided:
	Systems				a) Affected ice detector is deactivated.
					 Anti-ice systems are turned ON when icing conditions exist as defined in AFM.
		Α	2	0	Both may be inoperative provided:
					a) Flight is not conducted in known or forecast icing conditions, and
					b) Repairs are made within one flight day.

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30 – ICE AND RAIN			4. Remarks or Exceptions
<u>PROTECTION</u>			Remarques ou exceptions
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					Nombre d'articles à expédier
	NDICATING/ RECORDING SYSTEMS				4. Remarks or Exceptions Remarques ou exceptions
14-1	Master Warning Switch Lights (Glareshield) (light function only) (604)	С	2	1	
14-2	Master Caution Switch Lights (Glareshield) (light function only) (604)	С	2	1	
14-3	Master Warning/Caution Pushbutton Annunciators (Glareshield) (605, 650)	С	2	1	
	Warning Lights (light function only)	С	2	1	
	2) Caution Lights (light function only)	С	2	1	

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31 – <u>INDICATING/</u> <u>RECORDING SYSTEMS</u>					Remarques ou exceptions
31-1	Flight Data Recorder (FDR) System				
	1) (If CVR and	Α	1	0	May be inoperative provided:
	FDR are				a) Cockpit voice recorder is operative, and
	required by regulations)				b) Repairs are made within three flight days.
	2) DFDR Recording	Α	-	-	Up to three digital recording parameters may be inoperative provided:
	Parameters				a) Cockpit voice recorder is operative, and
	required by				b) Repairs are made within twenty calendar days.
	Regulations				27 Topano dio mano maniny dalomasi asyon
	3) DFDR Recording Parameters not required by Regulations	A	-	-	May be inoperative provided repairs are made before the completion of the next heavy maintenance visit.
	If not required by regulations	D	-	0	
31-2	Quick Access Recorder (QAR)***	С	1	0	(O)(M) May be inoperative for Flight Data Monitoring (FDM) purposes, provided approved alternate procedures, if appropriate to the FDM programme, are established and used.
		D	1	0	May be inoperative provided procedures do not require its use.
40-1	Cursor Control Panel (CCP) (605, 650)	С	2	1	(O) Right side may be inoperative.
	1) Joystick	С	2	1	
		С	2	0	May be inoperative provided:
					a) Maintenance Diagnostic Computer is considered inoperative,
					b) If installed, Graphical Weather function is considered inoperative, and
					c) If installed, Electronic Charts are considered inoperative.
	(Cont'd)				

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					Nombre d'articles à expédier
31 – <u>INDIC</u> RECC	CATING/ ORDING SYSTEMS				4. Remarks or Exceptions Remarques ou exceptions
(C0 (60	rsor Control Panel CP) 05, 650) ont'd)				
2)	JSTK Pushbuttons	С	2	0	
3)	MEM Pushbuttons	D	6	0	
4)	CHART Pushbuttons	С	2	1	
		С	2	0	May be inoperative provided If installed Electronic Charts are considered inoperative.
5)	ZOOM Pushbuttons	С	2	1	
		С	2	0	 May be inoperative provided: a) Maintenance Diagnostic Computer is considered inoperative, b) If installed, Graphical Weather function is considered inoperative, and c) If installed Electronic Charts are considered inoperative.
6)	Orient Pushbuttons (located under the CHART pushbutton)	С	2	1	
		С	2	0	 May be inoperative provided: a) Maintenance Diagnostic Computer is considered inoperative, b) If installed, Graphical Weather function is considered inoperative, and c) If installed, Electronic Charts are considered inoperative.
(Co	ont'd)				

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Nº ue s	systemersene article				Nombre d'articles à expédier
	DICATING/ ECORDING SYSTEMS				4. Remarks or Exceptions Remarques ou exceptions
40-1	Cursor Control Panel (CCP) (605, 650) (Cont'd)				
	7) LWR FRMT Pushbuttons	С	2	0	
	8) UPR MENU Pushbuttons	С	2	1	
	9) LWR MENU Pushbuttons	С	2	1	(O) One may be inoperative.
	10) ESC Pushbuttons	С	2	0	
	11) PUSH SELECT Pushbuttons (Menu)	С	2	1	
	12) MENU ADV knobs	С	2	1	
	13) DATA knobs (Menu)	С	2	1	
	14) TFC Pushbuttons	С	2	0	
	15) TR/WX Pushbuttons	С	2	0	
	16) SUMRY Pushbuttons	С	2	0	
	17) AC ELEC Pushbuttons	С	2	0	
	(Cont'd)				

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	DICATING/ ECORDING SYSTEMS				4. Remarks or Exceptions Remarques ou exceptions
40-1	Cursor Control Panel (CCP) (605, 650) (Cont'd)				
	18) HYD Pushbuttons	С	2	0	
	19) DC ELEC Pushbuttons	С	2	0	
	20) FLT Pushbuttons	С	2	0	
	21) CAS Pushbuttons	С	2	1	Right side may be inoperative.
	22) RADIO Pushbuttons	С	2	1	
	23) FREQ Pushbuttons	С	2	1	
	24) 1/2 Pushbuttons	С	2	1	
	25) DME-H Pushbuttons	С	2	1	
	26) IDENT Pushbuttons	С	2	1	
	27) ATC Pushbuttons	С	2	1	
	(Cont'd)				

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	DICATING/ ECORDING SYSTEMS				4. Remarks or Exceptions
<u> </u>	ECORDING STSTEMS				Remarques ou exceptions
40-1	Cursor Control Panel (CCP) (605, 650) (Cont'd)				
	28) PUSH SELECT Pushbuttons (Radio)	С	2	1	
	29) RADIO ADV knobs	С	2	1	
	30) DATA knobs (Radio)	С	2	1	
41-1	EICAS Control Panel (ECP) Discrete Buttons (604)				
	Without SB 604-34-033				
	 CKLST RCL EMER HYD ELEC UP DN SKP F/CTL Joystick 	B B B B B B B	1 1 1 1 1 1 1 1	0 0 0 0 0 0 0	May be inoperative provided UP and DN Discrete Buttons are operative.
	(Cont'd)				

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	IDICATING/				4. Remarks or Exceptions
RI	ECORDING SYSTEMS				Remarques ou exceptions
41-1	EICAS Control				
	Panel (ECP)				
	Discrete Buttons				
	(604)				
	(Cont'd)				
	With SB 604-34-033				
	1) CKLST	В	1	0	
	2) RCL	В	1	0	
	3) EMER	В	1	0	
	4) HYD	В	1	0	
	5) ELEC	В	1	0	
	6) UP/PLAN	В	1	0	
	7) DN/SIDE	В	1	0	
	8) SKP/3D	В	1	0	
	9) F/CTL	В	1	0	
	10) Joystick	В	1	0	May be inoperative provided UP/PLAN and DN/SIDE Discrete Buttons are operative.
41-2	Data Concentration	С	-	2	(O) Any in excess of two may be inoperative provided:
	Unit (DCU) (604)				 a) Associated AUDIO WARNING DISABLE switch is selected, and
					 b) Two separate audio warning channels are verified operative before each flight.
					NOTE 1: In the event of a DCU 1 failure, both pilots' Master Warning and Master Caution lights will not test during the lamp 1 test.
					NOTE 2: In the event of a DCU 2 failure, both pilots' Master Warning and Master Caution lights will not test during the lamp 2 test.
					NOTE 3: Aircraft without SB 604-31-003 installed: In the event of either a DCU 1 or 2 failure, a subsequent TRU 2 failure will generate one "HOT" icon ITT engine indication. This icon should be disregarded and the ITT monitored.

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Nº UC	systemerserie article				Nombre d'articles à expédier	
	DICATING/ ECORDING SYSTEMS				4. Remarks or Exceptions Remarques ou exceptions	
50-1	Auto CB Fail Light (10-Channel Annunciator) (600, 601, 601-3A, 601-3R)	В	1	0	May be inoperative provided the BRT / DIM function on the 10-channel annunciator panel is confirmed operative prior to each flight.	
50-2	Voice Advisory System (600, 601, 601-3A, 601-3R) ***	С	-	0	(M)(O) May be inoperative provided:a) System is secured, andb) Alternate procedures are established and used.	
		D	-	0	(M) May be inoperative provided:a) System is secured, andb) Procedures do not require its use.	I
50-3	Angle of Attack Indication System (Speed Cue Indexer) (600, 601, 601-3A, 601-3R)	С	-	0	(M)(O) May be inoperative provided:a) System is deactivated, andb) Alternate procedures are established and used.	
50-4	Speed Command Attitude Thrust (SCAT) System (600, 601, 601-3A, 601-3R)	С	-	0	(O) May be inoperative provided alternate procedures are established and used.	
	***	D	-	0	(M) May be inoperative provided:a) System is secured, andb) Procedures do not require its use.	I

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it ac systemorserie article					Nombre d'articles à expédier
	DICATING/ CORDING SYSTEMS				4. Remarks or Exceptions Remarques ou exceptions
50-5	Canadian Marconi Corp Omega Nav Status Display System WARNING Repeater lights (600)	D	2	1	(M) May be inoperative provided:a) Light is not illuminated, andb) All individual warning lights are operative.
60-1	Clocks	С	-	0	(O) May be inoperative provided:
					a) A reliable and functioning time-piece is readily available, and
					b) Alternate procedures are established and used to determine elapsed time.
	1) Universal Time Co-ordination Display (UTC) (605, 650)	С	1	0	
	2) Flight Time (FT) (605, 650)	С	1	0	
	3) Chronometer (CHR) (605, 650)	С	1	0	(O) May be inoperative provided alternate procedures are established and used to determine elapsed time.
	4) LCD Display Segments / Mode (Annuns.) (605, 650)	С	-	-	Individual segments or annunciations may be inoperative provided flight crews can readily determine mode of operation.

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					Nombre d'articles à expédier	
	DICATING/ CORDING SYSTEMS				4. Remarks or Exceptions Remarques ou exceptions	
61-1	Display Units					
	1) (604) A/C without SB 604-34-031)	В	6	5	EICAS display unit 1 or 2 may be inoperative provided the display reversionary mode selector switches are operative.	
	2) (604) For A/C 5500 and up or with SB 604-34-031 installed	В	6	5	(O) EICAS display unit 1 or 2 may be inoperative provided the display reversionary mode selector switches are operative.	
61-2	Adaptive Flight Displays (AFD) (605, 650)	В	4	3	(O) Right Inboard Display may be inoperative provided: a) Two Control Display Units are operative, and b) Unaffected displays reversion capabilities are verified operative before the first flight of the day.	

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14 46	systemerserie article				Nombre d'articles à expédier
32 – <u>L</u>	ANDING GEAR				4. Remarks or Exceptions Remarques ou exceptions
30-1	Landing Gear Selector Handle Anti-Retraction Mechanism	С	1	0	(M) May be inoperative in the LOCKED position (down) provided the downlock release mechanism is operative.
30-2	Landing Gear	Α	1	0	(M) May be inoperative provided:
	Retraction System (600, 601, 601, 601, 601, 605, 604, 604, 604, 604, 604, 604, 604, 604				 a) Operations are not conducted in known or forecast icing conditions,
	601-3R, 604, 605, 650)				 b) Ground lock pins are installed to ensure all three landing gears are locked down throughout flight,
					c) In-flight performance information given in the Flight Crew Operating Manual (FCOM), is used,
					d) Extended overwater operations are prohibited,
					e) Both pilot headsets are worn,
					f) Flight Compartment and Cabin Interphone Systems are operative,
					g) Both Flap Power Drive Units are Operative,
					h) CAT II operations are prohibited, and
					i) Repairs are made within one flight day.
					j) Operations are conducted in accordance with the appropriate AFM Supplement:
					AFM Supplement 16, Flight with Landing Gear Down (600, 601, 601-3A, 601-3R).
					AFM Supplement 6, Flight with Landing Gear Down (604, 605, 650).
43-1	Brake Accumulator Pressure Gauges	С	2	0	(M) One or both may be inoperative provided the accumulator pre-charge pressure is checked using a suitable pressure gauge before the first flight of each day.

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No de	e système/série article			3.	Number Required For Dispatch
32 – <u>I</u>	_ANDING GEAR				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
43-2	Brake Pressure Indications (600, 601, 601-3A, 601-3R)	С	2	0	 (M) One or both may be inoperative provided: a) Brake accumulator(s) nitrogen pressure is checked prior to the first flight of each day, b) Capability of brake accumulators to retain adequate hydraulic fluid for brakes is verified prior to the first flight of each day, and c) Hydraulic Pressure Indication is operative.
44-1	Anti-Skid System Channels (604, 605, 650)	В	2	1	 (M)(O) Either the inboard or the outboard channel may be inoperative provided: a) Inoperative channel is deactivated, b) Anti-skid is selected ON for take-off and landing, c) Nosewheel Steering is operative, d) Ground Spoilers are operative, e) Both Thrust Reversers are operative, f) Operations are conducted in accordance with AFM Supplement 8, Operations with Airplane Systems Inoperative.
45-1	Parking Brake Light (600, 601, 601-3A, 601-3R)	С	1	0	(O) May be inoperative provided the amber anti-skid fail lights INBD FAIL and OUTBD FAIL illuminate as the parking brake is set.
47-1	EICAS Brake Pressure Indications (604, 605, 650)	С	2	0	 (M) One or both may be inoperative provided: a) Brake accumulator(s) nitrogen pressure is checked prior to the first flight of the day, b) Capability of brake accumulators to retain adequate hydraulic fluid for brakes is verified prior to the first flight of the day, and c) Hydraulic Pressure Indication is operative.

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32 – <u>L</u>	<u>Anding Gear</u>				4. Remarks or Exceptions Remarques ou exceptions
50-1	Nosewheel Steering System	С	1	0	(M) May be inoperative provided:
	System				a) Nosewheel steering system is selected off,b) Nose landing gear solenoid selector valve is not failed open,
					c) Nosewheel is confirmed to be in a free- castoring mode,
					d) Crosswind limit is 20 kts for take-off and landing,
					e) Take-off and accelerate-stop distances are increased by 2 %,
					f) Accelerate-stop distance available and runway length available are reduced by 2 % when determining the AFM fields length limited take-off weight and V1 speed, and
					g) Take-off or landing is not conducted on a contaminated runway.
					NOTE 1. Care should be taken when operating in congested areas, or on contaminated surfaces when using asymmetric thrust.
					NOTE 2. Excessive use of brakes may lead to fuse plug failure(s).

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					Nombre d'articles à expédier		
32 – <u>L</u>	<u>ANDING GEAR</u>				4. Remarks or Exceptions Remarques ou exceptions		
50-2	Remote Nose Gear Door Release	D	1	0	(M) May be inoperative in the "DOOR CLOSED" position provided:		
	Mechanism (Additional Handle				 a) Mechanism is verified to be in the "DOOR CLOSED" position, and 		
	and cable mechanism) ***				b) Actuator arm is secured in the "DOOR CLOSED" position.		
		D	1	0	(M) May be inoperative in the "DOOR CLOSED" position provided:		
					a) Mechanism is verified to be in the "DOOR CLOSED" position, and		
					b) Actuator arm is removed from the nose wheel well fitting.		

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33 – <u>Ll</u> (<u>GHTS</u>				4. Remarks or Exceptions Remarques ou exceptions
12-1	Lighting Control Panels (Center Pedestal & L/R Facia Panels) (600, 601) 1) Floodlight System				
	a) BRT/DIM	С	3	0	Any or all may be inoperative provided:
	Systems				a) BRT/DIM switch is selected to OFF position, and
					b) Brightening System is operative.
		С	3	0	May be inoperative provided:
					a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and
					b) Ensure that direct rays are shielded from flight crewmember's eyes.
	b) Brightening Systems	С	3	0	Any or all may be inoperative provided the associated BRT/DIM system is operative.
		С	3	0	May be inoperative provided:
					a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and
					b) Ensure that direct rays are shielded from flight crewmember's eyes.
	2) Instrument	С	3	0	May be inoperative provided:
	Brightening Systems				a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and
					b) Ensure that direct rays are shielded from flight crewmember's eyes.
	3) Floor Light	С	2	0	May be inoperative provided:
	Systems				a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and
					b) Ensure that direct rays are shielded from flight crewmember's eyes.

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33 – <u>LIG</u>	<u>SHTS</u>				4. Remarks or Exceptions Remarques ou exceptions
12-2	Lighting Control Panels (Center Pedestal & L/R Facia Panels) (601-3A, 601-3R)				
	1) Floodlight System				
	a) BRT/DIM Systems	С	3	0	Any or all may be inoperative provided: a) BRT/DIM switch is selected to OFF position, and b) Brightening System is operative.
		С	3	0	May be inoperative provided: a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes.
	b) Brightening Systems	С	3	0	Any or all may be inoperative provided the associated BRT/DIM system is operative.
		С	3	0	May be inoperative provided:
					 a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes.
					Grewmeniber 3 eyes.
	2) Instrument	С	3	0	May be inoperative provided:
	Brightening Systems				 a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and
					 Ensure that direct rays are shielded from flight crewmember's eyes.
	(Cont'd)				

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33 – <u>LIC</u>	<u>GHTS</u>				4. Remarks or Exceptions Remarques ou exceptions	
12-2	Lighting Control Panels (Center Pedestal & L/R Facia Panels) (601-3A, 601-3R) (Cont'd)					
	3) Digits Brightening System	С	1	0	May be inoperative provided:	
	4) Floor Light Systems	С	2	0	May be inoperative provided: a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes.	
12-3	PBA BRT/DIM System (10-Channel Annunciator Panel) (600, 601,601-3A, 601-3R)	С	1	0	May be inoperative provided the DIM mode is operative for night operations.	
	001-3IX)	С	1	0	May be inoperative provided the BRT mode is operative for day operations.	
12-4	Overhead Instrument Panel Brightening System	С	1	0	May be inoperative provided: a) Cockpit lighting and intensity is sufficient for flight crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes. 	

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33 – <u>LIGHTS</u>				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
12-5 Lighting Control Panels (Center Pedestal) (604, 605, 650)				
1) Floodlight Brightening Systems (604, 605, 650)	С	3	0	May be inoperative provided: Cockpit lighting and intensity is sufficient for flight crewmembers to perform their duties throughout flight, and Ensure that direct rays are shielded from flight crewmember's eyes.
2) Integrated Brightening Systems (604, 605, 650)	С	3	0	May be inoperative provided: a) Cockpit lighting and intensity is sufficient for flight crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes.
3) Display Brightening Systems				
a) (604)	С	3	0	May be inoperative provided: a) Cockpit lighting and intensity is sufficient for flight crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes.
b) PFD/MFD Brightening Systems (605, 650)	С	4	0	May be inoperative provided: a) Cockpit lighting and intensity is sufficient for flight crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes.
(Cont'd)				

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33 – <u>I</u>	<u>lights</u>				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
12-5	Lighting Control Panels (Center Pedestal) (604, 605, 650) (Cont'd)				
	4) Floor Light Systems (604, 605, 650)	С	2	0	May be inoperative provided: a) Cockpit lighting and intensity is sufficient for flight crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes.
	5) Circuit Breaker Panel Brightening System (604)	С	1	0	May be inoperative provided: Cockpit lighting and intensity is sufficient for flight crewmembers to perform their duties throughout flight, and Ensure that direct rays are shielded from flight crewmember's eyes.
	6) Circuit Breaker Lighting ON/OFF Switch (605, 650)	С	1	0	May be inoperative provided: a) Cockpit lighting and intensity is sufficient for flight crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes.
12-6	Indicator (PBA) Lighting BRT/DIM System (604, 605, 650)	С	1	0	May be inoperative provided: Cockpit lighting and intensity is sufficient for flight crewmembers to perform their duties throughout flight, and Ensure that direct rays are shielded from flight crewmember's eyes.

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33 – <u>LIC</u>	<u>GHTS</u>				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions	
13-1	Cockpit and Instrument Panel Lighting Systems (Excluding EFIS)	С	-	-	 Individual lights may be inoperative provided: a) Sufficient lighting is operative to make each required instrument, control, and other device for which it is provided easily readable, b) Direct rays and reflections do not impair the pilots' view either inside or outside the aircraft, and c) Lighting intensity can be controlled or preset to a satisfactory level for the expected flight conditions. 	
		С	-	0	May be inoperative for day operations.	I
14-1	Map Reading Lights 1) 600, 601,	С	2	0	May be inoperative provided:	
	601-3A 601-3R, 604				 a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes. 	
	2) 605, 650	С	4	0	May be inoperative provided: a) Cockpit lighting and intensity is sufficient for crewmembers to perform their duties throughout flight, and b) Ensure that direct rays are shielded from flight crewmember's eyes.	
20-1	Cabin Interior Lights	С	-	-	May be inoperative provided: a) Cabin Emergency Lighting is operative, and b) Sufficient lighting is operative to permit the crew to perform their required duties.	

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33 – <u>I</u>	<u>LIGHTS</u>				4. Remarks or Exceptions Remarques ou exceptions
20-2	Miscellaneous Lights				
	1) Boarding and Dome Lights (604)	D	-	0	
	2) Boarding and Cabin Lights (605, 650)	D	-	0	
20-3	Lavatory Vanity / Reading Lights	D	-	0	
23-1	Fasten Seat Belt/No Smoking/Return to Seat Lights/Signs	С	-	0	(O) Passenger seats, flight attendant seats or lavatories from which a light/sign is not readily legible shall not be occupied and must be blocked and placarded "DO NOT OCCUPY".
		С	-	0	(O) May be inoperative provided alternate procedures are established and used to notify cabin occupants.
25-1	Push Button Annunciators (Light function only) (604, 605, 650)	В	-	-	One lamp may be inoperative in each independent module provided the nomenclature remains legible.
31-1	Service Lights	D	-	0	

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	33 – <u>LIGHTS</u>					Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
40-1		ition Lights vigation)				
	1)	Wing Tip Position Light Bulbs	С	4	2	Two may be inoperative provided at least one bulb is operative at each position.
			С	4	0	All may be inoperative provided aircraft is not operated at night.
	2)	Upper Aft Position Light Bulb	С	1	0	May be inoperative provided Lower Aft Position Light is operative.
			С	1	0	May be inoperative provided aircraft is not operated at night.
	3)	Lower Aft Position Light Bulbs				
		a) 600, 601, 601-3A	С	2	1	
			С	2	0	Both may be inoperative provided Upper Aft Position Light is operative.
			С	2	0	Both may be inoperative provided aircraft is not operated at night.
		b) 601-3R, 604, 605, 650	С	1	0	May be inoperative provided Upper Aft Position Light is operative.
			С	1	0	May be inoperative provided aircraft is not operated at night.

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33 – <u>LIC</u>	<u>GHTS</u>				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions	
40-2	High Intensity or Strobe Light System	С	1	0	May be inoperative provided an approved Anti-Collision Beacon Light System is installed and operative.	
		С	-	0	May be inoperative provided aircraft is not operated at night.	I
40-3	Anti-Collision Beacon Light System	С	-	0	May be inoperative provided an approved High Intensity or Strobe Light System is installed and operative.	
		С	-	0	May be inoperative provided aircraft is not operated at night.	I
40-4	Landing Lights					
	1) 600, 601	С	2	1	One may be inoperative provided both Taxi / Recognition Lights are operative.	
		С	2	0	Both may be inoperative provided aircraft is not operated at night.	
	2) 601-3A, 601-3R, 604, 605, 650	С	4	3		
		С	4	1	Two or three may be inoperative provided both Taxi / Recognition Lights are operative.	
		С	4	0	All may be inoperative provided aircraft is not operated at night.	
	3) Pulse Light Function (605, 650)	С	2	0	(O) May be inoperative provided Pulselight circuit breaker is pulled and collared.	

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33 – <u>Ll</u>	<u>GHTS</u>				4. Remarks or Exceptions Remarques ou exceptions
40-5	Pulselight Landing Light System				
	1) (600, 601, 601- 3A, 601-3R, 604)	С	1	0	(M) May be inoperative.
40-6	Logo Light System ***	D	-	0	
40-7	Taxi / Recognition Lights				
	1) 600, 601	С	2	0	One or both may be inoperative provided both Landing Lights are operative.
		С	2	0	May be inoperative provided aircraft is not operated at night.
	2) 601-3A, 601-3R, 604, 605, 650	С	2	0	One or both may be inoperative provided at least three Landing Lights are operative.
		С	2	0	May be inoperative provided aircraft is not operated at night.

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33 – <u>LIC</u>	<u>GHTS</u>				Remarks or Exceptions Remarques ou exceptions	
43-1	Wing Inspection	С	2	0	May be inoperative provided:	
	Lights				 a) Aircraft is not operated in known or forecast icing conditions at night, and 	
					 b) Ground deicing procedures do not require their use. 	
		С	2	0	 (O) May be inoperative provided: a) Both Ice Detectors are operative, b) Wing ANTI-ICE is turned ON below 22,000 feet. c) when icing conditions as defined in the AFM exist or are anticipated, and d) Ground de-icing procedures do not require their use. 	
50-1	Exterior Emergency Lights	С	4	0	Any may be inoperative provided aircraft is not operated at night.	
50-2	Floor Proximity Emergency Escape Path Marking System ***	С	-	-	Individual lights may be inoperative provided compliance is shown with minimum acceptable lighting levels specified in certification documents.	

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33 – <u>LIGHTS</u>			4. Remarks or Exceptions
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34 – <u>N</u>	<u>AVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions	
10-1	Metric Altimeter (600, 601, 601-3A 601-3A, 604) ***	D	1	0		
12-1	Mach Indications (600, 601)	В	2	1	(M) Pilot's Mach Indicator may be inoperative provided the remaining functions, IAS, V _{mo} and overspeed (aural) warning, Mach Trim, and TAS (read-out on the TAS/SAT/TAT indicator) are operative.	
		В	2	1	(M) Copilot's Mach Indicator may be inoperative provided both Mach trim and aural warnings are operative.	
		В	2	0	Both Mach Indicators may be inoperative provided maximum altitude is limited to FL 320 and maximum airspeed is limited to 250 KIAS.	
12-2	Copilot's Altimeter Vibrator (600, 601)	В	1	0	May be inoperative for day VMC provided the ATC transponder # 1 is used to transmit altitude data to Air Traffic Control Stations.	
12-3	V _{mo} Indicator (Barber Pole) (600, 601)	A	2	1	 (M) One may be inoperative provided: a) Both Mach airspeed indicators are operative, b) Inoperative V_{mo} indicator is on non-flying pilot side, c) Airspeed aural warning (clacker) is operative, and d) Repairs are made within ten flight hours. 	

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No de s	système/série article			3.	Number Required For Dispatch
34 – <u>N</u>	AVIGATION				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
12-4	Mach Airspeed Indicators (MASIs) (601-3A, 601-3R)	В	2	1	(M) Copilot's (right) MASI may be inoperative provided the Standby Airspeed Indicator is operative.
		В	2	1	(M) Either MASI may be inoperative provided:a) Cross-side MASI is operative, andb) Air Data Computer is operative.
		В	2	0	Both MASIs may be inoperative provided: a) Maximum altitude is limited to FL 320, and b) Maximum airspeed is limited to 250 KIAS.
12-5	Standby Airspeed Indicator (600, 601 ***) (601-3A, 601-3R, 604)	В	1	0	May be inoperative for day VMC.
12-6	Standby Altimeter (601-3A, 601-3R)	В	1	0	May be inoperative provided: a) Radio Altimeter is operative, and b) Flight is conducted in day VMC.

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System	9 Saguenas NO Itam	1.	2.		er Installed re d'articles installés
-	& Sequence Nº Item ystème/série article			3.	Number Required For Dispatch
11 400	yotomoroomo artiolo				Nombre d'articles à expédier
34 – <u>NA'</u>	<u>VIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions
12-7	Integrated Standby Instrument System (ISIS) (600, 601, 601-3A, 601-3R, 604) *** (605, 650)				
	1) Nav Function	С	1	0	
	2) Attitude Function	В	1	0	(O) May be inoperative provided: a) Operations are conducted in day VMC, and b) Instrument attitude display is covered.
					NOTE: Does not include VFR OTT.
		С	1	0	(O) May be inoperative provided: a) It is not required by regulations, and b) Instrument attitude display is covered.
	3) STD (Standard) Button	C	1	0	May be inoperative.

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Systen	n & Sequence Nº Item	1.	2.		per Installed pre d'articles installés
-	système/série article			3.	Number Required For Dispatch
11 40	Cyclomorodine unition				Nombre d'articles à expédier
34 – <u>N</u>	<u>AVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions
13-1	True Airspeed (TAS), Static Air Temperature (SAT), Total Air Temperature (TAT), Indicating Systems (600, 601, 601-3A, 601-3R)				
	1) TAS Indication	С	1	0	May be inoperative provided other required systems (FMS, OMEGA, etc) do not depend on TAS data.
	2) SAT Indication	С	1	0	May be inoperative provided:
					a) TAT is operative,
					 b) Other required systems (FMS, OMEGA, etc) do not depend on SAT data, and
					c) TAT/SAT conversion method is provided.
	3) TAT Indication	С	1	0	May be inoperative provided SAT is operative.
13-2	Vertical Speed Indicators (VSIs) (600, 601, 601-3A, 601-3R)	В	2	1	One may be inoperative provided: a) Aircraft is operated day VMC, and b) The operative VSI must be on the pilot flying side.
17-1	Altitude Alerting System				
	1) (600, 601, 601-3A,	Α	-	0	(O) Except where enroute operations require its use, may be inoperative provided:
	601-3R,)				a) Autopilot with altitude hold and altitude capture operates normally, and
					b) Repairs are made within three flight days.
	2) (604, 605, 650)	Α	2	0	(O) Except where enroute operations require its use, may be inoperative provided:
					 a) Autopilot with altitude hold and altitude capture operates normally,
					 b) Each crewmember side has the altitude tape operative on the displays, and
					c) Repairs are made within three flight days.

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Nº de s	systeme/serie article			0.	Nombre d'articles à expédier	
34 – <u>N</u>	AVIGATION				4. Remarks or Exceptions Remarques ou exceptions	
17-1	Altitude Alerting System (Cont'd) 2) (604, 605, 650) (Cont'd)					
	, ,	D	2	1	One may be inoperative.	
	3) Aural Alert (604, 605, 650)	С	1	0	May be inoperative provided: a) Visual alert operates normally, and b) Auto-pilot with altitude hold and altitude capture operates normally,	
	4) Visual Alert (604, 605, 650)	С	1	0	May be inoperative provided: a) Aural alert operates normally, and b) Auto-pilot with altitude hold and altitude capture operates normally,	
	5) Visual Alert (600, 601, 601-3A, 601-3R)	С	1	0	May be inoperative provided auto-pilot with altitude hold and altitude capture operates normally:	
20-1	Attitude Reference System Sensors (excludes Standby) (600, 601 ***)	С	-	2	Any in excess of two may be inoperative provided: a) Independent attitude indicating system is operative and available at each pilot's station, and b) Affected switches are not operated in flight.	
21-1	Instrument Comparator Monitor (600, 601)	С	1	0	May be inoperative provided approach minimums are not dependent on its use.	

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-	N & Sequence No Item				re d'articles installés
No de s	système/série article			3.	Number Required For Dispatch
34 – <u>NA</u>	AVIGATION				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
21-2	Non-EFIS ADI Annunciators / Displays (600, 601)				
	Glideslope Pointers	В	2	0	One or both may be inoperative provided Glideslope Pointers are operative on the adjacent HSI.
		В	2	0	One or both may be inoperative provided procedures are not dependent on their use.
	2) Localizer Pointers	В	2	0	One or both may be inoperative provided Localizer Pointers are operative on the adjacent HSI.
		В	2	0	One or both may be inoperative provided procedures are not dependent on their use.
	Pitch and Roll Command Bars	В	2	0	One or both may be inoperative provided approach minimums are not dependent on their use.
	Radio Altimeter Display	С	-	-	May be inoperative provided weather minimums or operating procedures are not dependent on its use.
					NOTE: Category II operation prohibited in this configuration.
	5) Rate of Turn Indicators (600, 601)	С	2	0	Both may be inoperative provided: a) Associated Turn and Slip-Skid indicators are operative and b) Standby Attitude Indicator is operative.

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System	n & Se	equence Nº Item				re d'articles installés
Nº de	systè	me/série article			3.	Number Required For Dispatch
34 – <u>NAVIGATION</u>				Nombre d'articles à expédier		
					4. Remarks or Exceptions	
						Remarques ou exceptions
21-3		n-EFIS HSI nunciators /				
		plays (600, 601)				
	1)	Bearing Pointers	С	2	0	One or both may be inoperative provided Equivalent display is available on the RMI.
			С	2	0	One or both may be inoperative provided procedures are not dependent on their use.
	2)	Distance Measuring Equipment (DME) Displays	С	2	0	As required by regulations.
	3)	Glideslope Pointers	В	2	0	One or both may be inoperative provided Glideslope Pointers are operative on the adjacent ADI.
			В	2	0	One or both may be inoperative provided procedures are not dependent on their use.
	4)	Heading	В	2	1	One may be inoperative provided:
	.,	Displays	_	_		a) Standby Compass is operative, and
						b) Adjacent RMI is operative.
	5)	Localizer Pointers	В	2	0	One or both may be inoperative provided Localizer Pointers are operative on the adjacent ADI.
			В	2	0	One or both may be inoperative provided procedures are not dependent on their use.

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		1.	2.		er Installed	٦
	em & Sequence Nº Item				re d'articles installés	
No de	e système/série article			3.	Number Required For Dispatch	
04	JAN JOATION				Nombre d'articles à expédier	
34 – <u>I</u>	<u>NAVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions	
22-1	Electronic Flight Instrument System (EFIS) (600, 601 ***) (601-3A, 601-3R)					
	1) Symbol	С	3	2	(O) One may be inoperative provided:	
	Generators				a) EFIS displays at pilot's and copilot's stations are independent, and	
					 b) Procedures are established to provide for loss of the associated Multi-Function Display (MFD). 	
	2) Multi-Function	С	1	0	May be inoperative provided:	
	[^] Display				a) Procedures are not dependent on its use, and	
					b) Weather Radar information is available at each pilot's station or the Weather Radar System must be considered inoperative.	
22-2	Turn and Slip Indicators (600, 601, 601-3A, 601-3R)	В	2	1	One may be inoperative on the pilot not-flying side.	I
	001-51()	С	2	0	One or both may be inoperative for day VFR.	١
					NOTE: Does not include VFR OTT.	
		С	2	0	One or both may be inoperative provided the associated Rate of Turn Indications are operative.	

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No de	e système/série article			3.	Number Required For Dispatch Nombre d'articles à expédier
34 – <u>1</u>	<u>NAVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions
23-1	Vertical Navigation System (VNAV) (600,601)	С	1	0	Except for RVSM Operations may be inoperative.
23-2	Inertial Reference Systems (IRSs)				
	1) 600, 601 ***	С	-	0	Except when enroute or terminal operations require its use, may be inoperative provided:
					a) Independent attitude indication is selected at each pilot's station, and
					 b) Independent directional compass indication is selected at each pilot's station.
	2) <i>(601-3A,</i>	С	-	2	Any in excess of two may be inoperative provided:
	601-3R, 604)				 a) Independent attitude indication is available at each pilot's station, and
					 b) Independent directional compass indication is available at each pilot's station.
	3) (605, 650)	С	-	2	Any in excess of two may be inoperative provided:
					 a) Independent attitude indication is available at each pilot's station,
					 b) Independent directional compass indication is available at each pilot's station, and
					 c) Autothrottle system is considered inoperative when the left Inertial Reference System is inoperative.

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1	système/série article			3.	Number Required For Dispatch	
	-				Nombre d'articles à expédier	
34 – <u>NA</u>	<u>AVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions	
23-3	Inertial Display System (601-3A, 601-3R)					
	Inertial System Display Unit (ISDU) ***	D	-	0	(O) May be inoperative provided alternate procedure is established and used.	
	2) Lasertrak - Navigation Display Unit (NDU) ***	D	-	0	(O) May be inoperative provided alternate procedure is established and used.	
23-4	Emergency Battery Packs for Backup COMM / NAV	D	-	0		
23-5	IRS Back-up Batteries					
	1) 600, 601 ***	D	-	0		
	2) 601-3A, 601-3R	С	-	1	Any in excess of one may be inoperative provided aircraft is not operated more than 60 minutes from a suitable airport.	
25-1	Directional Compass System Sensors (excludes Standby) (600, 601 aircraft 3001 to 3066)	С	-	2	Any in excess of two may be inoperative provided: a) Independent directional compass indicating system is operative and available at each pilot's station, and b) Affected switches are not operated in flight.	

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-	système/série article			3.	Number Required For Dispatch
	.,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,				Nombre d'articles à expédier
34 – <u>NA</u>	VIGATION				4. Remarks or Exceptions Remarques ou exceptions
27-1	Magnetic Compass (Standby)	В	1	0	May be inoperative provided any combination of three gyro or INS (IRU) stabilized compass system are operative.
		В	1	0	May be inoperative provided
					 a) Any combination of two Gyro or INS (IRU) stabilized compass system are operative, and
					b) Aircraft is operated:
					 With dual independent navigation capability, and
					 Under positive radar control by ATC during the enroute flight phase, or one of the navigation systems is a TSO'd GPS which provided track information.
		С	1	0	May be inoperative for flights that are entirely within areas of magnetic unreliability provided at least two stabilized directional gyro systems are installed, operative and used in conjunction with approved free gyro navigation techniques.

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-	Nº de système/série article			3.	Number Required For Dispatch			
					Nombre d'articles à expédier			
34 – <u>N</u> A	<u>AVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions			
28-1	Standby Attitude Indicator							
	1) 600, 601	В	1	0	(O) May be inoperative provided:			
					a) Flight is conducted in day VMC, and			
					b) Attitude Indicator is covered.			
					NOTE: Does not include VFR OTT.	١		
		С	1	0	(O) May be inoperative provided:	ı		
					a) It is not required by regulations, and			
					b) Attitude Indicator is covered.			
	2) 601-3A, 601-	В	1	0	(O) May be inoperative provided:	Ιİ		
	3R, 604				a) Flight is conducted in day VMC,			
					b) Independent attitude indication is available at			
					each pilot's station, and c) Attitude Indicator is covered.			
					,			
					NOTE: Does not include VFR OTT.	- 1		
		С	1	0	(O) May be inoperative provided:			
					a) It is not required by regulations, and	il		
					b) Attitude Indicator is covered.	ij		
32-1	Head-up Guidance	D	1	0	(O) May be inoperative provided procedures are not	Ш		
	Systems ***				dependent on its use.			
32-2	Enhanced Vision	D	_	0	(M) May be inoperative provided EVS is deactivated.			
32-2	System (EVS)	ט	_					
	STC SA10-63 ***				NOTE: Where the EVS image is adversely affected, the system is to be considered inoperative.			
	() =\(\(\frac{1}{2}\)\(\frac{1}2\)\(\frac{1}{2}\)\(\frac{1}2\)\(\frac{1}2\)\(\frac{1}2\)\(_						
	1) EVS Yoke Switch	D	-	0	May be inoperative provided EVS is considered inoperative.			

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		1	•.	Nombre d'articles à expédier	
34 – <u>N</u>	<u>AVIGATION</u>				Remarks or Exceptions Remarques ou exceptions
34-1	Synthetic Vision System (SVS) (including Database) (605***, 650)	D	1	0	(O) May be inoperative provided SVS is deactivated. NOTE: SVS is considered inoperative if associated database is out of date.
40-1	Windshear Alert	В	-	0	(O) May be inoperative provided:
	System ***				a) Alternate procedures are established and used, and
					 b) Takeoffs and landings are not conducted in known or forecast windshear conditions.
		С	-	0	(O) May be inoperative provided:
					a) Alternate procedures are established and used, and
					b) Windshear Warning and Guidance System (Reactive) operates normally.
40-2	Stormscope System ***	D	-	0	
40-3	Lightning Detection System (LDS) ***	D	-	0	
41-1	Weather Radar System	С	1	0	May be inoperative provided no hazardous weather conditions or thunderstorms are forecast in the area of the planned route.
41-2	Weather Radar Control Panel (600,601,601-3A 601-3R, 604)	D	-	1	
		С	1	0	May be inoperative provided no hazardous weather conditions or thunderstorms are forecast in the area of the planned route.

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System & Sequence N Nº de système/série 34 – NAVIGATION 42-1 Radio Altimer Systems	article	 	3.	Number Required For Dispatch Nombre d'articles à expédier 4. Remarks or Exceptions
42-1 Radio Altime	ter A	_		
1	ter A	-		Remarques ou exceptions
			0	(O) May be inoperative for three flight days provided: a) Radio altitude data is either available to, or not required by other required systems, and b) Approach minimums or operating procedures are not dependent on its use.
				NOTE 1 : Radio altitude data is required for Autopilot, Flight Director System, TCAS, GPWS (TAWS) and Autothrottle System (ATS).
				NOTE 2 : (605, 650)(604 with ATS) ATS retard mode not available when pilot's radio altimeter is inoperative.
				NOTE 3 : EFIS COMP INOP (Caution) message might be displayed.
42-2 Terrain Aware and Warning System (TAW				
Class A TAW Equipment Re				
1) GPWS	A	1	0	(O) Except where enroute operations or approach procedures require its use, may be inoperative provided:
				a) Alternate procedures are established and used, and
				b) Repairs are made within three flight days.
				NOTE: For RNP AR approaches, refer to AFM chapter 07, supplement 18, RNP Authorization Required Approach Operations for required equipment.
a) Mode	es 1-4 A	4	0	(O) May be inoperative provided:
				 a) Alternate procedures are established and used, and
				b) Repairs are made within three flight days.
b) Test	Mode A	1	0	May be inoperative provided: a) GPWS is considered inoperative, and b) Repairs are made within three flight days.
(Cont'd)				

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	_	s/série article			3.	Number Required For Dispatch	
11 45	<u> </u>	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,				Nombre d'articles à expédier	
34 – <u>N</u>	AVIGATI	<u>ON</u>				4. Remarks or Exceptions	
						Remarques ou exceptions	
42-2							
		A TAWS nent Required)					
	1) GP (Cd	PWS ont'd)					
	c)	Glideslope Deviation(s) (Mode 5)	С	-	1		
			В	-	0		
	d)	Advisory Callouts (Mode 6) ***	C	-	0	(O) May be inoperative provided alternate procedures are established and used.	
	(Cont'o	d)					

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OA NAVIOATION					Nombre d'articles à expédier	
34 – <u>N</u>	<u>AVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions	
42-2	Terrain Awareness and Warning System (TAWS) (Cont'd)					
	Class A TAWS Equipment Required (Cont'd)					
	1) GPWS (Cont'd)					
	e) Windshear Reactive Mode (Mode 7)	В	1	0	 (O) May be inoperative provided: a) Alternate procedures are established and used, and b) Takeoffs and landings are not conducted in known or forecast windshear conditions. 	
		С	1	0	(O) May be inoperative provided: a) Alternate procedures are established and used, and b) Windshear Detection and Avoidance System (Predictive) is operative.	
	2) Terrain System – Forward Looking Terrain Avoidance (FLTA) and Premature Descent Alert (PDA) Functions ***	В	1	0	May be inoperative.	I
	3) Terrain Displays	С	-	1		
	(Cont'd)	В	-	0		

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11- 46 3	ystemersene article		1		Nombre d'articles à expédier
34 – <u>NA</u>	<u>VIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions
42-2	Terrain Awareness and Warning System (TAWS) (Cont'd)				
	Class A TAWS Equipment Required (Cont'd)				
	4) Runway Awareness and Advisory System (RAAS) (Includes Smart Runway / Smart Landing (SR/SL) functions (605 aircraft with SB 605-34-022) *** or (650 aircraft with SB 650-34-010) ***	С	1	0	(O) May be inoperative provided the RAAS system is selected OFF.
	Equipment Required				
	1) GPWS	Α	1	0	(O) May be inoperative provided: a) Alternate procedures are established and used, and
					b) Repairs are made within three flight days.
	a) Modes 1 & 3	Α	2	0	 (O) May be inoperative provided: a) Alternate procedures are established and used, and b) Repairs are made within three flight days.
	b) Test Mode	Α	1	0	May be inoperative provided: a) GPWS/TAWS is considered inoperative, and b) Repairs are made within three flight days.
	c) Modes 2, 4 & 5 ***	С	3	0	
	(Cont'd)				

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04 NAVIOATION				Nombre d'articles à expédier	
34 – <u>NA</u>	AVIGATION				4. Remarks or Exceptions Remarques ou exceptions
42-2	Terrain Awareness and Warning System (TAWS) (Cont'd)				
	If Class B TAWS Equipment Required (Cont'd)				
	1) GPWS (Cont'd)				
	d) Advisory Callouts	С	-	0	(O) May be inoperative provided alternate procedures are established and used.
	e) Windshear Reactive Mode (Mode 7)	С	1	0	(O) May be inoperative provided: a) Alternate procedures are established and used, and b) Takeoffs and landings are not conducted in known or forecast windshear conditions.
	2) Terrain System – Forward Looking Terrain Avoidance (FLTA) and Premature Descent Alert (PDA) Functions	В	1	0	
	3) Terrain Displays	С	-	0	
	(Cont'd)				

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-	* Sequence No Item			Nomb	re d'articles installés Number Required For Dispatch
Nº de s	système/série article			3.	Nombre d'articles à expédier
34 – <u>NA</u>	AVIGATION				4. Remarks or Exceptions Remarques ou exceptions
42-2	Terrain Awareness and Warning System (TAWS) (Cont'd)				
	If Class B TAWS Equipment Required (Cont'd)				
	4) Runway Awareness & Advisory System (RAAS) (Includes Smart Runway / Smart Landing (SR/SL) functions (605 aircraft with SB 605-34- 022) *** or (650 aircraft with SB 650-34-010) ***	C	1	0	(O) May be inoperative provided the RAAS system is selected OFF.
	If Class C TAWS Equipment required				
	1) GPWS	С	1	0	(O) May be inoperative provided alternate procedures are established and used.
					NOTE: Any mode that is operative may be used.

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System & Sequence No Item No de système/série article				3.	Number Required For Dispatch		
11- 40	JyJio	merserie urticie				Nombre d'articles à expédier	
34 – <u>N</u>	AVIG/	<u>ATION</u>				4. Remarks or Exceptions Remarques ou exceptions	
43-1	Ave	affic Collision oidance System CAS)					
		AS II	В	-	0	(O) May be inoperative provided:	
		0, 601, 601-3A,				a) System is deactivated and secured, and	
	(60 (Ind fun Sur (TS	cluding TCAS ction of the Traffic rveillance System				b) Enroute or approach procedures do not require its use.	
	(00	,	С	_	0	(O) May be inoperative provided:	ı
						a) Not required by regulations,	i
						b) System is deactivated and secured, and	-
						c) Enroute or approach procedures do not require its use.	1
	1)	Combined Traffic Alert (TA) and Resolution Advisory (RA) Dual Displays	С	2	1	(O) One may be inoperative on the non-flying pilot side provided TA and RA visual display and audio functions are operative on flying pilot side.	I
	2)	RA Display System(s)	С	2	1	May be inoperative on the non-flying pilot side.	
			С	_	0	(O) May be inoperative provided:	
						a) Traffic Alert (TA) visual display and audio functions are operative,	I
						b) TA only mode is selected by the crew, and	
						c) Enroute or approach procedures do not require its use.	
	3)	TA Display	С	_	0	(O) May be inoperative provided:	
	,	System(s)				a) RA visual display and audio functions are operative, and	
						b) Enroute or approach procedures do not require its use.	1
	(Co	ont'd)					

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					Nombre d'articles à expédier
34 – <u>N</u>	<u>AVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions
43-1	Traffic Collision Avoidance System (TCAS) (Cont'd)				
	TCAS I (600, 601, 601-3A, 601-3R) ***	В	-	0	(O) May be inoperative provided: a) System is deactivated and secured, and b) Enroute or approach procedures do not require its use.
		С	-	0	(O) May be inoperative provided: a) Not required by regulations, b) System is deactivated and secured, and c) Enroute or approach procedures do not require its use.
	TCAS Display System(s)	С	-	0	May be inoperative provided all installed audio functions are operative.
50-1	Long Range Navigation Systems (OMEGA, LORAN, VLF, GPS, etc) (600, 601, 601-3A, 601-3R, 604)	D	-	-	(O) May be inoperative provided alternate procedures are established and used.

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Nº de	e système/série article			3.	Number Required For Dispatch
					Nombre d'articles à expédier
34 – <u>1</u>	<u>NAVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions
50-2	Canadian Marconi ONA-T16 Status Display System (SDS) ***	D	-	0	
51-1	VHF Navigation Systems (VOR / ILS)	С	-	-	Any in excess of those required by regulations and not powered by an emergency or standby electrical bus may be inoperative.
51-2	Marker Beacon Systems	D	-	-	May be inoperative provided approach minimums are not dependent on its use.
51-3	NAV / COM / ADF / ATC System (600, 601 ***) (601-3A, 601-3R)				
	1) Preselect Tuning	С	-	0	May be inoperative provided direct tuning mode is operative.
	2) Memory Channels	С	-	0	May be inoperative provided direct tuning mode is operative.
51-4	Radio Magnetic Indicators (RMIs)				
	1) (600, 601)	С	2	0	One or both may be inoperative provided both HSIs are operative.
	2) (601-3A, 601- 3R)***	С	-	0	May be inoperative provided both EHSIs are operative.

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Nº de système/série article				3.	Number Required For Dispatch		
34 – <u>NAVIGATION</u>					Nombre d'articles à expédier		
34 – <u>r</u>	<u>NAVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions		
52-1	Automatic Direction Finding (ADF) Systems	D	-	-	Any in excess of those required by regulations may be inoperative.		
53-1	Distance Measuring	С	-	-	May be inoperative provided:		
	Equipment (DME) Systems				 a) Procedures are not dependent on its use, and b) FMS accuracy without DME update is acceptable for the intended operation (600, 601 ***) (601-3A, 601-3R, 604, 605, 650). 		
		D	-	-	Any in excess of those required by regulations may be inoperative provided: a) Procedures are not dependent on its use, and b) FMS accuracy without DME update is acceptable for the intended operation (600, 601 ***) (601-3A, 601-3R, 604, 605, 650).		
53-2	DME Repeater Indicators ***	D	-	0			

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Nº de	e système/série article			3.	Number Required For Dispatch	
24 1	JAN/ICATION				Nombre d'articles à expédier 4. Remarks or Exceptions	
34 – <u>r</u>	<u>NAVIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions	
54-1	Air Traffic Control	В	2	0	May be inoperative provided:	ļ
	(ATC) Transponders (Includes transponder				a) Operations do not require its use, and	!
	function of Traffic Surveillance System (TSS)				b) Prior to flight, approval is obtained from ATC facilities having jurisdiction over the planned route of flight.	
	(650))	D	2	1	Any in excess of those required by regulations may be inoperative.	
	1) Elementary and	Α	-	0	May be inoperative provided:	ı
	Enhanced Downlink				a) Operations do not require its use, and	•
	Aircraft Parameters not required by regulations (604 aircraft with SB 604-34-040 or SB 604-34-041) (605, 650)				b) Repairs are made prior to completion of the next heavy maintenance visit.	
	2) ADS-B Squitter	Α	-	0	May be inoperative provided:	
	Transmissions (604 aircraft with SB 604-34-058 and 605 aircraft with SB 605-34-013, or SB 605-34-030) (650)				a) Operations do not require its use, and b) Repairs are made prior to completion of the next heavy maintenance visit.	
54-2	Remote Transponder Ident Buttons ***	D	-	0	 (M) All may be inoperative provided: a) IDENT button had not faulted to the ON condition, and b) IDENT button on Transponder Control Unit / RTU is operative. 	
60-1	Datanav System ***	D	-	0		

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-	System & Sequence No Item		-				re d'articles installés
Nº de système/série article 34 – NAVIGATION				3.	Number Required For Dispatch		
					Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions		
60-5	Global Positioning System (GPS) (605, 650)	С	2	0	(O) Except where enroute operations require its use, may be inoperative provided alternate procedures are established and used		
		D	2	1	Except where enroute operations require dual GPS.		
61-1	Flight Management Systems (FMS) 1)(600, 601)	С	-	0	(O) Except when operations require its use, may be inoperative provided alternate procedures are established and used.		
		D	-	-	May be inoperative provided procedures do not require its use.		
	2) (601-3A, 601-3R)	С	-	1	Except when enroute operations require dual FMS use, any in excess of one may be inoperative.		
		D	-	-	May be inoperative provided procedures do not require its use.		
	3) (604 without SB 604-34-031)	С	-	1	Except when enroute operations require dual FMS use, any in excess of one may be inoperative.		
					NOTE: Autothrottle system is considered inoperative when the FMS No. 1 is inoperative (604 with autothrottle system).		
		D	-	-	May be inoperative provided procedures do not require its use.		
	4) (604 with SB 604-34-031)	С	-	1	(O) Except when enroute operations require dual FMS use, any in excess of one may be inoperative.		
					NOTE 1: Autothrottle system is considered inoperative when the FMS No. 1 is inoperative (604 with autothrottle system).		
					NOTE 2: When operating with a single FMS operative, FMS-computed take off and approach performance (FMS V speed) and FMS-computed N1 thrust data must be verified using applicable AFM data.		
		D	-	-	May be inoperative provided procedures do not require its use.		
	(Cont'd)		I	I			

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No de système/série article				3.	Number Required For Dispatch	
it ac systemersence article					Nombre d'articles à expédier	
34 – <u>NA</u>	<u>VIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions	
61-1	Flight Management Systems (FMS) (Cont'd)					
	5) (605, 650)					
	a) Flight Management Computer (No. 1 and No. 2)	С	2	1	 (O) Except when enroute operations require dual FMS use, any in excess of one may be inoperative: a) One GPS system is operative, and b) Autothrottle system is considered inoperative when the FMS No. 1 is inoperative. NOTE: When operating with a single FMS operative, FMS-computed take off and approach 	
					performance (FMS V speed) and FMS- computed N1 thrust data must be verified using applicable AFM data.	
	b) Control Display Units (No. 1 and No. 2)	С	2	1	 (O) CDU 2 may be inoperative provided: a) All Adaptive Flight Displays (AFD) are operative, b) Associated Flight Management Computer is considered inoperative, c) CDU 2 INHIB switch/light is pressed in, and d) Both Cursor Control Panels (CCP) are operative. 	
					NOTE: When operating with a single FMS operative, FMS-computed take off and approach performance (FMS V speed) must be verified using applicable AFM data.	
	c) CDU Radio Tuning Function (No. 1 and No. 2)	С	2	1	 (O) CDU Radio Tuning No. 2 may be inoperative provided: a) All Adaptive Flight Displays (AFD) are operative, b) Radio tuning reversion capabilities are verified operative before each flight., and c) CDU 2 INHIB switch/light is pressed in d) Both Cursor Control Panels (CCP) are operative. 	
	(Cont'd)				d) Both Cursor Control Panels (CCP) are	

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					Nombre d'articles à expédier
34 – <u>NA</u>	<u>VIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions
61-1	Flight Management Systems (FMS) (Cont'd)				
	5) (605, 650)				
	d) Flight Management Computer (FMC) No. 3 ***	D	1	0	
	e) Control Display Unit (CDU) No. 3 ***	D	1	0	
	Cont'd)				

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Nº de système/série article				3.	Number Required For Dispatch
					Nombre d'articles à expédier
34 – <u>NA</u>	<u>VIGATION</u>				4. Remarks or Exceptions Remarques ou exceptions
61-1	Flight Management Systems (FMS) (Cont'd)				
	6) Navigation Databases	С	-	0	(O) One or more may be inoperative for the intended flight route where conventional (non-RNAV/RNP) navigation is sufficient, provided:
					a) Current aeronautical information (e.g. charts) is available for the entire route and for the aerodromes to be used,
					b) Navigation database information is disregarded, and
					c) Radio navigation aids, which are required to be flown for departure, arrival and approach procedures are manually tuned and identified.
		С	-	1	(O) Any in excess of one may be inoperative provided: a) The operative database must be up to date for routes, departures, arrival and approach procedures that require the use of navigation Database for RNAV/RNP,
					b) The operative database is available and used by the flight crew member(s) responsible for navigation, and
					c) Radio navigation aids, which are required to be flown for departure, arrival and approach procedures are manually tuned and identified.
	(Cont'd)		l		

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Nº de s	système/série article			3.		r Required For Dispatch
24 NI/	AVIGATION				Nombr	e d'articles à expédier Remarks or Exceptions
54 – <u>IN</u>	AVIGATION				4.	Remarques ou exceptions
61-1	Flight Management Systems (FMS) (Cont'd)	A		0	(0) 0	
	6) Navigation Databases	^	-			ne or more may be out of date for a maximum of endar days provided:
						Area Navigation (RNAV/RNP) departure, arrival and approach procedures are checked not to depend on the data amended in the current database cycle or Conventional (Non-RNAV/RNP) or ANSP assistance are used as an alternative to RNAV/RNP procedures which have been amended in the current database cycle, Before each flight, current aeronautical information is used to verify the database Navigation Fixes, the coordinates, frequencies, status (as applicable) and suitability of Navigation Facilities required for the intended flight route, and
					c)	Radio navigation aids, which are required to be flown for departure, arrival and approach procedures and which have been amended in the current database cycle, are manually tuned and identified.

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Nº UE	systemerserie article				Nombre d'articles à expédier
34 – N	<u>AVIGATION</u>				4. Remarks or Exceptions
- <u> </u>					Remarques ou exceptions
61-2	Flight Management System (FMS) Performance Data Base (604, 605, 650)	D	-	0	(O) May be inoperative (not installed) provided alternate procedures are established and used.
61-3	Data Base Unit (DBU) (604, 605, 650)	С	1	0	
61-4	Flight Management System (FMS) Vspeed Data Base (604, 605, 650)	D	-	0	(O) May be inoperative (not installed) provided alternate procedures are established and used.
61-5	Flight Management System (FMS) N1 Thrust Data Base (604***, 605, 650)	D	-	0	(O) May be inoperative (not installed) provided alternate procedures are established and used.

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_	de système/série article		-			3.	Number Required For Dispatch
35 – <u>OXYGEN</u>					Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions		
10-2	Crew Oxygen Pressure Gauge (600, 601, 601-3A, 601-3R)	В	1	0	(M) May be inoperative provided both the fill point gauge and the bottles mounted gauge are checked prior to each flight.		
12-1	Crew Oxygen Pressure (604, 605, 650)						
	1) EICAS Readout	В	1	0	(M) May be inoperative provided the Ground Service Panel pressure gauge or the bottle pressure gauge is operative and checked prior to each flight.		
	2) Ground Service Panel Pressure Gauge	С	1	0	May be inoperative provided the EICAS Readout or the bottle pressure gauge is operative and checked prior to each flight.		
	3) Bottle Pressure Gauge (604, 605, 650)	С	-	0			
20-1	Passenger Oxygen System	В	1	0	 (O) May be inoperative provided: a) Minimum enroute altitude does not exceed 13,000 feet above MSL, b) Both air conditioning units are operative, c) Pressurization system is operative, d) Aircraft is operated at FL 250 or below, e) Portable oxygen units are provided for all crewmembers and 10% of the passengers for half an hour (supplemental oxygen), and f) Passengers are appropriately briefed. 		

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	XYGEN				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions	
20-2	Passenger Oxygen System – Automatic Presentation System	В	1	0	May be inoperative provided: a) Manual deployment system is operative, and b) Aircraft is operated at FL 300 or below.	
20-3	Passenger Service Units (PSUs)					
	1) Individual PSUs	D	-	0	(M)(O) May be inoperative with no flight altitude restriction provided:	
					 a) Affected seats are blocked and placarded to prevent occupancy, 	
					 b) No more than two consecutive banks of seats and their adjacent banks of seats have inoperative PSU, and 	
					 c) Units operate normally for all usable lavatory and flight attendant locations. 	

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_	système/série article			3.	Number Required For Dispatch	
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35 – <u>C</u>	<u>DXYGEN</u>				4. Remarks or Exceptions Remarques ou exceptions	
20-4	Lavatory Oxygen System	С	1	0	May be inoperative provided: a) The lavatory is not used for any purpose, and b) The lavatory door is locked and placarded "INOPERATIVE DO NOT ENTER". NOTE: This does not preclude storage of in-flight service waste bags in the associated lavatory.	
		С	1	0	May be inoperative provided the aircraft is not operated above FL250.	
25-1	Protective Breathing Equipment	D	-	-	 (O) Any in excess of those required by regulation may be inoperative or missing provided: a) Required distribution of operative units is maintained throughout the aircraft, b) The inoperative protective breathing equipment unit is removed from the passenger cabin and its location is placarded INIOPERATIVE, or it is removed from the installed location, secured out of sight and the protective breathing equipment unit and its installed location are placarded INOPERATIVE, and c) Procedures are established and used to alert crew members of inoperative or missing equipment. 	
30-1	Portable Oxygen Dispensing Units – Bottles with Demand Flow Masks (including medical portable oxygen)	D	-	-	 (O) Any in excess of those required by regulations may be inoperative or missing provided: a) Required distribution of operative units is maintained throughout the aircraft, b) The inoperative portable oxygen dispensing unit is removed from the aircraft or removed from the installed location, placarded "INOPERATIVE" and secured out of sight so it can not be mistaken for an operative unit, and c) Procedures are established and used to alert crewmember of inoperative or missing equipment. 	

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35 – <u>OXYGEN</u>			4. Remarks or Exceptions Remarques ou exceptions
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System					Number Installed Nombre d'articles installés			
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10-1	Engine Bleed Air SOVs (14 th Stage) (601, 601-3A, 601-3R, 604, 605, 650)	С	2	1	 (M) One may be inoperative provided: a) Affected valve is secured CLOSED, b) Both Ice Detection Systems are operative, c) Associated Thrust Reverser is deactivated, stowed and locked in forward thrust position, d) Flight is not conducted in known or forecast icing conditions, e) Aircraft is operated at or below FL 400, and f) 14th Stage Engine Bleed Air Isolation Valve is operative. NOTE: Refer to MMEL Item 78-30-1. 			
20-1	Bleed Air Pressure Indicators (LH & RH)							
	1) 8 th Stage (600)	С	2	1	(O) One may be inoperative provided procedures are used to ensure that the pressure is present at both sides.			
	2) 10 th Stage (601, 601-3A, 601-3R)	С	2	1	(O) One may be inoperative provided procedures are used to ensure that the pressure is present at both sides.			
20-2	EICAS Bleed Air Pressure Readout (604, 605, 650)	С	2	1	(O) One may be inoperative provided procedures are used to ensure that the pressure is present at both sides.			

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Nº GE	e sys	teme/serie article			0.	Nombre d'articles à expédier
36 – <u>F</u>	PNEL	<u>IMATICS</u>				4. Remarks or Exceptions Remarques ou exceptions
23-1		ed Air Overheat nsing Loops A and				
	1)	600, 601, 601-3A, 601-3R	С	2	1	Either loop A or B may be inoperative provided: a) Remaining loop is operative, and b) DUCT MON switch is selected to operative loop position.
	2)	604, 605, 650	С	2	1	(O) Either loop A or B may be inoperative provided neither loop is failed open.

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38 – <u>W</u>	ATER/WASTE				4. Remarks or Exceptions Remarques ou exceptions
10-1	Potable Water System	С	-	0	(M) May be inoperative provided appropriate procedures are established to deactivate applicable system components (i.e. tank drained) to prevent its servicing, inspect system for leaks, and to provide for crewmember inspection.
		С	-	-	(M) Individual components may be inoperative provided:
					Associated components are deactivated or isolated, and
					 b) Associated system components are verified not to have leaks.
					NOTE: Any portion of the system that operates normally may be used.
10-2	Lavatory Waste Systems	С	-	-	(M) Associated lavatory system may be inoperative provided:
					 a) Associated components are deactivated or isolated to prevent leaks, and
					 b) Associated lavatory door is secured closed and placarded, "INOPERATIVE - DO NOT ENTER".
					NOTE:
					These provisions are not intended to prohibit inspections by crewmembers.
					Aviation Occupational Health & Safety (AOH&S) requirements should be addressed.
		С	-	-	(M)(O) Individual components may be inoperative provided:
					Associated components are deactivated or isolated, and
					 b) Associated system components are verified not to have leaks.
					NOTE: Any portion of the system that operates normally may be used.

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	système/série article			3.	Number Required For Dispatch
					Nombre d'articles à expédier
38 – <u>W</u>	ATER/WASTE				4. Remarks or Exceptions Remarques ou exceptions
10-3	Lavatory External Service Cap	C	1	0	

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System & Sequence No Item	1.	2.		er Installed re d'articles installés
No de système/série article 45 – CENTRAL MAINTENANCE SYSTEMS			3.	Number Required For Dispatch Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
45-1 Maintenance Diagnostic Computer (MDC) (604, 605, 650)	C	1	0	(O) May be inoperative provided: a) Electronic Checklists if installed is considered inoperative, and b) Alternate procedures are established and used.

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			Number Required For Dispatch
Nº de système/série article	1		
45 CENTRAL			Nombre d'articles à expédier 4. Remarks or Exceptions
45 – <u>CENTRAL</u> <u>MAINTENANCE SYSTEMS</u>			4. Remarks or Exceptions Remarques ou exceptions
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		1.	2.		er Installed	
_	n & Sequence No Item			Nomb	re d'articles installés Number Required For Dispatch	
Nº de s	système/série article			0.	Nombre d'articles à expédier	
46 – <u>INF</u>	FORMATION SYSTEMS				4. Remarks or Exceptions Remarques ou exceptions	
10-1	Electronic Flight Bags (EFB)	С	-	-	(O) May be inoperative provided alternate procedures are established and used.	
					NOTE: Any function, program or document, which operates normally, may be used.	ı
		D	-	0	May be inoperative provided procedures do not require its use.	
	1) Data Connectivity	С	-	-	(O) May be inoperative provided alternate procedures are established and used.	
		D	-	0	May be inoperative provided procedures do not require its use.	
	2) Power Connection	С	-	-	(M)(O) May be inoperative provided alternate procedures are established and used.	
		D	-	0	May be inoperative provided procedures do not require its use.	
	3) Mounting	С	_	_	(M)(O) May be inoperative provided:	ı
	Devices ***				a) Associated EFB and hardware is secured by an alternate means approved by the local airworthiness authority or removed from the aircraft, and	
					b) Alternate procedures are established and used.	
		D	_	0	May be inoperative provided:	
		2			a) Associated EFB and hardware is secured by an alternate means approved by the local airworthiness authority or removed from the aircraft, and	
					b) Procedures do not require its use.	

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		1.	2.		per Installed	
•	n & Sequence No Item				ore d'articles installés	
Nº de	système/série article			3.	Number Required For Dispatch	
46 INI	EODMATION SVSTEMS				Nombre d'articles à expédier 4. Remarks or Exceptions	
40 – <u>IIVI</u>	FORMATION SYSTEMS				Remarques ou exceptions	
10-2	Integrated Flight Information System (IFIS) (605, 650)					
	1) File Server Unit (FSU)	С	1	0	(O) May be inoperative provided alternate procedures are established and used.	
		D	1	0	May be inoperative provided procedures do not require its use. NOTE: Any function which operates normally may be	I
	2) Electronic Chart Displays (Including Electronic CHART Database)	С	2	0	used. (O) May be inoperative provided alternate procedures are established and used.	
		D	2	0	May be inoperative provided procedures do not require its use.	I
	3) Graphical Weather Function (Including Graphical Weather Database)	С	1	0	(O) May be inoperative provided alternate procedures are established and used.	
		D	1	0	May be inoperative provided procedures do not require its use.	I
	(Cont'd)					

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		1.	2.		er Installed
-	n & Sequence Nº Item				re d'articles installés
Nº de	système/série article			3.	Number Required For Dispatch
46 – <u>INI</u>	FORMATION SYSTEMS				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
10-2	Integrated Flight Information System (IFIS) (605, 650) *** (Cont'd)				
	4) Enhanced Map Overlay Function (Including Enhanced Map Overlay Database) ***	С	1	0	(O) May be inoperative on either or both MFDs provided alternate procedures are established and used.
		D	1	0	May be inoperative on either or both MFDs provided procedures do not require its use.
20-1	Airborne Flight Information System (AFIS) (600, 601, 601-3A, 601-3R, 604)	D	-	0	May inoperative provided procedures do not require its use.
20-2	XM Satellite System ***	С	1	0	(O) May be inoperative provided alternate procedures are established and used.
		D	1	0	May be inoperative provided procedures do not require its use.

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Nº de système/série article			3.			
		-		Nombre d'articles à expédier		
46 -INFORMATION SYSTEMS				4. Remarks or Exceptions		
				Remarques ou exceptions		
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1. System & Seguence NO Item			2.		per Installed	
System & Sequence No Item No de système/série article			Nome	ore d'articles installés Number Required For Dispatch		
Nº de s	systeme/serie article			J.	Nombre d'articles à expédier	
49 – <u>AF</u>	P[]				4. Remarks or Exceptions	
10 /11	<u> </u>				Remarques ou exceptions	
00-1	APU Bleed Air SOV (Load Control Valve)	С	1	0	May be inoperative closed.	
00-2	Auxiliary Power Unit (APU)					
	1) Aircraft with Generator P/Ns 720845, 720845A, 720845B	С	1	0	(M) May be inoperative provided any main generator with these P/Ns has more than 150 operating hours.	
	Aircraft with other Generator P/Ns	С	1	0		
00-3	APU Load Control Valve (APU LCV) Fail Switchlight (light function only) (604, 605, 650)	С	1	0		
00-4	APU Hour Meter	С	-	0	(O) May be inoperative provided alternate means of recording is established.	I
		D	-	1		I
00-5	APU / START / STOP / AVAIL (Light function only) (604, 605, 650)	С	1	0	(O) May be inoperative provided, APU parameters are used to confirm that the APU has started or stopped.	
40-1	APU Start Auxiliary Battery (600, 601 ***) (601-3A)				Deleted, moved to item 24-30-3	

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			1.	2.		er Installed
Systen	n & S	equence Nº Item			Nombre d'articles installés	
Nº de	l ^o de système/série article				3.	Number Required For Dispatch
						Nombre d'articles à expédier
49 – <u>Af</u>	<u> PU</u>					4. Remarks or Exceptions Remarques ou exceptions
61-1		ectronic Control it (ECU)				
	1)	With IDG P/Ns 720845 720845A and 720845B	С	1	0	(M) May be inoperative provided any main generator with one of these P/Ns has more than 150 operating hours.
	2)	With Other IDG P/Ns	С	1	0	

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01	O O NO Kara	1.	2.	Number Installed Nombre d'articles installés			
1	& Sequence Nº Item système/série article			3. Number Required For Dispatch			
14° de s	systemersene article				Nombre d'articles à expédier		
52 – <u>DC</u>	<u>OORS</u>				4. Remarks or Exceptions Remarques ou exceptions		
00-1	Exterior Door Key Locks						
	Passenger Door Key Lock	D	1	0	May be inoperative or missing provided door opens and closes normally.		
	Cargo/Baggage Door Key Lock	D	1	0	May be inoperative or missing provided door opens and closes normally.		
	Aft-Equipment Compartment Door Key Lock	D	1	0	May be inoperative or missing provided door opens and closes normally.		
	4. External Service Doors Key Locks ***	D	-	0	May be inoperative or missing provided door opens and closes normally.		
11-1	Passenger Door Manual Retract Handle	С	1	0	May be inoperative provided door is verified manually operative (opens and closes) without any interference.		
11-2	Storage Compartment Door (Main Entry Door)	С	1	0	(M) May be inoperative provided:a) Storage content is relocated, andb) Door is secured closed.		
12-1	Main Cabin Door Power Assist System (600, 601 ***) (601-3A, 601-3R, 604, 605, 650)	С	-	0	May be inoperative provided the door can be opened and closed manually without any interference.		

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Syston	n & Sequence Nº Item	1.	2.		er Installed re d'articles installés
-	système/série article			3.	Number Required For Dispatch
52 – <u>D(</u>	<u>DORS</u>				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
30-1	Baggage Door Counterbalance Systems (balance springs)	С	2	0	(O) Both may be inoperative provided baggage door is verified closed, latched, locked and not used. NOTE: Baggage compartment is accessible through the door located in the aft cabin.
70-1	Main Cabin Door Indication System (600, 601, 601-3A, 601-3R)	С	1	0	May be inoperative provided: a) All four latches are engaged as indicated by witness marks, b) Inner Handle is verified stowed, c) External handle is verified stowed, and d) Airstair (CL-600 if installed) is properly stowed and the actuator arming latch is disengaged. NOTE: Canadair Service Bulletin 600-0621 or 601-0395 must be incorporated.
70-2	Passenger Door Indication System (604, 605, 650)	В	1	0	 (O) Any or all of the following caution messages: "PAX DOOR LATCH" "PAX DOOR STOW" "PAX DOOR OUT HNDL" may be either inoperative or malfunction provided: a) Door is closed, latched and locked prior to each flight, b) Green marks on the two middle roll latches are aligned, c) Green marks on the two upper roll latches are aligned, d) Flag indicator indicates "LOCKED", e) Inner knob is verified stowed, f) External handle is verified stowed, and g) External pressure vent flap is verified operative.

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-	système/série article			3.	Number Required For Dispatch
	•		1		Nombre d'articles à expédier
52 – <u>DC</u>	<u>OORS</u>				4. Remarks or Exceptions Remarques ou exceptions
70-3	BAG DR UNSAFE Caution Light (600, 601, 601-3A, 601-3R)	С	1	0	May be inoperative provided the door is confirmed closed and latched prior to each flight.
70-4	Baggage Door Indication System (604, 605, 650)	С	1	0	Caution message "BAGGAGE DOOR" may be inoperative provided: a) Associated door is closed, latched and locked, and b) External handle is verified stowed prior to each flight.
70-5	Refueling Panel Door Open Caution Light ***	D	-	0	All may be inoperative provided it can be determined that the door is closed and latched prior to each flight.
70-6	Hydraulic # 3 Access Door Open Caution Light	D	1	0	May be inoperative provided access door is confirmed closed and latched prior to each flight.
70-7	Lavatory Servicing Outlet Door Open Caution Light	D	1	0	May be inoperative provided outlet door is confirmed closed and latched prior to each flight.
70-8	Aft Equipment Bay Door Open Caution Light ***	D	1	0	May be inoperative provided door is confirmed closed and latched prior to each flight.
70-9	Nose Door (Oxygen Service) Open Caution Light	D	1	0	May be inoperative provided door is confirmed closed and latched prior to each flight.
70-10	Water Fill Control Access Door Open Caution Light	D	1	0	May be inoperative provided door is confirmed closed and latched prior to each flight.

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Nº de système/série article				3.	Number Required For Dispatch
14- 46.	systemerserie article				Nombre d'articles à expédier
52 – <u>DC</u>	<u>DORS</u>				4. Remarks or Exceptions Remarques ou exceptions
70-11	DC Ground Access Door Open Caution Light	D	1	0	May be inoperative provided door is confirmed closed and latched prior to each flight.
70-12	AC Ground Access Door Open Caution Light	D	1	0	May be inoperative provided door is confirmed closed and latched prior to each flight.
70-13	APU Control and Ground Start Access Door Open Caution Light ***	D	1	0	May be inoperative provided door is confirmed closed and latched prior to each flight.

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		3.	Number Required For Dispatch
			Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
Α	2	1	(M) The face ply of one windshield may be cracked provided:
			 a) Vision is not impaired through remaining windows,
			b) Tape and face-ply is checked for integrity prior to each flight,
			c) Visibility through the affected windshield is acceptable,
			 d) Affected windshield heater is considered inoperative,
			e) Pilot's (Left) side window heater and at least one additional windshield or side window heater are operative,
			 f) Flight into known or forecast icing conditions, precipitation, thunderstorms or other conditions which could cause fogging or misting is prohibited,
			g) Airspeed is limited to 280KIAS or less below 8, 000 feet MSL, and
			h) Repairs are made within 6 flights.
			3.

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1. System & Sequence No Item	2.	Numb Nomb	er Installed re d'articles installés
Nº de système/série article		3.	Number Required For Dispatch
it do oyotomercone undelo	1		Nombre d'articles à expédier
56 – <u>WINDOWS</u>			4. Remarks or Exceptions
			Remarques ou exceptions
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System & Sequence No Item	1.	2.		er Installed re d'articles installés
Nº de système/série article			3.	Number Required For Dispatch
11 de dystemersene union				Nombre d'articles à expédier
71 – <u>POWER PLANT</u>				4. Remarks or Exceptions Remarques ou exceptions
Figetor Pumps (600, 601, 601-3A)	C	2	1	(M) One may be inoperative provided the tank is drained: a) Prior to the first flight of each day, b) After three normal shutdowns, and c) After two false wet starts.

DEPARTMENT OF TRANSPORT

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No de système/série article		Number Required For Dispatch
14° de systeme/serie article	1	Nombre d'articles à expédier
71 – <u>POWER PLANT</u>		4. Remarks or Exceptions Remarques ou exceptions
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Sveto	em & Sequence Nº Item	1.	2.		er Installed re d'articles installés
_	e système/série article			3.	Number Required For Dispatch
11 41	o oyotomoroono artiolo				Nombre d'articles à expédier
	ENGINE FUEL &				4. Remarks or Exceptions
<u>(</u>	<u>CONTROL</u>				Remarques ou exceptions
30-1	Fuel Temperature Indicating Systems	С	2	1	One may be inoperative provided icing inhibitor is added to the fuel.
31-1	Fuel Flow Indicating	В	2	1	One may be inoperative provided:
	Systems (600, 601, 601-3A, 601-3R)				 a) Fuel Quantity Indicating System is operative, and
	,				 b) N1, N2 and ITT indications for the associated engine are operative.
					NOTE: Inoperative Fuel Flow Indicating System may cause Fuel Used Indicating System to be inoperative.
	1) Left Vertical Scale	С	28	14	(O) May be inoperative provided:
	Lamps				a) No two adjacent lamps on the affected scale are inoperative.
					b) Engine parameters are monitored during flight.
	2) Right Vertical	С	28	14	(O) May be inoperative provided:
	Scale Lamps				 a) No two adjacent lamps on the affected scale are inoperative.
					b) Engine parameters are monitored during flight.
	3) Digital indications	С	2	1	(M) May be inoperative provided:
					a) Associated vertical scale indicator is operative
					b) The affected digital indicator is covered.
31-2	EICAS Fuel Flow "FF"	В	2	1	One may be inoperative provided:
	Readouts (604, 605, 650)				a) EICAS Fuel Tank quantity readouts are operative,
	, , , ,				b) FMS Fuel Used Indication is considered inoperative and not used, and
					c) FMS Fuel Remaining Indication is considered inoperative and not used.

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_	système/série article			3.	Number Required For Dispatch	
					Nombre d'articles à expédier	
	NGINE FUEL & ONTROL				4. Remarks or Exceptions Remarques ou exceptions	
31-3	Fuel Used Indicating System ***	С	-	0	May be inoperative provided procedures are not dependent on its use.	
31-4	Fuel Low Pressure Indication					
	1) 600, 601, 601- 3A, 601-3R	С	2	1	One Fuel LOW PRESS Light may be inoperative provided both fuel boost pumps are operative.	
	2) 604, 605, 650	C	2	1	One "L/R FUEL LO PRESS" EICAS caution message may be inoperative provided both fuel boost pumps are operative. NOTE: The Fuel Fed Check Valve Test (First flight of the day) is waived for the inoperative fuel low pressure indication.	

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•	N ^o de système/série article			3.	Number Required For Dispatch
74 – <u>IG</u>	<u>NITION</u>				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
10-1	Ignition Exciters (600)	В	4	2	One exciter per engine may be inoperative provided take-off in heavy precipitation, or on a runway surface covered with standing water, slush or snow is prohibited.
10-2	Igniter Plugs (600)	В	4	2	One Igniter Plug per engine may be inoperative provided take-off in heavy precipitation, or on a runway covered with standing water, slush or snow is prohibited.
11-1	Ignition Systems (601, 601-3A, 601-3R, 604, 605, 650)				
	1) A Systems	В	2	1	One may be inoperative provided: a) Both B Systems are operative, and b) Take-off in heavy precipitation, or on a runway surface covered with standing water, slush or snow is prohibited.
	2) B Systems	В	2	0	One or both may be inoperative provided: a) Both A Systems are operative, and b) Take-off in heavy precipitation, or on a runway surface covered with standing water, slush or snow is prohibited.
11-2	Relight Ignition System Switch (600, 601)	С	1	0	May be inoperative provided continuous ignition systems A and B modes for both engines are confirmed operative prior to each flight.
		С	1	0	May be inoperative provided In Flight Start Ignition is confirmed operative on both engines and is used in lieu of relight switch.

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	système/série article			3.	Number Required For Dispatch	
74 – <u>IGNITION</u>					Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions	
30-1	IGNITION A/B "ARM/ON" Switch Lights (light function only) (604, 605, 650)	С	2	0		
30-2	IGNITION CONT"ON" Switch Light (light function only) (604, 605, 650)	C	1	0		

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Cueter	1.		2.		er Installed re d'articles installés		
-	System & Sequence Nº Item Nº de système/série article			3.	Number Required For Dispatch		
Nº de :	systeme/serie article			J.	Nombre d'articles à expédier		
76 – <u>EN</u>	NGINE CONTROL				4. Remarks or Exceptions Remarques ou exceptions		
00-1	Engine Speed Control Systems						
	1) (601, 601-3A, 601-3R, 604) without Autothrottle System	С	2	0	 (O) One or both may be inoperative provided: a) Both ENG SPEED CONTROL switches are selected OFF, b) APR is selected OFF, and c) Operations are conducted in accordance with AFM APR inoperative performance data. 		
	2) (601, 601-3A, 601-3R, 604) with Autothrottle System) (605, 650)	С	2	0	 (O) One or both may be inoperative provided: a) Both ENG SPEED CONTROL switches are selected OFF, b) APR is selected OFF, c) Operations are conducted in accordance with AFM APR inoperative performance data, and d) Autothrottle System is considered inoperative. 		
10-1	Automatic Power Reserve (APR) System (601, 601-3A, 601- 3R, 604, 605, 650)	С	1	0	May be inoperative in accordance with the performance requirements defined in the AFM applicable to take-off with the APR selected OFF.		
11-1	Engine Speed Synchronizer System (600)	С	1	0	(O) May be inoperative provided the ENG SYNC CONTROL switch is selected OFF.		

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ил де	système/série article			J.	Nombre d'articles à expédier		
76 – <u>El</u>	NGINE CONTROL				4. Remarks or Exceptions Remarques ou exceptions		
20-1	Engine Overspeed Pickups (Primary Overspeed Protection System (600)	С	4	2	(O) One per engine may be inoperative provided the associated engine(s) are shut down using the engine overspeed procedure once each flight day.		
20-2	Engine Overspeed Pickups (Secondary) Overspeed Protection System) (Post Service Bulletin ALF502L- 76-0166) (600 ***)	C	2	0	(M)(O) One or both may be inoperative provided the associated engine(s) are shutdown using the engine overspeed procedure once each flight day.		

DEPARTMENT OF TRANSPORT

MINISTÈRE DES TRANSPORTS MASTER MINIMUM EQUIPMENT LIST LISTE PRINCIPALE D'ÉQUIPEMENT MINIMAL

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System 9 Seguence NO Item		1.	2.		er Installed re d'articles installés
System & Sequence No Item No de système/série article 77 – ENGINE INDICATING				3.	Number Required For Dispatch
				0.	Nombre d'articles à expédier
					4. Remarks or Exceptions Remarques ou exceptions
10-1	N1 Vertical Scale Indicators (600, 601, 601-3A, 601-3R)	С	2	1	(O) One may be inoperative provided all other engine instruments are operative.
	Vertical Scale Lamps				
	a) Left	С	28	14	 (O) One or more lamps may be inoperative provided: a) No two adjacent lamps are inoperative, b) The lowest lamp in the red zone and the highest lamp in the green zone are operative, c) Both amber lamps are operative, d) A test is conducted before each flight to determine the inoperative lamps, and e) Engine parameters are monitored during flight.
	b) Right	C	28	14	 (O) One or more lamps may be inoperative provided: a) No two adjacent lamps are inoperative, b) The lowest lamp in the red zone and the highest lamp in the green zone are operative, c) Both amber lamps are operative, d) A test is conducted before each flight to determine the inoperative lamps, and e) Engine parameters are monitored during flight.

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System & Sequence No Item	1.	2.		er Installed ere d'articles installés
Nº de système/série article			3.	Number Required For Dispatch
77 – ENGINE INDICATING				Nombre d'articles à expédier 4. Remarks or Exceptions Remarques ou exceptions
10-2 N2 Vertical Scale Indicators (600, 601,601-3A, 601-3R)	С	2	1	(O) One may be inoperative provided all other engine instruments are operative.
1) Vertical scale lamps (600)				
a) Left	С	28	14	 (O) One or more lamps may be inoperative provided: a) No two adjacent lamps are inoperative, b) The two amber lamps adjacent to the green zone (upper and lower sides) and the lowest lamp in the red zone are operative, c) The top and bottom green lamps (adjacent to the upper and lower amber zones) are operative, d) A test is conducted before each flight to determine the inoperative lamps, and e) Engine parameters are monitored during flight.
b) Right	С	28	14	 (O) One or more lamps may be inoperative provided: a) No two adjacent lamps are inoperative, b) The two amber lamps adjacent to the green zone (upper and lower sides) and the lowest lamp in the red zone are operative, c) The top and bottom green lamps (adjacent to the upper and lower amber zones) are operative' d) A test is conducted before each flight to determine the inoperative lamps, and e) Engine parameters are monitored during flight.
(Cont'd)				

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Suptom 9 Sequence NO Item		1.	2.	Number Installed Nombre d'articles installés				
-	System & Sequence Nº Item Iº de système/série article			3.				
14° de systeme/serie article					Nombre d'articles à expédier			
77 – <u>EN</u>	GINE INDICATING				4. Remarks or Exceptions Remarques ou exceptions			
10-2	N2 Vertical Scale Indicators (600, 601, 601-3A, 601-3R) (Cont'd)							
	2) Vertical scale lamps (601, 601-3A, 601-3R)							
	a) Left	С	28	14	 (O) One or more lamps may be inoperative provided: a) No two adjacent lamps are inoperative, b) The highest lamp in the green zone, the yellow lamp and the lowest lamp in the red zone are operative, c) A test is conducted before each flight to determine the inoperative lamps, and d) Engine parameters are monitored during flight. 			
	b) Right	С	28	14	 (O) One or more lamps may be inoperative provided: a) No two adjacent lamps are inoperative, b) The highest lamp in the green zone, the yellow lamp and the lowest lamp in the red zone are operative, c) A test is conducted before each flight to determine the inoperative lamps, and d) Engine parameters are monitored during flight. 			

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System	& Sequence No Item	1.	2.		er Installed re d'articles installés
-	ystème/série article			3. Number Required For Dispatch Nombre d'articles à expédier	
77 – <u>EN</u>	GINE INDICATING				4. Remarks or Exceptions Remarques ou exceptions
20-1	ITT Indicators (600, 601, 601-3A, 601-3R)	В	4	2	Either the digital or the vertical scale indicators, but not both, may be inoperative for each engine. If starting with DC Electrical Power Only: NOTE 1: When the Right Engine Digital ITT Indication is inoperative, a Left Engine start must be accomplished first. NOTE 2: When the Left Engine Vertical Scale ITT Indication is inoperative, a Right Engine start must be accomplished first.
	Vertical Scale Lamps				
	a) Left	В	28	14	 (O) One or more lamps may be inoperative provided: a) No two adjacent lamps may be inoperative, b) The highest lamp in the green zone, the yellow lamps and the lowest lamp in the red zone are operative, c) A test is conducted before each flight to determine the inoperative lamps, and d) Engine parameters are monitored during flight.
	b) Right	В	28	14	 (O) One or more lamps may be inoperative provided: a) No two adjacent lamps may be inoperative b) The highest lamp in the green zone, the yellow lamps and the lowest lamp in the red zone are operative. c) A test is conducted before each flight to determine the inoperative lamps, and d) Engine parameters are monitored during flight.

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_	System & Sequence No Item			Nomb	re d'articles installés Number Required For Dispatch
No de s	système/série article			3.	Nombre d'articles à expédier
77 – FN	IGINE INDICATING				4. Remarks or Exceptions
77 <u>L.</u>	<u> </u>				Remarques ou exceptions
20-2	DGT OFF switch (600, 601, 601-3A, 601-3R)	С	1	0	(M) May be inoperative in the ON position.
30-1	Engine Vibration Monitor	C	2	0	One or both may be inoperative provided: a) Flight is not conducted in known or forecast icing conditions, and b) One Ice Detection System is operative.

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1 System & Sequence No Item	. 2.		er Installed re d'articles installés
Nº de système/série article		3.	Number Required For Dispatch
			Nombre d'articles à expédier
77 – ENGINE INDICATING			4. Remarks or Exceptions Remarques ou exceptions
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1.		2.	Number Installed Nombre d'articles installés			
•	stem & Sequence Nº Item			3.	Number Required For Dispatch	
Nº de s	système/série article			J.	Nombre d'articles à expédier	
78 – F)	KHAUST				4. Remarks or Exceptions	
70 <u>L7</u>	<u> </u>				Remarques ou exceptions	
30-1	Thrust Reverser	С	2	1	(M) One may be inoperative provided:	
	Systems				a) The affected thrust reverser does not have structural damages other than cracks in the cascades.	
					 b) Affected Thrust Reverser is deactivated and secured in the forward thrust position, 	
					c) Nosewheel Steering System is operative,	
					 d) For Take-off on wet or contaminated runways and landing on contaminated runways, operations are conducted in accordance with: 	
					(600) AFM Unapproved Supplement 3 (Operations on Contaminated Runways.)	
					(601, 601-3A, 601-3R) AFM Unapproved Supplement 2 (Operations on Contaminated Runways.)	
					(604, 605, 650) AFM Supplement 2 (Operations on Wet and Contaminated Runways.)	
					NOTE: Cracks in the cascades must be repaired or the cascades replaced before the affected thrust reverser is activated.	

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1. System & Sequence No Item	2.	Numb Nomb	per Installed pre d'articles installés
Nº de système/série article	İ	3.	Number Required For Dispatch
	1		Nombre d'articles à expédier
78 – <u>EXHAUST</u>			4. Remarks or Exceptions
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System	& Sequence Nº Item	1.	2.	Number Installed Nombre d'articles installés	
1	ystème/série article			3. Number Required For Dispatch	
					Nombre d'articles à expédier
79 – <u>OIL</u>	=				4. Remarks or Exceptions Remarques ou exceptions
12-1	Oil Replenishment System (601, 601-3A, 601- 3R, 604, 605, 650)	С	1	0	(M) One may be inoperative provided alternate methods for checking engine oil levels and servicing engine oil are established.
30-1	Oil Pressure Indicators (600, 601, 601-3A, 601-3R)	С	2	0	One or both may be partially inoperative to the extent that every alternate segment of the indicator strip may be inoperative provided: a) Associated Low Pressure Warning Light is operative, and b) Associated Oil Temperature Indicator in operative.
30-2	Oil Pressure Readouts (604, 605, 650)	В	2	1	 (M) One may be inoperative provided: a) Associated EICAS "ENG OIL PRESS" warning message is operative, b) Oil quantity is checked prior to departure, and c) Engine is operated at idle for 2 minutes prior to departure.
30-3	Oil Temperature Indicators (600, 601, 601-3A, 601-3R)	С	2	0	One or both may be partially inoperative to the extent that every alternate segment of the indicator strip may be inoperative provided the associated oil pressure indicator is operative.
30-4	Oil Low Pressure Warning Lights L or R LOP (600, 601, 601-3A, 601-3R)	В	2	1	One may be inoperative provided: a) Indicator light is extinguished or blanked, and b) Associated oil pressure indicator is operative and monitored.
30-5	Oil Low Pressure Switch (604, 605, 650)	В	2	1	One may be inoperative provided both oil pressure readouts are operative.

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System	a & Sequence Nº Item	1.	2.	Number Installed Nombre d'articles installés	
1 -	système/série article			3. Number Required For Dispatch	
	•				Nombre d'articles à expédier
79 – <u>Ol</u>	<u>L</u>				4. Remarks or Exceptions
					Remarques ou exceptions
30-6	Engine Oil Chip Detector Indication (Aircraft with PRE SB 601-0639 or 604-79-002 or 605-79-001 or 650-79-001)	A	2	2	Remarques ou exceptions (O) May be dispatched with left or right chip detector activated provided: a) Associated chip detector check must be completed within 25 engine hours of initial indication, and b) Associated engine oil system parameters and the engine vibration must be monitored closely. If there is a parameter over the limits, troubleshooting of the engine must be done before the next flight.